

# **LECTURE # 09: FLOW VISUALIZATION TECHNIQUES: SHADOWGRAPH AND SCHLIEREN**

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**Sources/ Further reading:**

Hecht, "Optics" 4<sup>th</sup> ed.

Raffel, Willert, Wereley, Kompenhans, "Particle image velocimetry: A practical guide" 2<sup>nd</sup> ed.

Tropea, Yarin, & Foss, "Springer Handbook of Experimental Fluid Mechanics," Part B Ch 6

# SCHLIEREN IMAGING IN SLOW MOTION

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- ***How Well Do Masks Work? (Schlieren Imaging In Slow Motion!)***

<https://www.youtube.com/watch?v=0Tp0zB904Mc>



*Credit: Matthew Stagmates/NIST*

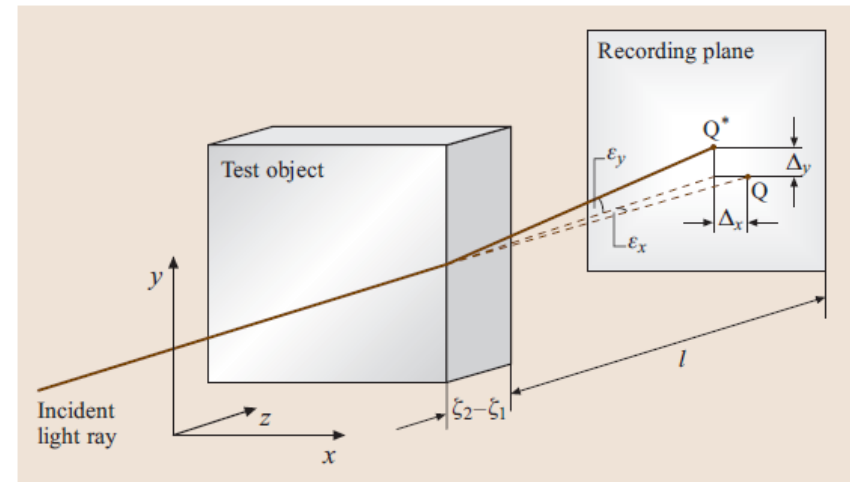
# Index of refraction and thermodynamic state

- Index of refraction is a function of thermodynamic state (density) for homogeneous medium:
- Lorenz-Lorentz relationship:  $\frac{1}{\rho} \frac{n^2 - 1}{n^2 + 2} = K$
- When  $n \approx 1$ , for gaseous flow:  $\frac{n-1}{\rho} = K \Rightarrow K\rho = n-1$  ← Gladstone-Dale Eqn
- At standard condition, with  $n_0$  and  $\rho_0$ :  $\frac{n_0 - 1}{\rho_0} = K \Rightarrow n - 1 = \frac{\rho}{\rho_0} (n_0 - 1)$   
 $\Rightarrow \rho = \rho_0 \frac{n - 1}{n_0 - 1}$
- First- and second-derivative is determined by schlieren and shadowgraph apparatus:
 
$$\frac{\partial \rho}{\partial y} = \frac{1}{const} \frac{\partial n}{\partial y} \Rightarrow \frac{\partial \rho}{\partial y} = \frac{\rho_0}{n_0 - 1} \frac{\partial n}{\partial y}$$

$$\frac{\partial^2 \rho}{\partial y^2} = \frac{1}{const} \frac{\partial^2 n}{\partial y^2} \Rightarrow \frac{\partial^2 \rho}{\partial y^2} = \frac{\rho_0}{n_0 - 1} \frac{\partial^2 n}{\partial y^2}$$

# Shadowgraphy and Schlieren Techniques

- **Index of refraction:**  $n = c / v = \frac{\lambda_0}{\lambda} > 1$
- **Depends on the variation of the index of refraction in a transparent medium, which affects the light rays passing through.**
- **Shadowgraphy:** used to indicate the variation of the **second derivatives** (normal to the light beam) of the index of refraction.
- **Schlieren systems:** used to indicate the variation of the **first derivative** of the index of refraction



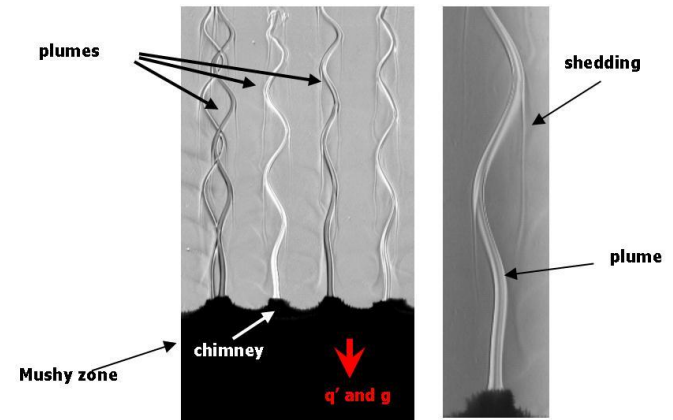
**Fig. 6.1** Refractive deflection of a light ray in an object field (flow) with varying refractive index (caused by varying fluid density)



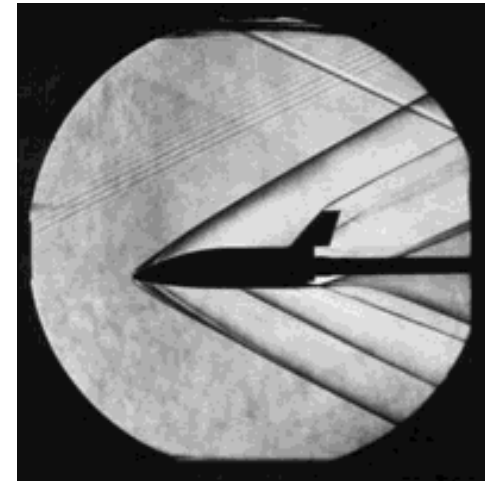
Schlieren of a .30-06 caliber high-powered rifle muzzle blast from (by Gary S. Settles)

# Shadowgraphy and Schlieren Techniques

- *Shadowgraphy and Schlieren systems are often used in shock waves and flame phenomena, in which density gradient is quite big.*
- *While these techniques are mostly used for qualitative flow visualization, they can be used to map pressure, density, or temperature measurements theoretically.*
- *These techniques are often used to determine the integrated quantity over the length of light beam.*



shadowgraph image of plumes during solidification process (by Lum Chee)



Schlieren image

# Introduction-3

- *Index of refraction is a function of thermodynamic state (density) for homogeneous medium:*

- *Lorenz-Lorentz relationship:* 
$$\frac{1}{\rho} \frac{n^2 - 1}{n^2 + 2} = \text{const}$$

- *When  $n \approx 1$ , for gaseous flow:* 
$$\frac{n - 1}{\rho} = \text{const} \Rightarrow \rho = \frac{n - 1}{\text{const}}$$

- *at standard condition, with  $n_0$  and  $\rho_0$ :* 
$$\frac{n_0 - 1}{\rho_0} = \text{const} \Rightarrow n - 1 = \frac{\rho}{\rho_0} (n_0 - 1)$$
$$\Rightarrow \rho = \rho_0 \frac{n - 1}{n_0 - 1}$$

- *When first and second derivative is determined as in Schlieren and shadowgraph apparatus:*

$$\frac{\partial \rho}{\partial y} = \frac{1}{\text{const}} \frac{\partial n}{\partial y} \Rightarrow \frac{\partial \rho}{\partial y} = \frac{\rho_0}{n_0 - 1} \frac{\partial n}{\partial y}$$
$$\frac{\partial^2 \rho}{\partial y^2} = \frac{1}{\text{const}} \frac{\partial^2 n}{\partial y^2} \Rightarrow \frac{\partial^2 \rho}{\partial y^2} = \frac{\rho_0}{n_0 - 1} \frac{\partial^2 n}{\partial y^2}$$

# Introduction-4

- *Application of the Schlieren and shadowgraphy techniques:*
  - *Compressible flow with shock waves  $\Rightarrow$  density changes*
  - *Natural convective flow  $\Rightarrow$  density changes*
  - *Flame and combustion system:  $\Rightarrow$  density changes*
- *Temperature changes inside flows:*
  - *For low speed flow with heat transfer:*

–  *$P = \text{constant}$*

$$\rho = P / RT \Rightarrow \frac{\partial \rho}{\partial y} = \frac{P}{RT^2} \frac{\partial T}{\partial y} = \frac{\rho}{T} \frac{\partial T}{\partial y}$$

$$\Rightarrow \frac{\partial n}{\partial y} = \frac{n_0 - 1}{\rho_0} \frac{\partial \rho}{\partial y} = \frac{n_0 - 1}{T} \frac{\rho}{\rho_0} \frac{\partial T}{\partial y}$$

$$\Rightarrow \frac{\partial T}{\partial y} = \frac{T}{n_0 - 1} \frac{\rho_0}{\rho} \frac{\partial n}{\partial y}$$

$$\Rightarrow \frac{\partial^2 n}{\partial y^2} = \frac{n_0 - 1}{\rho_0} \left[ -\frac{\rho}{T} \frac{\partial^2 T}{\partial y^2} + \frac{2\rho}{T^2} \left( \frac{\partial T}{\partial y} \right)^2 \right]$$

# Deflection of light rays

- According to definition of index of refraction, the light velocity will be  $V=C_0/n$ .
- The slope of the wave front of the light:  $\frac{dy}{dz}$
- If the angle  $\Delta\alpha'$  is quite small:

$$\Delta Z = \frac{C_0}{n} \Delta \tau$$

$$\Delta^2 Z = \Delta Z - \Delta Z_{y+\Delta y} = -C_0 \left( \Delta \left( \frac{1}{n} \right) / \Delta y \right) \Delta \tau \Delta y$$

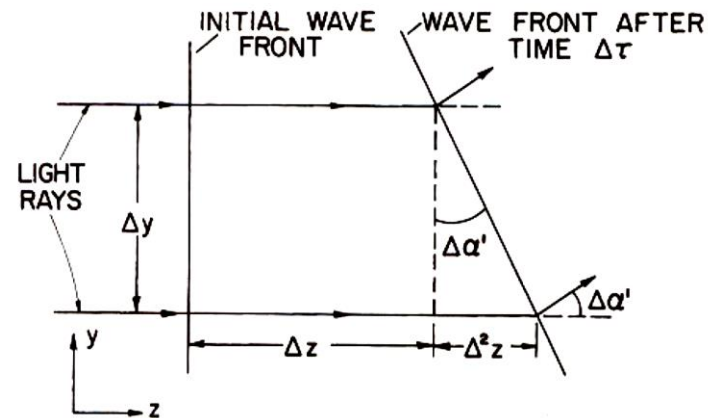
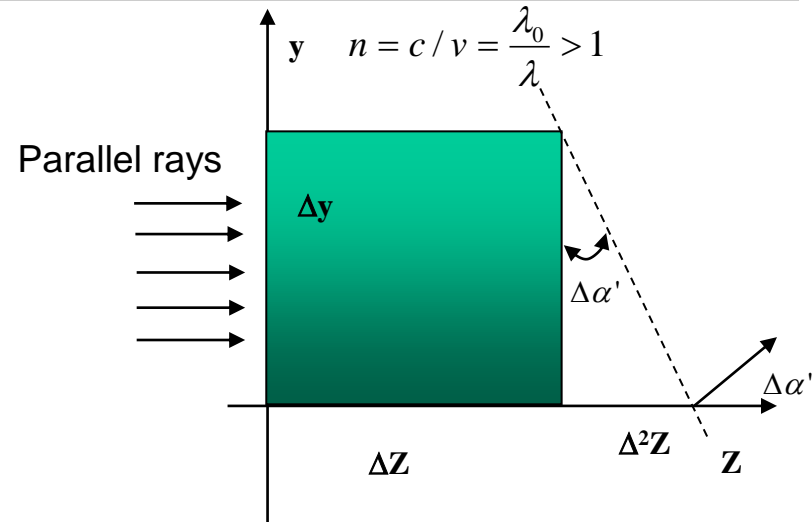
$$\Delta \alpha' = \frac{\Delta^2 Z}{\Delta y} = -n \left( \Delta \left( \frac{1}{n} \right) / \Delta y \right) \Delta Z$$

$$\frac{dy}{dz} = d\alpha' = -n \left[ \frac{d \left( \frac{1}{n} \right)}{dy} \right] dz = n \frac{1}{n^2} \left[ \frac{dn}{dy} \right] dz = \frac{1}{n} \left( \frac{dn}{dy} \right) dz = \frac{d(\ln n)}{dy} dz$$

$$\frac{d^2 y}{dz^2} = \frac{d(\ln n)}{dy}$$

$$d\alpha' = -n \left[ \frac{d \left( \frac{1}{n} \right)}{dy} \right] dz = n \frac{1}{n^2} \left[ \frac{dn}{dy} \right] dz = \frac{1}{n} \left( \frac{dn}{dy} \right) dz = \frac{d(\ln n)}{dy} dz$$

$$\Rightarrow \alpha' = \int \frac{1}{n} \left( \frac{dn}{dy} \right) dz \quad n \approx 1 \quad \Rightarrow \quad \alpha' = \int \frac{dn}{dy} dz$$



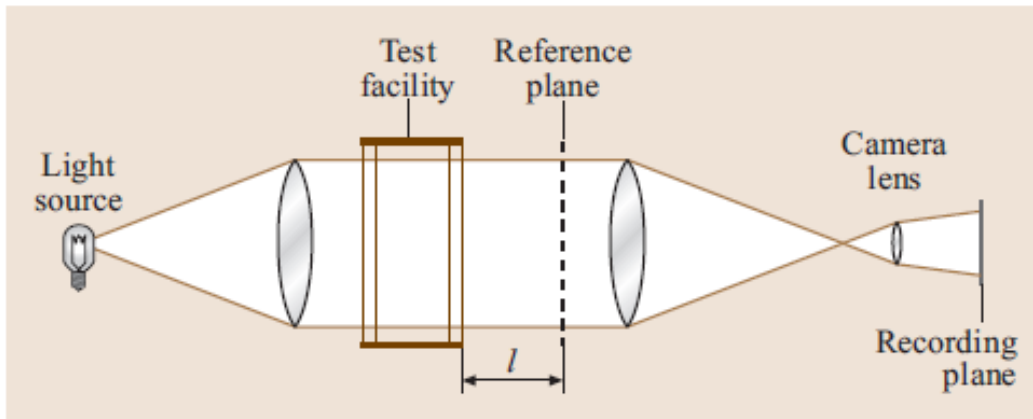


# Shadowgraphy

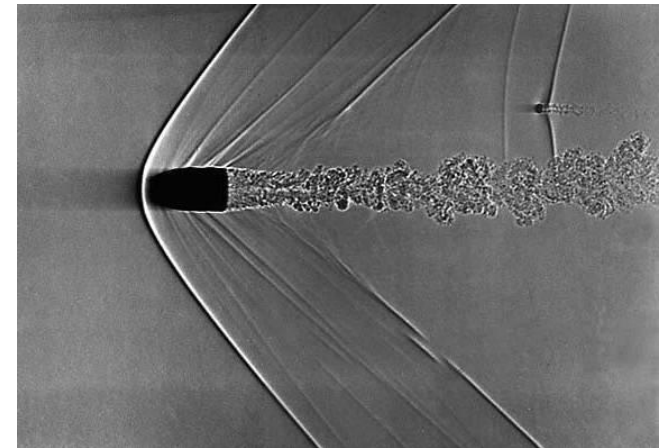
- In shadowgraphy, as light rays pass through the measurement region, the deflection of the light rays as they interact with variations in the optical index lead to an intensity distribution:

$$\frac{\Delta I}{I} = l \int_{\zeta_1}^{\zeta_2} \left( \frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} \right) (\ln n) dz$$

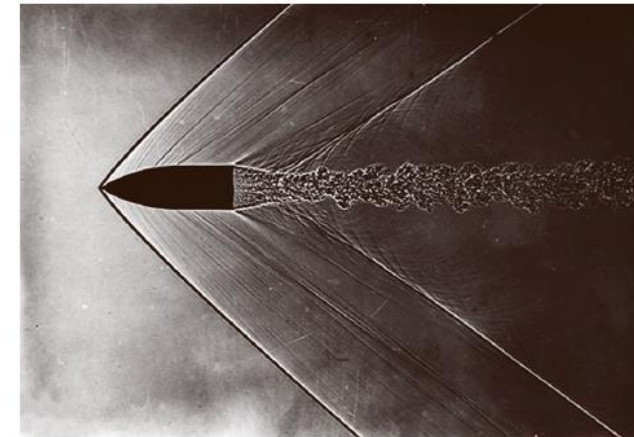
- For *weak refraction*, and applying the Gladstone-Dale formula reveals a dependence on the second partial derivatives of density.



**Fig. 6.2** Shadowgraph setup with parallel beams through the test object

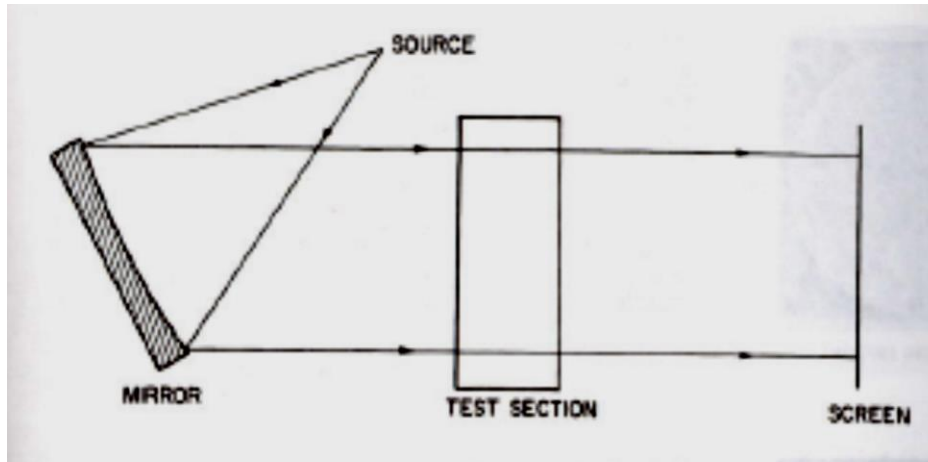


Shadowgraph of a bullet (by Andrew Davidhazy )

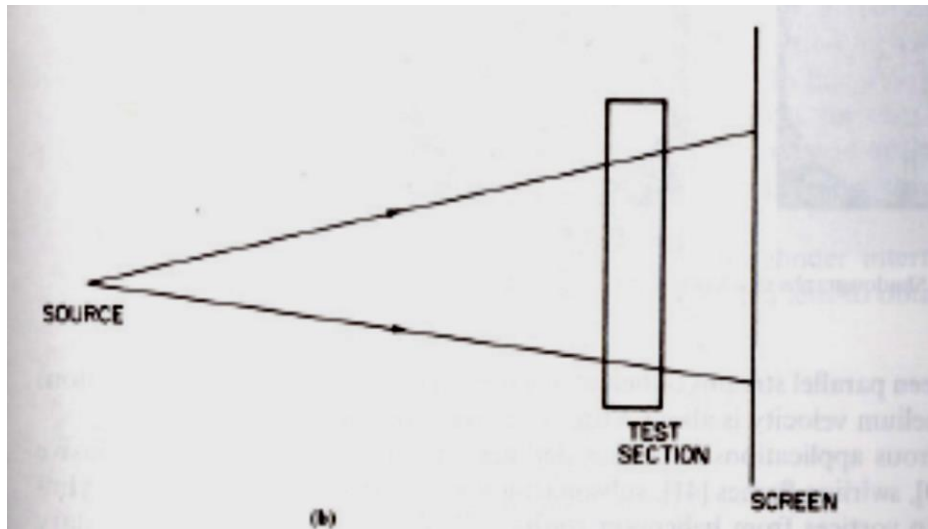


**Fig. 6.3** Shadowgraph of a bullet flying at supersonic velocity (courtesy Deutsch-Französisches Forschungsinstitut, ISL, St. Louis, France)

# Setup of a Shadowgraph imaging system

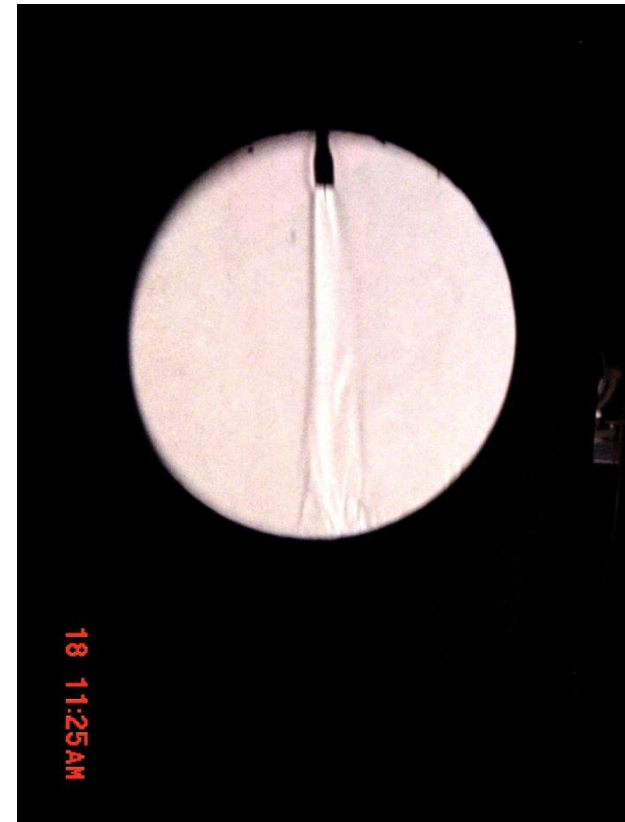
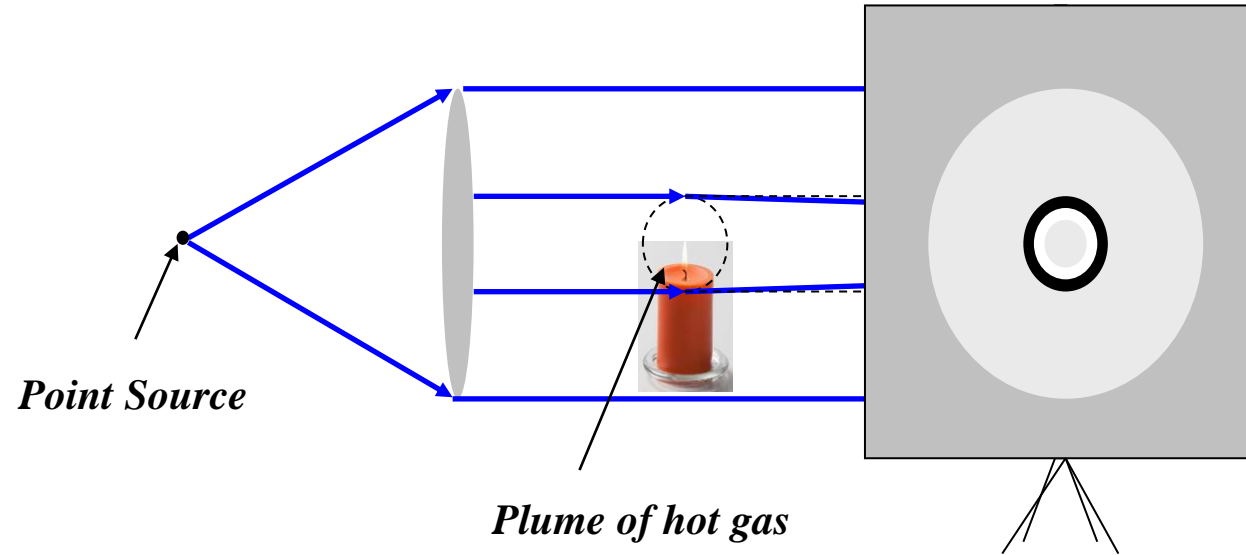


Experimental setup with one converging mirror

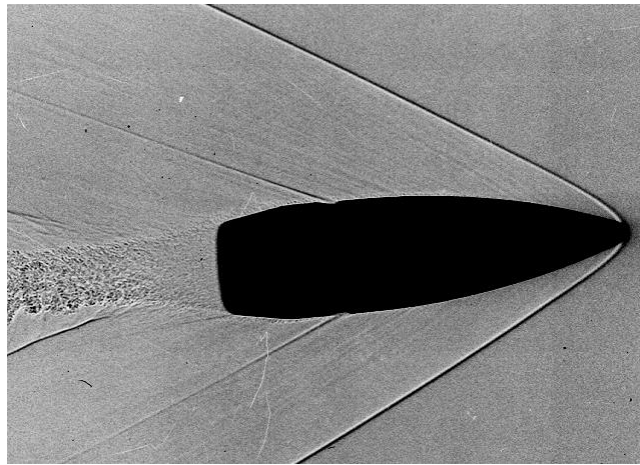
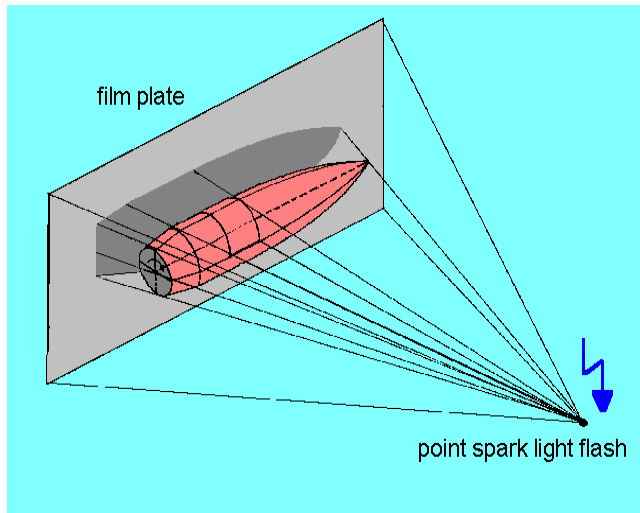


Experimental setup without lens or mirror

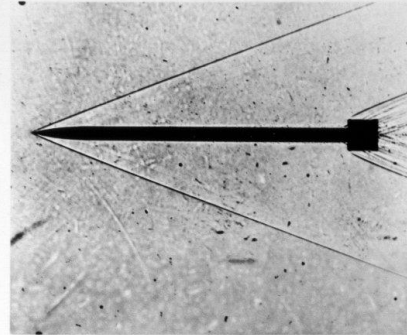
# Direct Shadowgraph



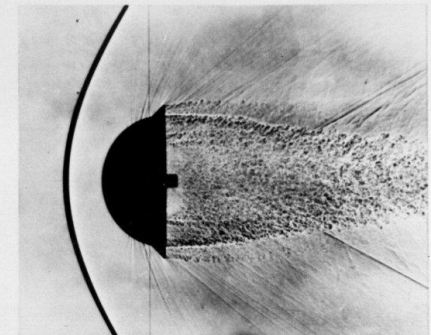
# Examples: Shadowgraph images



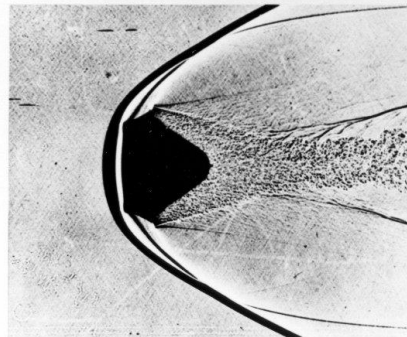
## RESEARCH CONTRIBUTING TO PROJECT MERCURY



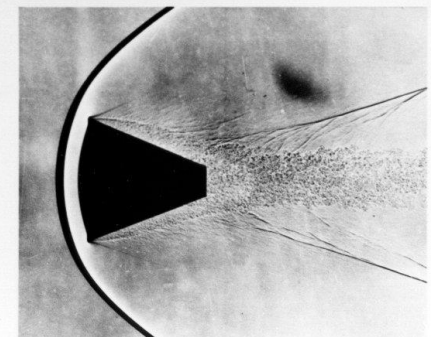
INITIAL CONCEPT



BLUNT BODY CONCEPT 1953



MISSILE NOSE CONES 1953-1957

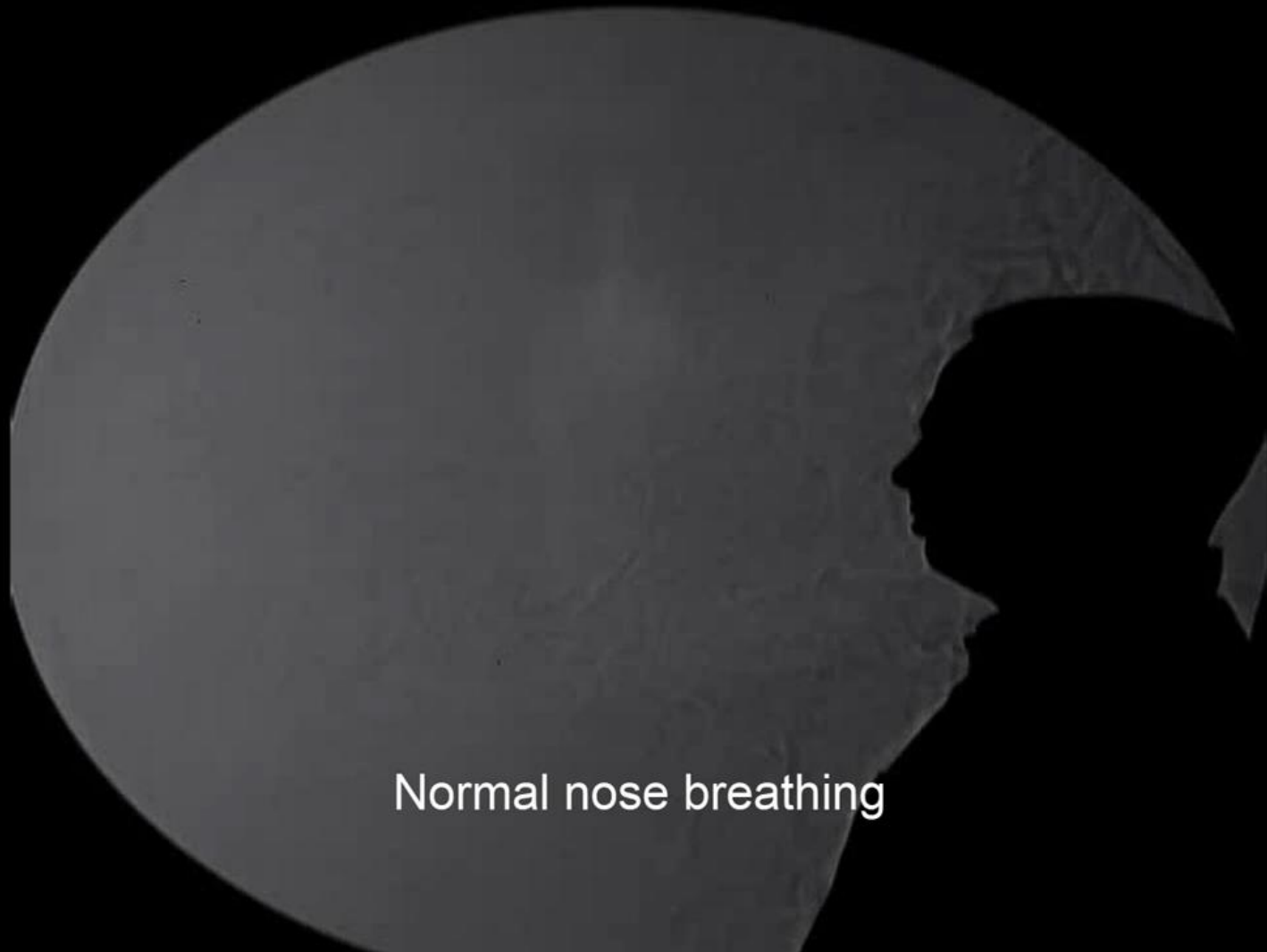


MANNED CAPSULE CONCEPT 1957

Shadowgraph Images of Re-entry Vehicles

# SHADOWGRAPH IMAGING EXAMPLE

- *Shadowgraph Imaging of Human Exhaled Airflows: An Aid to Aerosol Infection Control*
- <https://www.youtube.com/watch?v=gElHX1AII0Y>



Normal nose breathing

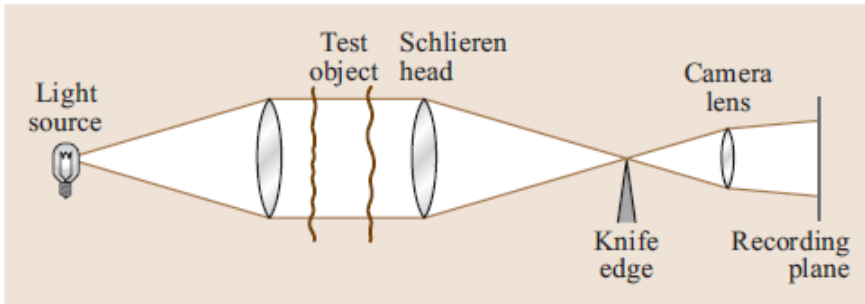
# Schlieren

- In Schlieren, as light rays pass through index variations in the measurement region, the deflection of the light rays cause them to be either blocked or pass a knife edge:

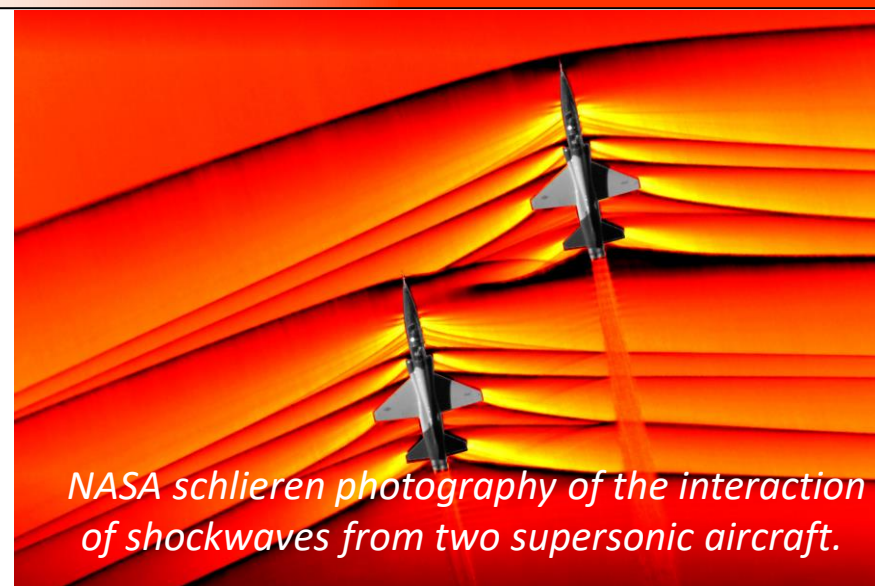
$$\frac{\Delta I}{I} = \frac{f_2}{a} \int_{\zeta_1}^{\zeta_2} \frac{1}{n} \frac{\partial n}{\partial y} dz.$$

- For small angles of deflection, and applying the Gladstone-Dale formula reveals a dependence on the partial derivatives of density.

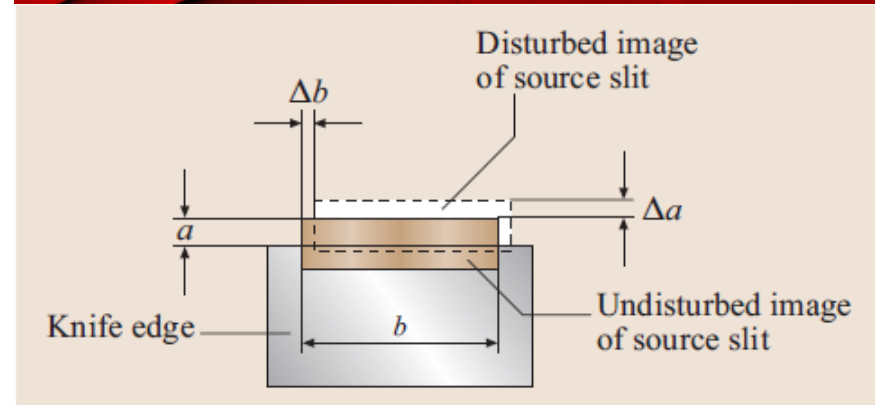
$$\frac{\Delta I}{I} = \frac{K f_2}{a} \int_{\zeta_1}^{\zeta_2} \frac{\partial \rho}{\partial y} dz$$



**Fig. 6.5** Schlieren setup with parallel light through the test field



*NASA schlieren photograph of the interaction of shockwaves from two supersonic aircraft.*



**Fig. 6.6** Image of a light source of size  $a \times b$  in the focal plane of the schlieren head, as seen in the direction of the optical axis; shift of the light source by  $\Delta a$  and  $\Delta b$ , respectively, caused by light deflection in the refractive index object

# FUNDAMENTALS OF SCHLIEREN TECHNIQUE

- According to definition of index of refraction, the light velocity will be  $V=C_0/n$ .

- The slope of the wave front of the light:  $\frac{dy}{dz}$

- If the angle  $\Delta\alpha'$  is quite small.

$$\Delta Z = \frac{C_0}{n} \Delta \tau$$

$$\Delta^2 Z = \Delta Z - \Delta Z_{y+\Delta y} = -C_0 \left( \Delta \left( \frac{1}{n} \right) / \Delta y \right) \Delta \tau \Delta y$$

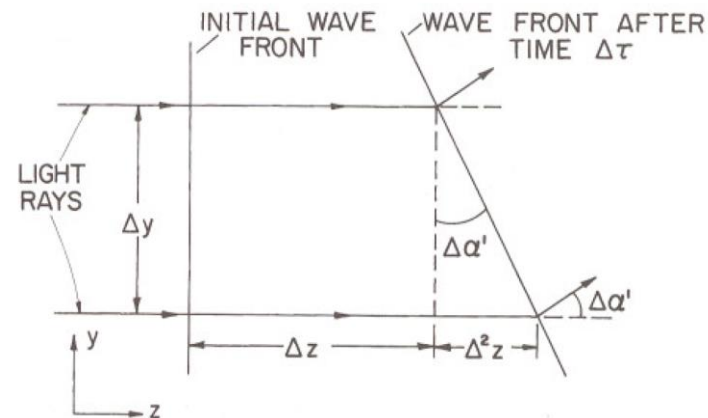
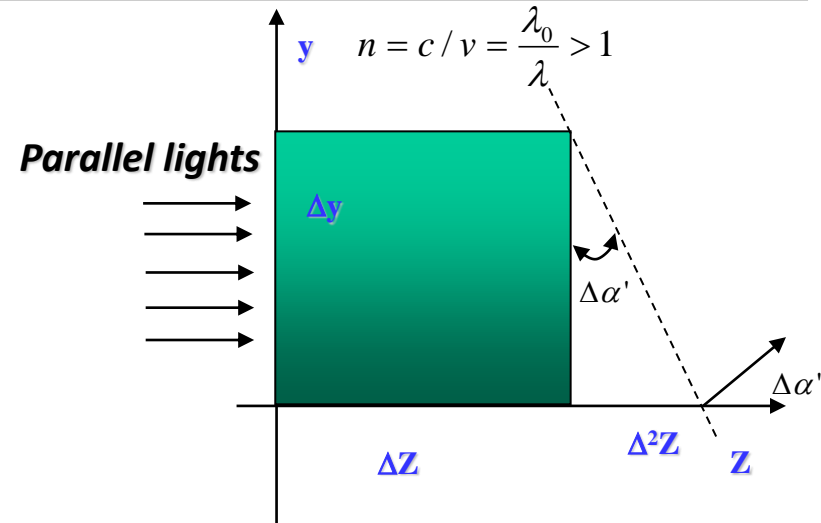
$$\Delta \alpha' = \frac{\Delta^2 Z}{\Delta y} = -n \left( \Delta \left( \frac{1}{n} \right) / \Delta y \right) \Delta Z$$

$$\frac{dy}{dz} = d\alpha' = -n \left[ \frac{d\left(\frac{1}{n}\right)}{dy} \right] dz = n \frac{1}{n^2} \left[ \frac{dn}{dy} \right] dz = \frac{1}{n} \left( \frac{dn}{dy} \right) dz = \frac{d(\ln n)}{dy} dz$$

$$\frac{d^2 y}{dz^2} = \frac{d(\ln n)}{dy}$$

$$d\alpha' = -n \left[ \frac{d\left(\frac{1}{n}\right)}{dy} \right] dz = n \frac{1}{n^2} \left[ \frac{dn}{dy} \right] dz = \frac{1}{n} \left( \frac{dn}{dy} \right) dz = \frac{d(\ln n)}{dy} dz$$

$$\Rightarrow \alpha' = \int \frac{1}{n} \left( \frac{dn}{dy} \right) dz \quad n \approx 1 \quad \Rightarrow \quad \alpha' = \int \frac{dn}{dy} dz$$



# Shadowgraph technique

$$I_{sc} = \frac{\Delta y}{\Delta y_{sc}} I_0$$

$$\Delta y_{sc} = \Delta y + Z_{sc} \cdot d\alpha$$

$$\frac{\Delta I}{I_0} = \frac{I_{sc} - I_0}{I_0} = \frac{\Delta y}{\Delta y_{sc}} - 1$$

$$= -Z_{sc} \cdot \frac{d\alpha}{\Delta y_{sc}} \approx -Z_{sc} \cdot \frac{d\alpha}{dy}$$

$$\Rightarrow \frac{\Delta I}{I_0} \approx -Z_{sc} \cdot \frac{d\alpha}{dy}$$

since

$$\alpha = \frac{1}{n_a} \int \frac{dn}{dy} dz$$

$$\Rightarrow \frac{\Delta I}{I_0} = \frac{-Z_{sc}}{n_a} \cdot \int \frac{d^2 n}{dy^2} dz$$

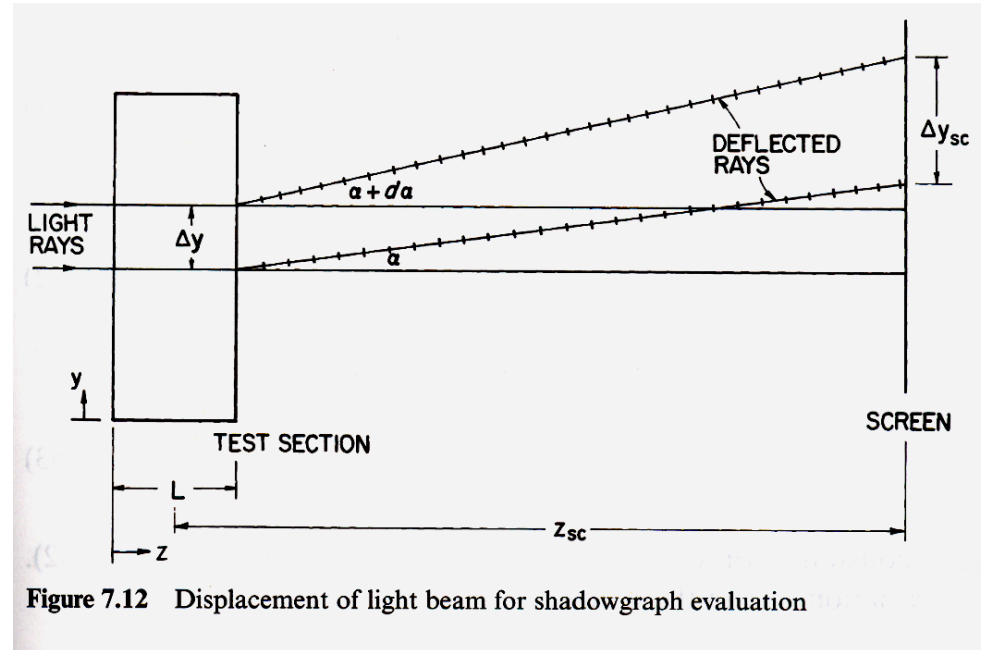
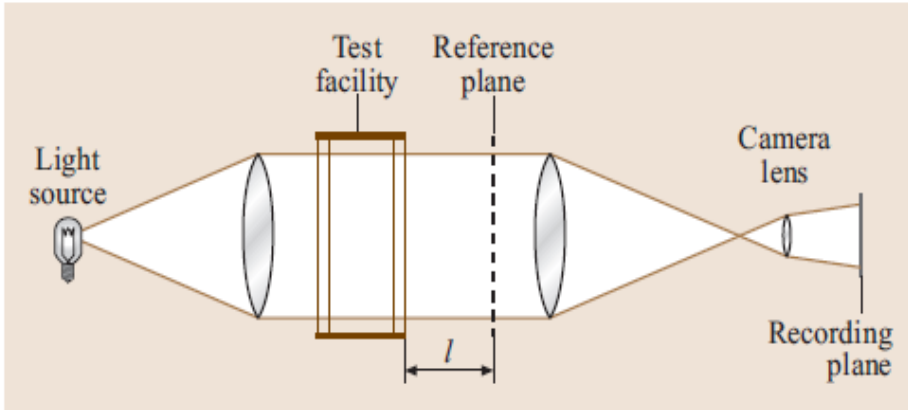


Figure 7.12 Displacement of light beam for shadowgraph evaluation

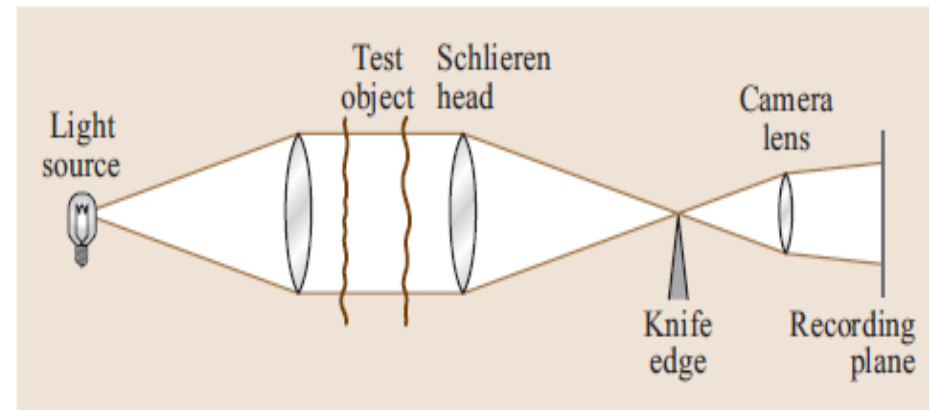
- Sensitivity is proportional to index of refraction  $1/n$ , and screen distance  $Z_{sc}$



# Schlieren concept

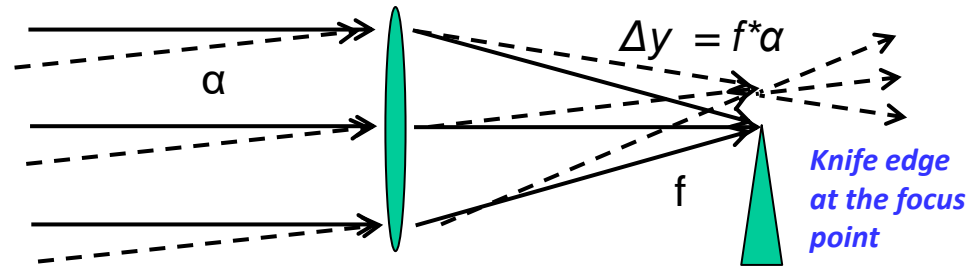


**Fig. 6.2** Shadowgraph setup with parallel beams through the test object

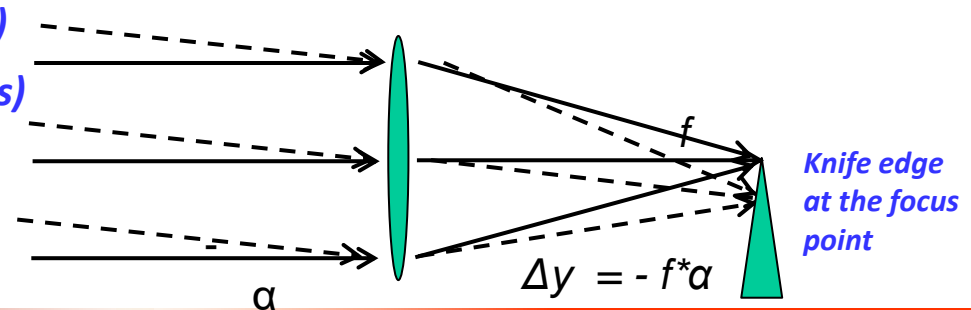


**Fig. 6.5** Schlieren setup with parallel light through the test field

- *Parallel rays are focused at len's focal distance*
- *Deflected rays are focused off-axis*
- *Parallel rays at angle  $\alpha$  to optical axis are displaced  $\Delta y = f \cdot \alpha$*
- *Suppose a knife edge is added*
- *Rays deflected away are passed (bright regions)*
- *Rays deflected toward are blocked (dark regions)*



- *Schlieren technique*



# FUNDAMENTALS OF SCHLIEREN TECHNIQUE

- The intensity after the sharp razor blade (knife edge) before the experiment

$$I_k = \frac{a_K}{a_0} I_0$$

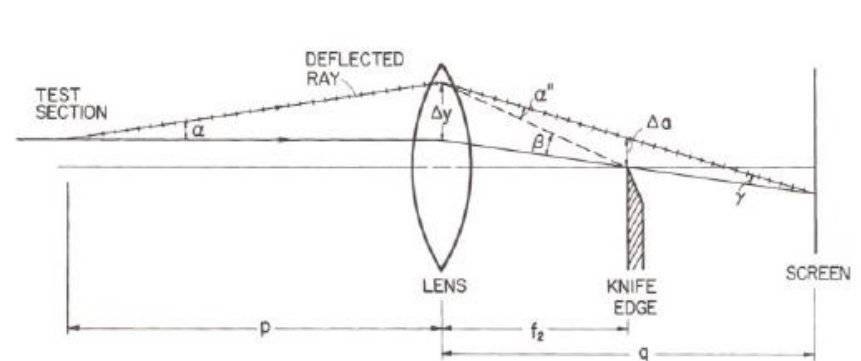
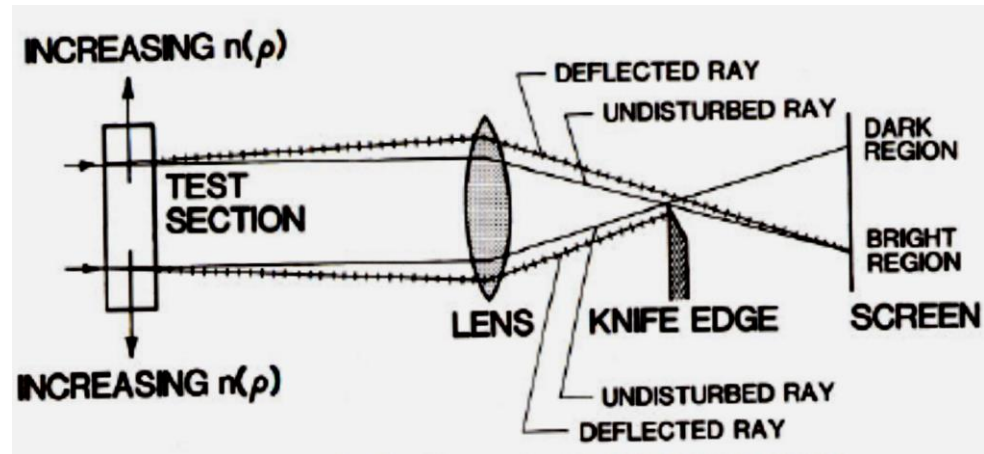
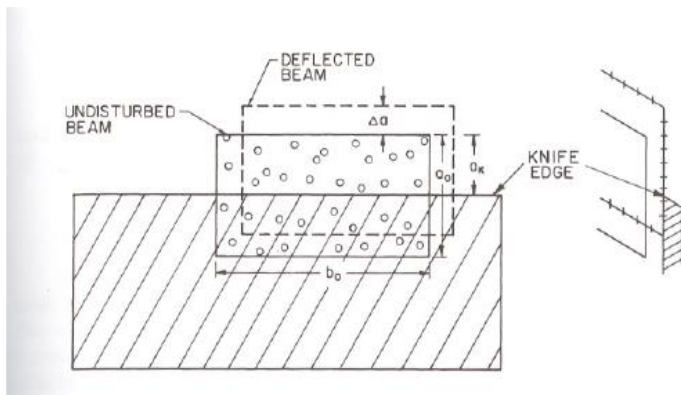
- The intensity after the deformation due to the variation of the index of refraction

$$I_d = I_k + \frac{\Delta a}{a_K} I_k = \left(1 + \frac{\Delta a}{a_K}\right) I_k$$

$$\text{contrast} = \frac{\Delta I}{I_k} = \frac{I_d - I_k}{I_k} = \frac{\Delta a}{a_K} = \pm \frac{\alpha f_2}{a_K}$$

$$\text{sensitivity} : \frac{d(\text{contrast})}{d\alpha} = \frac{f_2}{a_K}$$

- Sensitivity is proportional to  $f_2$  and inversely to  $a_k$ .



FOR  $\alpha$  SMALL

$$\alpha = \Delta y / p$$

$$\beta = \Delta y / f_2$$

$$\gamma = \Delta y / q$$

$$\alpha'' = \beta - \gamma$$

$$= \Delta y (1/f_2 - 1/q) = \Delta y / p = \alpha$$

$$\therefore \Delta a = \alpha f_2$$

Figure 7.4 Ray displacement at knife-edge for a given angular deflection

# FUNDAMENTALS OF SCHLIEREN TECHNIQUE

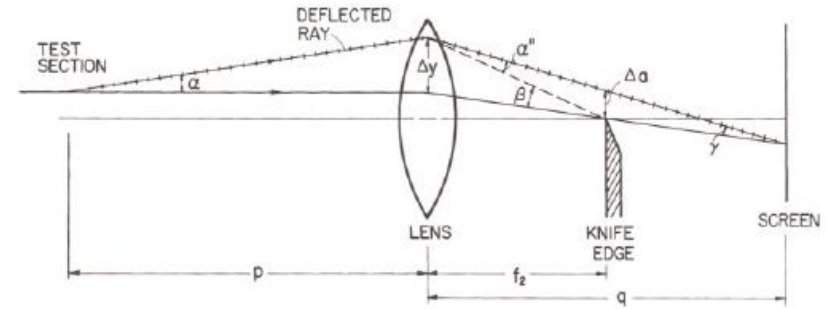
- For a gas flow with density change:

$$\frac{\Delta I}{I_k} = \pm \frac{\alpha f_2}{a_K}$$

$$\alpha' = \int \frac{dn}{dy} dz \Rightarrow \frac{\Delta I}{I_k} = \pm \frac{f_2}{a_K} \int \frac{dn}{dy} dz$$

$$\frac{\partial \rho}{\partial y} = \frac{\rho_0}{n_0 - 1} \frac{\partial n}{\partial y} \Rightarrow \frac{\Delta I}{I_k} = \pm \frac{f_2}{a_K} \frac{n_0 - 1}{\rho_0} \int \frac{d\rho}{dy} dz$$

$$n \approx 1 \Rightarrow \frac{\Delta I}{I_k} = \pm \frac{f_2}{a_K} \frac{n_0 - 1}{\rho_0} \frac{d\rho}{dy} L$$



FOR  $\alpha$  SMALL

$$\alpha = \Delta y/p$$

$$\beta = \Delta y/f_2$$

$$\gamma = \Delta y/q$$

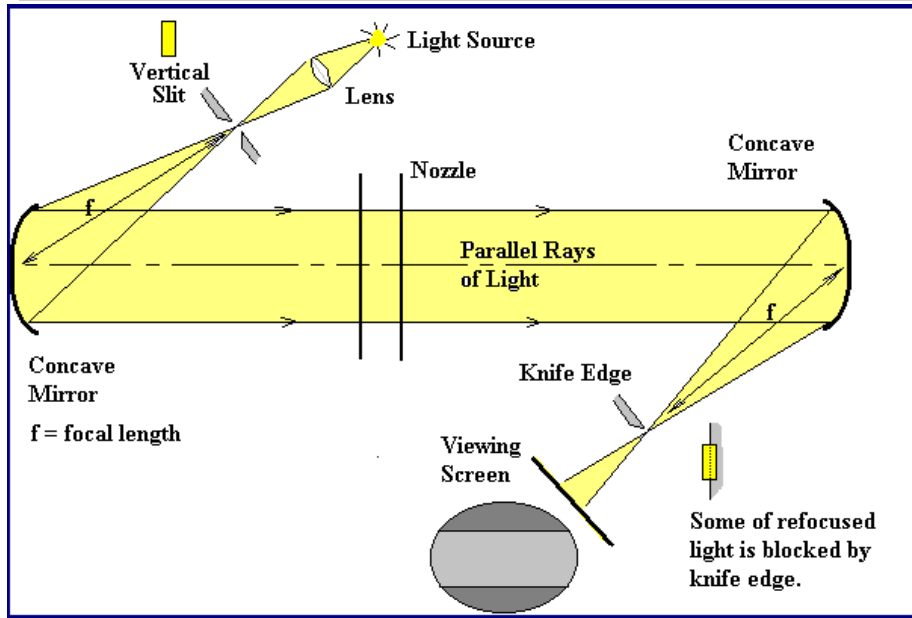
$$\alpha' = \beta - \gamma$$

$$= \Delta y(1/f_2 - 1/q) = \Delta y/p = \alpha$$

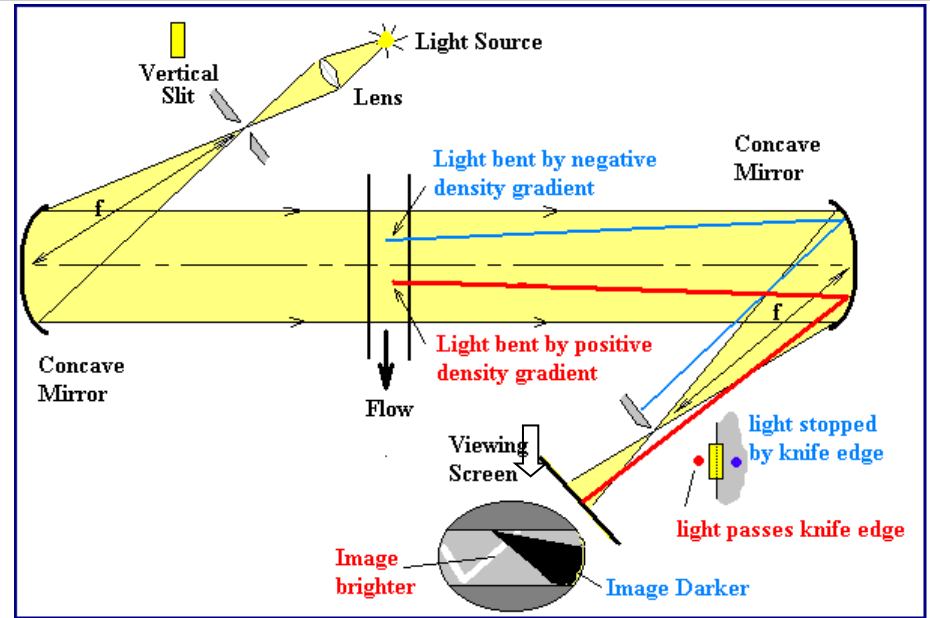
$$\therefore \Delta a = \alpha f_2$$

Figure 7.4 Ray displacement at knife-edge for a given angular deflection

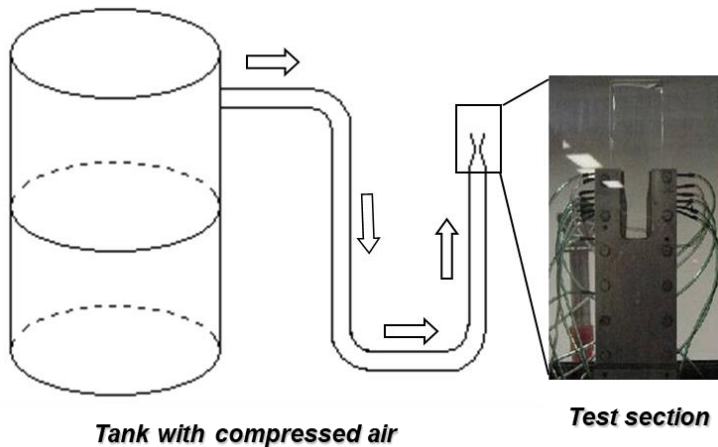
# Visualization of shock waves in a transonic/supersonic nozzle using Schlieren technique



Before turning on the Supersonic jet



After turning on the Supersonic jet



• Over-expanded flow

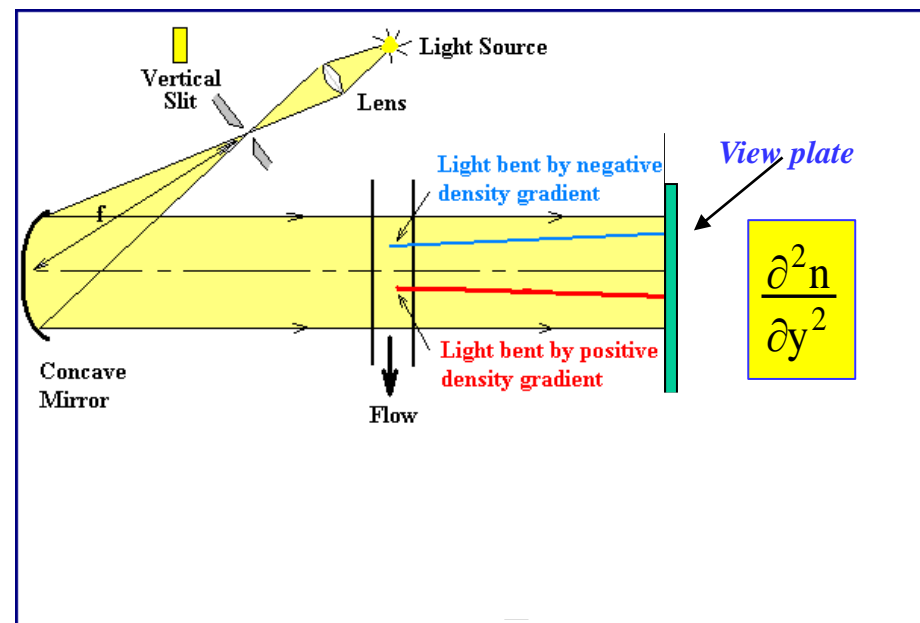
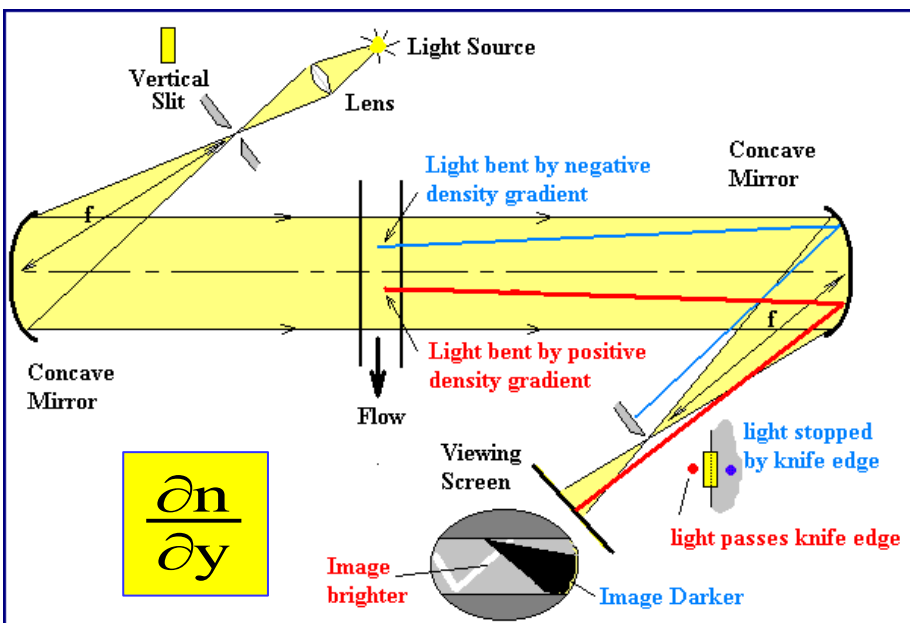
# COMPARISON OF SCHLIEREN VS. SHADOWGRAPH

## Schlieren:

- Displays a focused image
- Shows ray refraction angle,  $\epsilon$
- Contrast level responds to the 1<sup>st</sup> derivative of refractive index changes.
- Knife edge used for cutoff

## Shadowgraph:

- Displays a mere shadow
- Shows light ray displacement
- Contrast level responds to the 2<sup>nd</sup> derivative of refractive index changes.
- No knife edge used



# SCHLIEREN & SHADOWGRAPH FOR QUANTITATIVE MEASUREMENTS

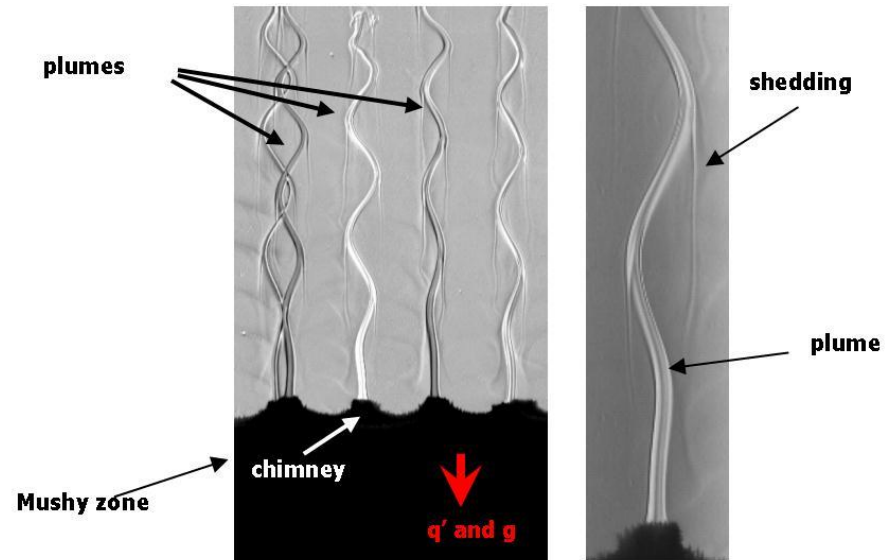
- *Application of the Schlieren and shadowgraph techniques:*
  - *Compressible flow with shock waves  $\Rightarrow$  density changes*
  - *Natural convective flow  $\Rightarrow$  density changes*
  - *Flame and combustion system:  $\Rightarrow$  density changes*
- *Temperature changes inside flows:*
  - *For low speed flow with heat transfer:*
  - *$P = \text{constant}$*

$$\rho = P / RT \Rightarrow \frac{\partial \rho}{\partial y} = \frac{P}{RT^2} \frac{\partial T}{\partial y} = \frac{\rho}{T} \frac{\partial T}{\partial y}$$

$$\Rightarrow \frac{\partial n}{\partial y} = \frac{n_0 - 1}{\rho_0} \frac{\partial \rho}{\partial y} = \frac{n_0 - 1}{T} \frac{\rho}{\rho_0} \frac{\partial T}{\partial y}$$

$$\Rightarrow \frac{\partial T}{\partial y} = \frac{T}{n_0 - 1} \frac{\rho_0}{\rho} \frac{\partial n}{\partial y}$$

$$\Rightarrow \frac{\partial^2 n}{\partial y^2} = \frac{n_0 - 1}{\rho_0} \left[ -\frac{\rho}{T} \frac{\partial^2 T}{\partial y^2} + \frac{2\rho}{T^2} \left( \frac{\partial T}{\partial y} \right)^2 \right]$$



# SCHLIEREN & SHADOWGRAPH FOR QUANTITATIVE DENSITY MEASUREMENT

- *Index of refraction is a function of thermodynamic state (density) for homogeneous medium:*

- *Lorenz-Lorentz relationship:* 
$$\frac{1}{\rho} \frac{n^2 - 1}{n^2 + 2} = \text{const}$$

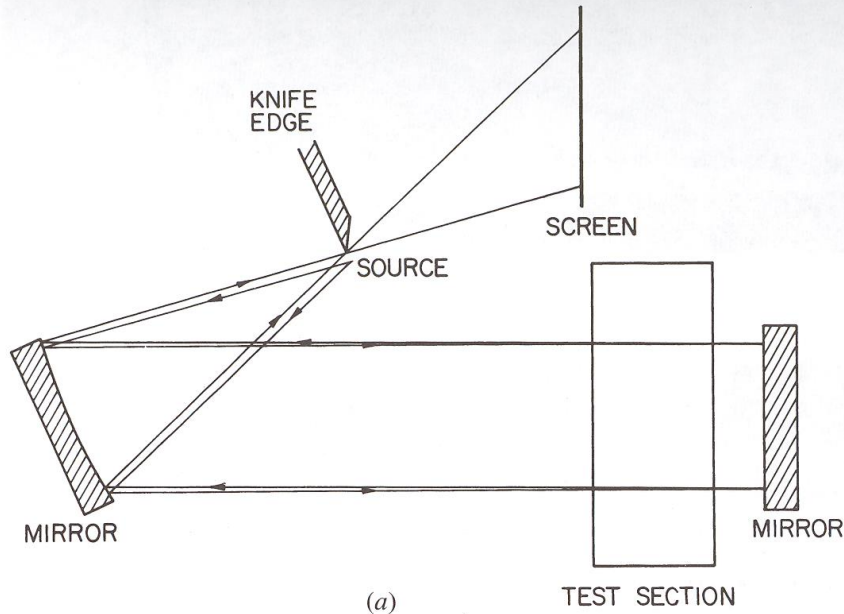
- *When  $n \approx 1$ , for gaseous flow:* 
$$\frac{n - 1}{\rho} = \text{const} \Rightarrow \rho = \frac{n - 1}{\text{const}}$$

- *at standard condition, with  $n_0$  and  $\rho_0$ :* 
$$\frac{n_0 - 1}{\rho_0} = \text{const} \Rightarrow n - 1 = \frac{\rho}{\rho_0} (n_0 - 1)$$
$$\Rightarrow \rho = \rho_0 \frac{n - 1}{n_0 - 1}$$

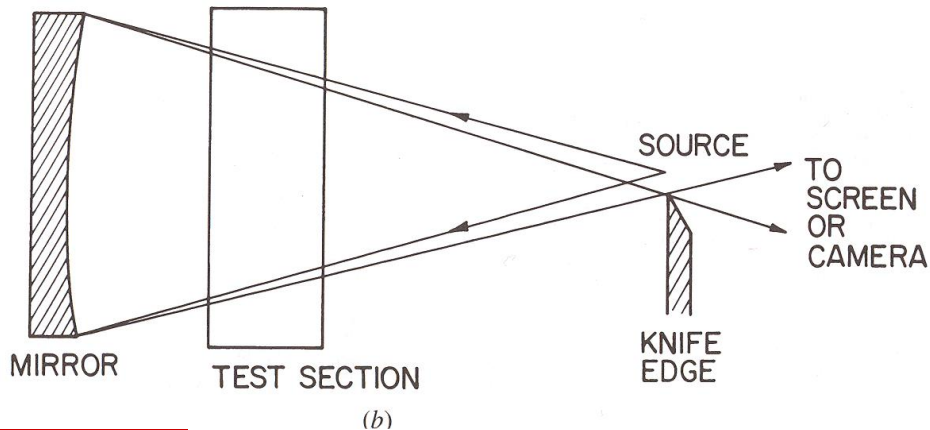
- *When first and second derivative is determined as in Schlieren and shadowgraph apparatus:*

$$\frac{\partial \rho}{\partial y} = \frac{1}{\text{const}} \frac{\partial n}{\partial y} \Rightarrow \frac{\partial \rho}{\partial y} = \frac{\rho_0}{n_0 - 1} \frac{\partial n}{\partial y}$$
$$\frac{\partial^2 \rho}{\partial y^2} = \frac{1}{\text{const}} \frac{\partial^2 n}{\partial y^2} \Rightarrow \frac{\partial^2 \rho}{\partial y^2} = \frac{\rho_0}{n_0 - 1} \frac{\partial^2 n}{\partial y^2}$$

# Alternative Schlieren system



*A. Setup with one converging and one plane mirror*



*A. Setup with one converging mirror*



# Holographic Schlieren system

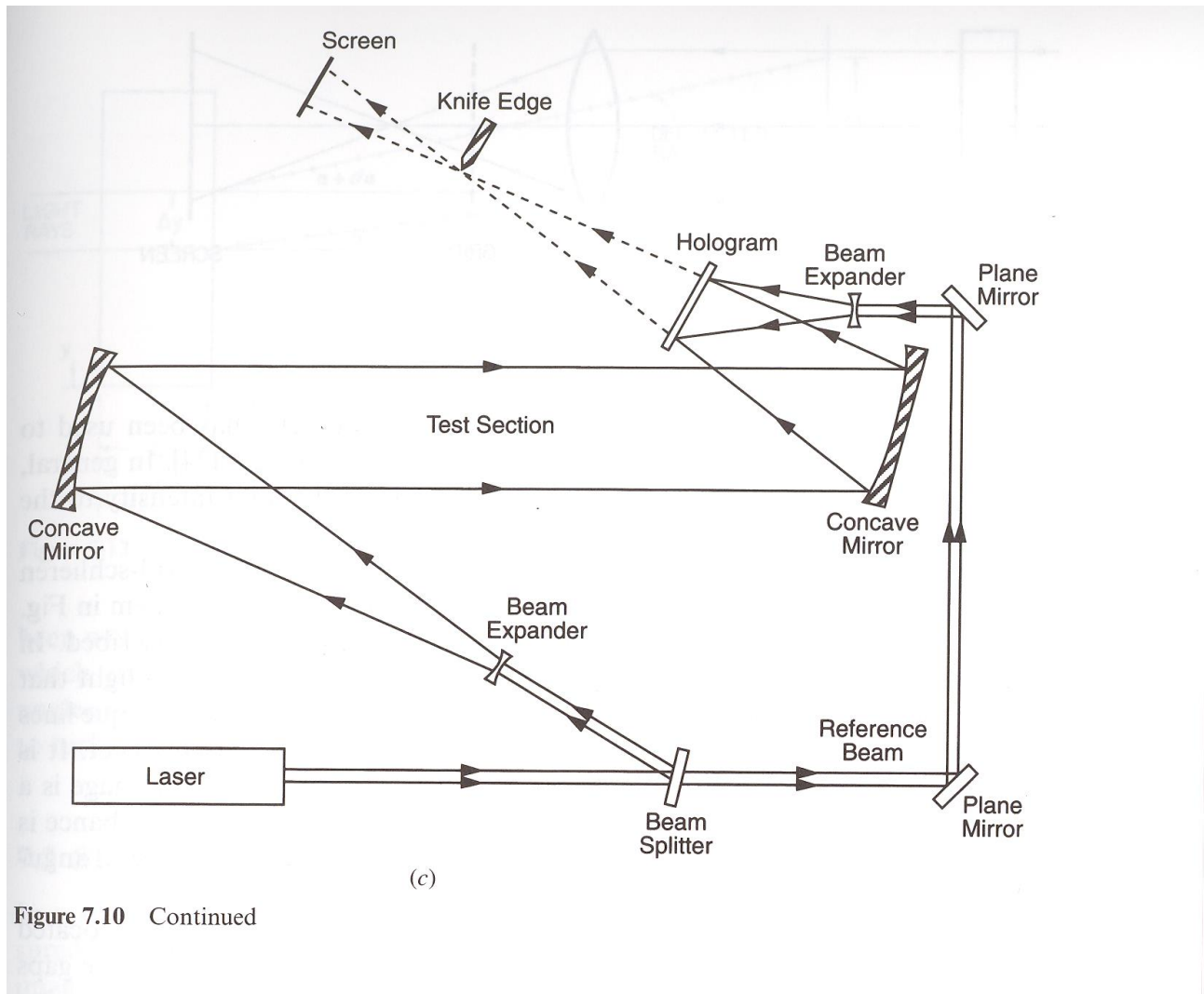
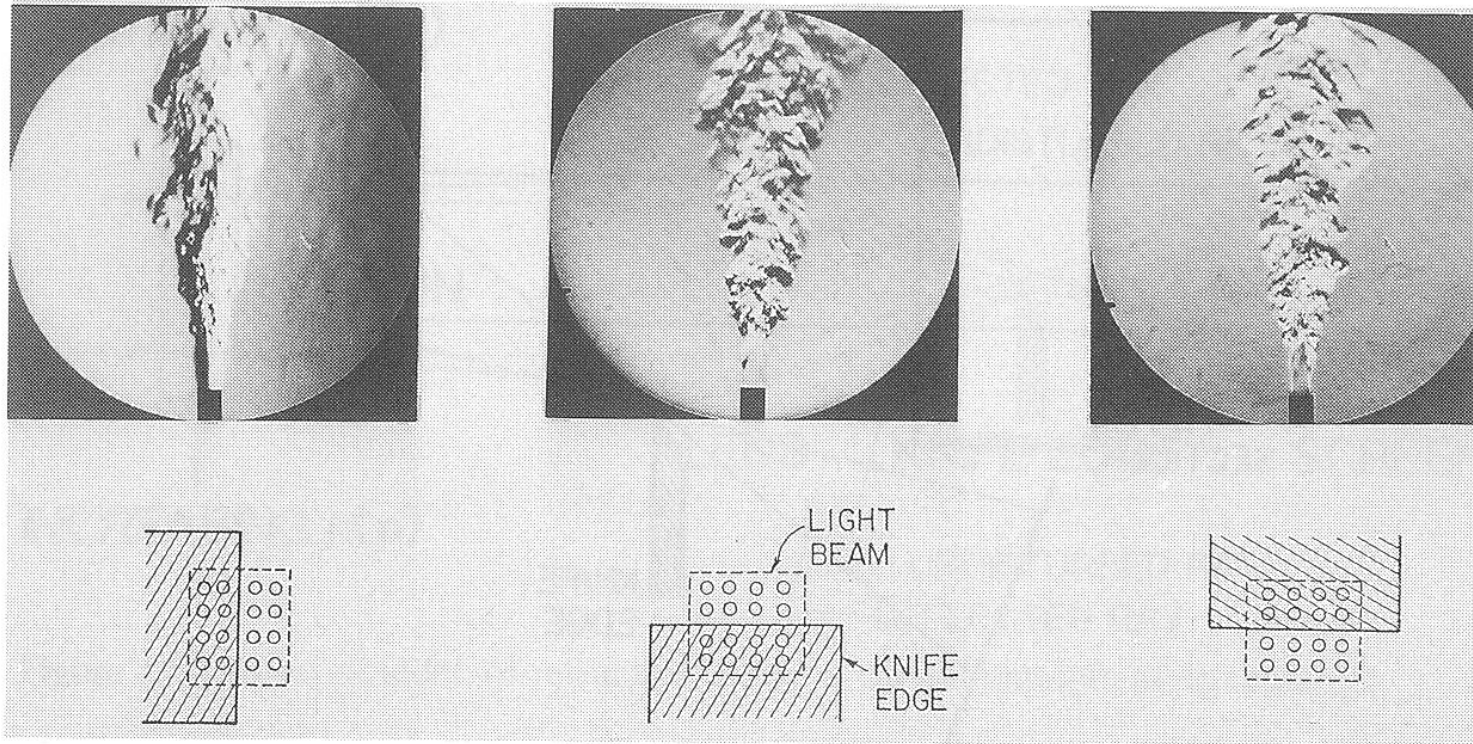


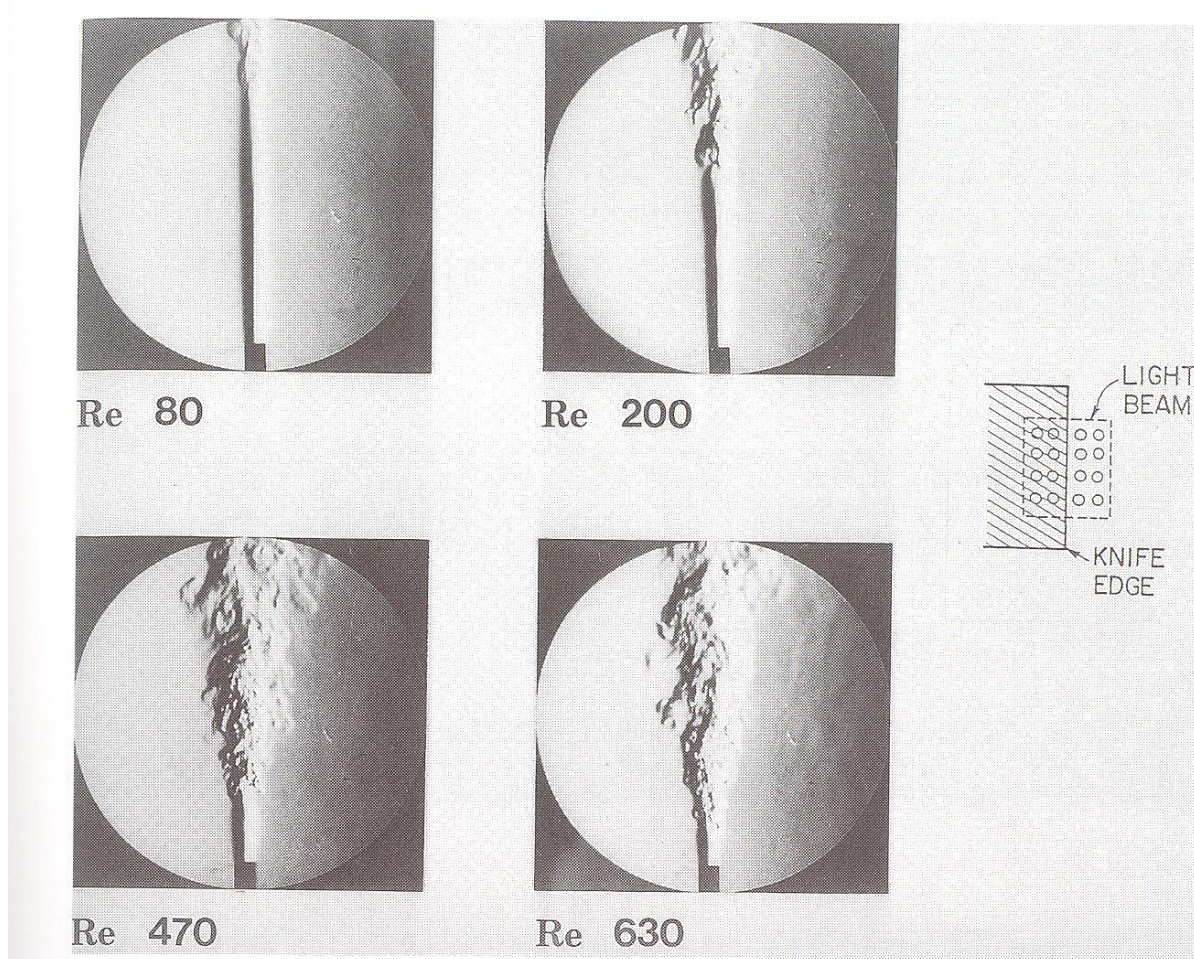
Figure 7.10 Continued

# Fundamentals of Schlieren System



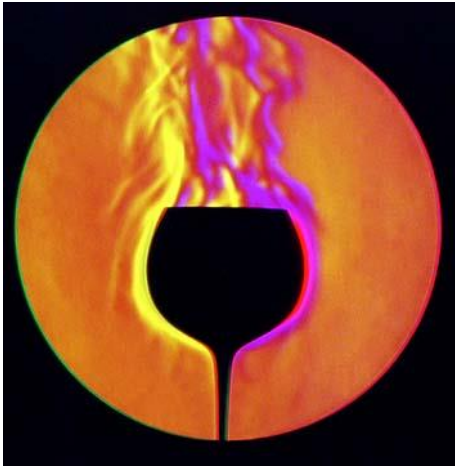
**Figure 7.7** Schlieren images of a helium jet entering an atmosphere of air: The effect of knife-edge orientation ( $Re = 630$ )

# Fundamentals of Schlieren System



**Figure 7.8** Schlieren images of the flow structure of a helium jet entering air at different numbers

# Examples: Shlieren Photography



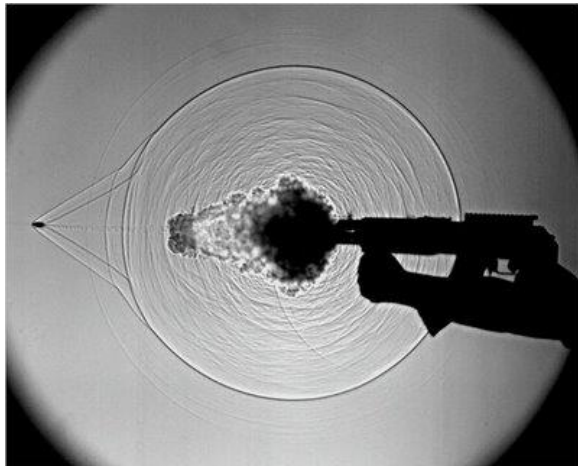
Warm water



A cough



A gas leak



The firing of an AK-47.



A simulated explosion in an airplane cabin.



Hair dryer

# Schlieren Application Examples

- *Seeing the Invisible: SLOW MOTION Schlieren Imaging results*  
<https://www.youtube.com/watch?v=4tgOyU34D44>

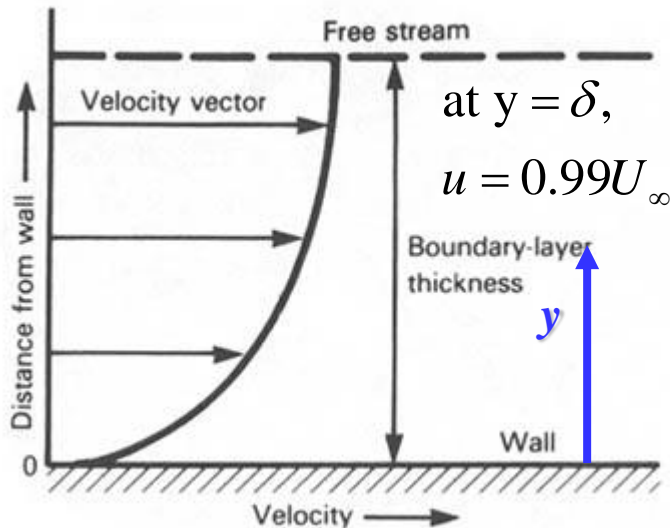
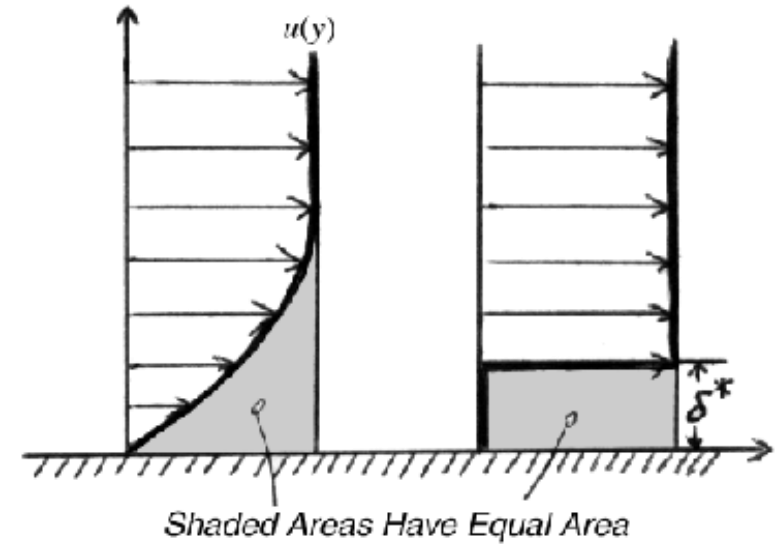
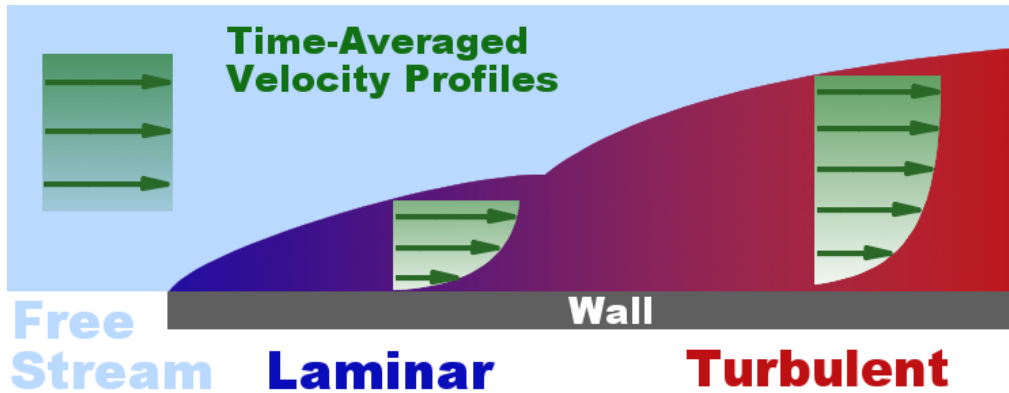


# Summary of the 1<sup>st</sup> Survey of AerE344 Course

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- **Group size:** *Most of the students prefer smaller group*
- **Lecture:** *Most of students think lectures are helpful*
  - *Sound of the embedded videos is not good...*
  - *Better linkage to the labs*
  - *More "Pre-Lab Videos" and more on the data analysis*
  - *On what the week's lab is about first, then tie it to the material next.*
- **Lab:** *Most of students feel AerE344 labs are helpful for you to better understanding the concepts and principles taught in AerE310 and AerE311.*
- **TAs:** *Most of students think the TAs are doing nice jobs.*
  - *Some TAs is restricted from offering as much help as they could, especially when it comes to Matlab.*
- **Other comments:**
  - *It would be nice to have more guidelines around lab writing and the coding.*
  - *What does the final exam look like*

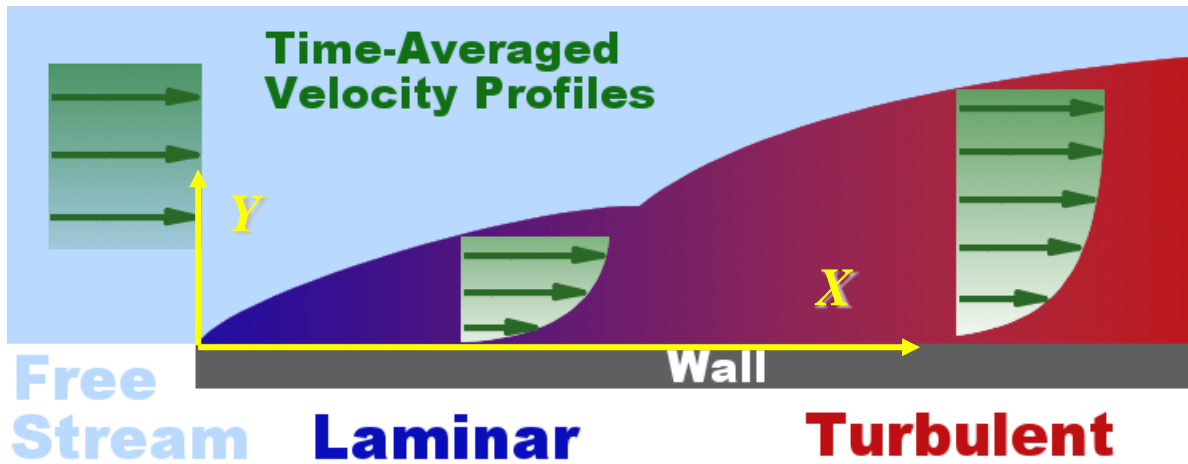
# AerE344 Lab#8: Measurements of Boundary Layer over a Flat Plate



*Displacement thickness:*  $\delta^* \equiv \int_0^\infty \left(1 - \frac{u}{U}\right) dy$

*Momentum thickness:*  $\theta \equiv \int_0^\infty \frac{u}{U} \left(1 - \frac{u}{U}\right) dy$

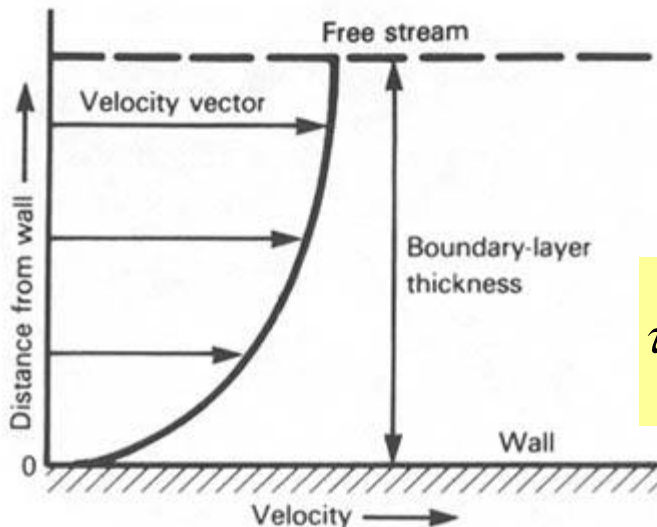
# AerE344 Lab#8: Measurements of Boundary Layer over a Flat Plate



$$\frac{\partial p}{\partial y} \approx 0$$

*Laminar boundary layer:*

*Turbulent boundary layer:*



$$\tau_w = \mu \left. \frac{\partial U}{\partial y} \right|_{wall}$$

$$Re_x = \frac{\rho U_\infty X}{\mu}$$

$$Re_x = \frac{\rho U_\infty X}{\mu}$$

$$C_f = \frac{1.328}{\sqrt{Re_x}}$$

$$C_f = \frac{0.074}{(Re_x)^{1/5}}$$

$$\delta = \frac{5.0X}{\sqrt{Re_x}}$$

$$\delta = \frac{0.37X}{(Re_x)^{1/5}}$$

$$\delta^* = \frac{1.72X}{\sqrt{Re_x}}$$

$$\theta = \frac{0.664X}{\sqrt{Re_x}}$$



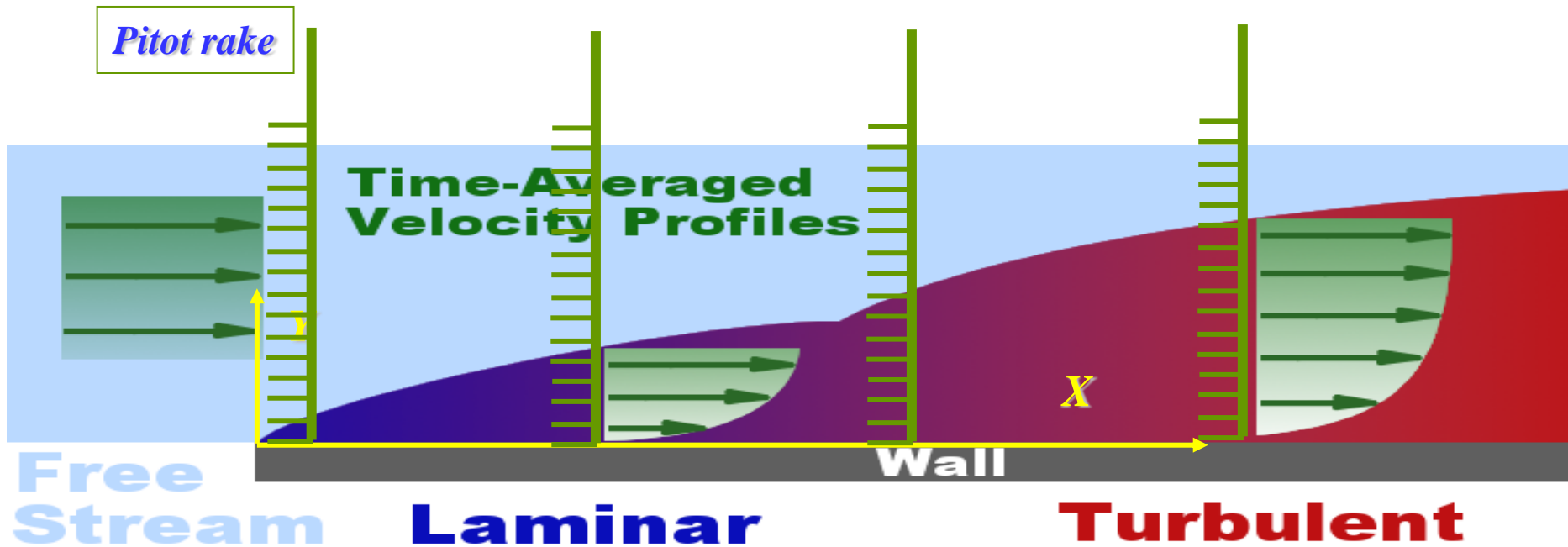
# AerE344L Lab#8: Measurements of Boundary Layer over a Flat Plate

- To conduct velocity profile measurements at 10 downstream locations.
- To determine boundary layer thickness and drag coefficient based on the velocity measurement results.

Displacement thickness:  $\delta^* \equiv \int_0^\infty \left(1 - \frac{u}{U}\right) dy$

Momentum thickness:  $\theta \equiv \int_0^\infty \frac{u}{U} \left(1 - \frac{u}{U}\right) dy$

Drag coefficient:  $C_d = 2 \frac{\theta}{L}$



# AerE344 Lab#8: Measurements of Boundary Layer over a Flat Plate

*Plots needed for lab report:*

- *Mean velocity profiles based on your measurements:  $U/U_\infty$  vs  $y/\delta$*
- *Plot the experimental values of  $\delta(x)$  and  $\theta(x)$ . The plot should also include comparison to the analytical expressions .*
- *Using the momentum thickness and the integral momentum equations to estimate:*
  - *Local shear stress coefficient  $C_f$  , as a function of  $x$ .*
  - *Find the drag coefficient  $C_d$*

