Lecture # 09:Conservation Equations of Mass andMomentum in Differential Form-P1

Dr. Hui HU

Department of Aerospace Engineering Iowa State University, 2251 Howe Hall, Ames, IA 50011-2271 Tel: 515-294-0094 / Email: <u>huhui@iastate.edu</u>



Reynolds Transport Theorem



Physical principle: Mass can be neither created or destroyed

$$m_{system} = constant$$

$$\left(\frac{dm}{dt}\right)_{system} = 0$$
with $m_{system} = \int_{system} dm = \int_{system} \rho d\forall$

$$\frac{DN_s}{Dt} = \frac{D\int \alpha \rho \ d\Psi}{Dt} = \frac{\partial}{\partial t} \int_{C.V.} \alpha \rho \ d\Psi + \int_{C.S.} (\alpha \rho \vec{V}) \bullet d\vec{A}$$



Make:
$$\alpha = 1$$
 $\frac{DM_s}{Dt} = \frac{\partial}{\partial t} \int_{C.V.} \rho \, d\Psi + \int_{C.S.} (\rho \vec{V}) \bullet d\vec{A} = 0$

• Integral form of the Mass Conservation Equation:

$$\frac{\partial}{\partial t} \int_{C.V.} \rho \, d\Psi + \int_{C.S.} (\rho \vec{V}) \bullet d\vec{A} = 0$$



Physical principle: Mass can be neither created or destroyed.

• Integral form:

$$\int_{C.\Psi} \frac{\partial \rho}{\partial t} \, d\Psi + \int_{C.S.} (\rho \vec{V}) \bullet d\vec{A} = 0$$



 Applying Guess divergence theorem, we convert the surface integral to volume integral to obtain:

$$\int_{C.\Psi.} \frac{\partial \rho}{\partial t} \, d\Psi + \int_{C.S.} (\rho \vec{V}) \bullet d\vec{A} = \int_{C.V.} \frac{\partial \rho}{\partial t} \, d\Psi + \int_{C.V.} \nabla \bullet (\rho \vec{V}) \, d\Psi = \int_{C.V.} \left[\frac{\partial \rho}{\partial t} + \nabla \bullet (\rho \vec{V}) \right] d\Psi = 0$$

• Differential form of the mass conservation equation (or continuity equation):

$$\frac{\partial \rho}{\partial t} + \nabla \bullet (\rho \vec{V}) = \frac{\partial \rho}{\partial t} + \vec{V} \bullet \nabla \rho + \rho \nabla \bullet \vec{V}$$
$$= \frac{D\rho}{Dt} + \rho \nabla \bullet \vec{V} = 0$$

Simplifications: Form incompressible flows: ρ is constant, then $\frac{\partial \rho}{\partial t} = 0;$ $\nabla \rho = 0$ Therefore, $\nabla \bullet \vec{V} = 0$



Example 01:

- The x-component velocity is given by u(x,y)=Ay²+C in an 2D incompressible flow.
 - Please determine y-component velocity v(x,y) if v(x,0)=0 as would be the case in flow between parallel plates.





Solution to Example 01:

$$\frac{D\rho}{Dt} + \rho \nabla \bullet \vec{V} = 0$$

$$\Rightarrow \nabla \bullet \vec{V} = 0$$

$$\Rightarrow \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$\Rightarrow \frac{\partial v}{\partial y} = -\frac{\partial u}{\partial x} = -\frac{\partial (Ay^2)}{\partial x} = 0$$

$$\frac{\partial y}{\partial y} = 0 \Rightarrow v(x, y) = f(x)$$

$$But \Rightarrow v(x, 0) = 0 \Rightarrow f(x) = 0$$

$$\Rightarrow v(x, y) = 0$$



$$u(y) = u_{\max} \left[1 - \left(\frac{y}{h}\right)^2 \right]$$

Aerospace Engineering

IOWA STATE UNIVERSITY Copyright © by Dr. Hui Hu @ Iowa State University. All Rights Reserved!

Example 02:

For a two-dimensional steady, inviscid, incompressible flow around a cylinder of radius of R as shown in the figure, the velocity field is given as :

$$\vec{V}(r,\theta) = U_{\infty}(1 - \frac{R^2}{r^2})\cos\theta \cdot \hat{e}_r - U_{\infty}(1 + \frac{R^2}{r^2})\sin\theta \cdot \hat{e}_{\theta};$$

where U_{∞} is the velocity of the undisturbed stream (therefore, U_{∞} is constant).

Is the flow with the velocity field given above physically possible?

$$\frac{\partial \rho}{\partial t} + \nabla \bullet (\rho \vec{V}) = \frac{\partial \rho}{\partial t} + \vec{V} \bullet \nabla \rho + \rho \nabla \bullet \vec{V}$$
$$= \frac{D\rho}{Dt} + \rho \nabla \bullet \vec{V} = 0$$

$$\nabla \bullet \vec{V} = \frac{1}{h_1 h_2 h_3} \left[\frac{\partial (h_2 h_3 V_1)}{\partial q_1} + \frac{\partial (h_1 h_3 V_2)}{\partial q_2} + \frac{\partial (h_1 h_2 V_3)}{\partial q_3} \right]$$



For a two-dimensional steady, inviscid, incompressible flow around a cylinder of radius of R as shown in the figure, the velocity field is given as :

$$\vec{V}(r,\theta) = U_{\infty}(1 - \frac{R^2}{r^2})\cos\theta \cdot \hat{e}_r - U_{\infty}(1 + \frac{R^2}{r^2})\sin\theta \cdot \hat{e}_{\theta};$$

where U_{∞} is the velocity of the undisturbed stream (therefore, U_{∞} is constant).

Is the flow with the velocity field given above physically possible?

For a steady, inviscid, incompressible flow, the mass conservation equation will be: $\nabla \cdot \vec{V} = 0$ Expend it in the cylindrical system, it will be:

$$\frac{1}{r} \left[\frac{\partial (rV_r)}{\partial r} + \frac{\partial V_{\theta}}{\partial \theta} + \frac{\partial (rV_z)}{\partial z} \right] = 0 \quad \Rightarrow \quad \frac{\partial (rV_r)}{\partial r} + \frac{\partial V_{\theta}}{\partial \theta} + \frac{\partial (rV_z)}{\partial z} = 0$$

For the velocity field given above:

$$V_r = U_{\infty} \left(1 - \frac{R^2}{r^2}\right) \cos \theta; \quad V_{\theta} = -U_{\infty} \left(1 + \frac{R^2}{r^2}\right) \sin \theta; \quad V_z = 0$$

Since

$$\frac{\partial(rV_r)}{\partial r} = \frac{\partial}{\partial r} [U_{\infty}(r - \frac{R^2}{r})\cos\theta] = U_{\infty}(1 + \frac{R^2}{r^2})\cos\theta$$
$$\frac{\partial V_{\theta}}{\partial \theta} = \frac{\partial}{\partial \theta} [-U_{\infty}(1 + \frac{R^2}{r^2})\sin\theta] = -U_{\infty}(1 + \frac{R^2}{r^2})\cos\theta$$
$$\frac{\partial(rV_z)}{\partial z} = 0$$

Therefore:

$$\frac{\partial (rV_r)}{\partial r} + \frac{\partial V_{\theta}}{\partial \theta} + \frac{\partial (rV_z)}{\partial z} = U_{\infty}(1 + \frac{R^2}{r^2})\cos\theta - U_{\infty}(1 + \frac{R^2}{r^2})\cos\theta + 0 = 0$$

The velocity satisfies the mass conservation equation, i. e., with the velocity field given above is physically possible!



$$\nabla \bullet \vec{V} = \frac{1}{h_1 h_2 h_3} \left[\frac{\partial (h_2 h_3 V_1)}{\partial q_1} + \frac{\partial (h_1 h_3 V_2)}{\partial q_2} + \frac{\partial (h_1 h_2 V_3)}{\partial q_3} \right]$$

$$\frac{R^2}{r^2}$$
) cos θ

0

Conservation of Momentum

Newton's second law states that:

[Time change rate of momentum of a system] = [Resultant external force acting on the system]

$$\frac{d\vec{M}_{S}}{dt} = \sum \vec{F}_{S} = \sum \vec{F}_{Surface} + \sum \vec{F}_{body}$$

Reynolds Transport Theorem:

Λ

$$\frac{dN_s}{dt}\Big|_{system} = \frac{d\int \alpha \rho \, d\Psi}{Dt} = \frac{\partial}{\partial t} \int_{C.V.} \alpha \rho \, d\Psi + \int_{C.S.} (\alpha \rho \vec{V}) \bullet d\vec{A}$$

Make: $\alpha = \vec{V} \quad \frac{DM_s}{Dt}\Big|_{system} = \frac{\partial}{\partial t} \int_{C.V.} \rho \vec{V} \, d\Psi + \int_{C.S.} (\vec{V} \rho \vec{V}) \bullet d\vec{A} = 0$

Integral form of the Mass Conservation Equation:

$$\frac{\partial}{\partial t} \int_{C.V.} \vec{V} \rho \, d\Psi + \int_{C.S.} (\vec{V} \rho \vec{V}) \bullet d\vec{A} = \vec{F}_{surface} + \vec{F}_{body}$$





Conservation of Momentum

 $ec{F}_{\it surface}$

 \vec{F}_{bod}

$$\frac{\partial}{\partial t} \int_{C.V.} \vec{V} \rho \, d\Psi + \int_{C.S.} (\vec{V} \rho \vec{V}) \bullet d\vec{A} = \vec{F}_{surface} + \vec{F}_{body}$$

Surface forces such as pressure and shear stress. The surface forces usually can be expressed as $\vec{F}_{surface} = \int_{C.S.} \widetilde{P} \cdot d\vec{A}$, where \widetilde{P} is the stress tensor exerted by the surroundings on the particle surface. $\widetilde{P} = -P\widetilde{I} + \widetilde{\tau}$



--- System boundary at time $t + \delta t$

Body forces such as electromagnetic, gravitational forces. <u>Usually</u> the body force can be expressed as $\vec{F}_{body} = \int_{C.V.} \rho \vec{f} dV$, where \vec{f} is a vector which references the resultant force per unit mass.

$$\frac{\partial}{\partial t} \int_{C.V.} \rho \vec{V} \, d\Psi + \int_{C.S.} (\rho \vec{V} \vec{V}) \bullet d\vec{A} = \int_{C.S.} \widetilde{P} \bullet d\vec{A} + \int_{C.V.} \rho \vec{f} \, d\Psi$$





IOWA STATE UNIVERSITY Copyright © by Dr. Hui Hu @ Iowa State University. All Rights Reserved!

Conservation of Momentum

 Using divergence theorem for the control surface integrals, we obtained following equation after noting that the limits do not change.

$$\frac{\partial}{\partial t} \int_{C.V.} \rho \vec{V} \, d\Psi + \int_{C.S.} \nabla \bullet (\rho \vec{V} \vec{V}) \, d\Psi = \int_{C.S.} \nabla \bullet \tilde{P} \, d\Psi + \int_{C.V.} \rho \, \vec{f} \, d\Psi$$
$$\Rightarrow \int_{C.V.} \left[\frac{\partial (\rho \vec{V})}{\partial t} + \nabla \bullet (\rho \vec{V} \vec{V}) - \nabla \bullet \tilde{P} - \rho \, \vec{f} \, \right] d\Psi = 0$$
$$\Rightarrow \frac{\partial (\rho \vec{V})}{\partial t} + \nabla \bullet (\rho \vec{V} \vec{V}) - \nabla \bullet \tilde{P} - \rho \, \vec{f} = 0$$

• Expand the above equation using $\nabla \bullet (\phi \vec{A}) = (\vec{A} \bullet \nabla)\phi + \phi \nabla \bullet \vec{A}$

$$= (\vec{A} \bullet \nabla)\phi + \phi \nabla \bullet \vec{A}$$

Control volume

$$\frac{\partial(\rho V)}{\partial t} + \nabla \bullet (\rho \vec{V} \vec{V}) - \nabla \bullet \tilde{P} - \rho \vec{f} = 0$$

$$\Rightarrow \vec{V} \frac{\partial \rho}{\partial t} + \rho \frac{\partial \vec{V}}{\partial t} + \vec{V} \nabla \bullet (\rho \vec{V}) + (\rho \vec{V} \bullet \nabla) \vec{V} - \nabla \bullet \tilde{P} - \rho \vec{f} = 0$$

$$\Rightarrow \vec{V} [\frac{\partial \rho}{\partial t} + \nabla \bullet (\rho \vec{V})] + \rho \frac{\partial \vec{V}}{\partial t} + (\rho \vec{V} \bullet \nabla) \vec{V} - \nabla \bullet \tilde{P} - \rho \vec{f} = 0$$

$$\Rightarrow \rho \frac{\partial V}{\partial t} + (\rho \vec{V} \bullet \nabla) \vec{V} - \nabla \bullet \vec{P} - \rho \vec{f} = 0$$

$$\Rightarrow \rho \frac{D\vec{V}}{Dt} - \nabla \bullet \vec{P} - \rho \vec{f} = 0$$

 The differential form of the momentum equation is:

$$\rho \frac{D\vec{V}}{Dt} - \nabla \bullet \vec{P} - \rho \vec{f} = 0$$

The Navier-Stokes Equations

$$\frac{\partial(\rho\vec{V})}{\partial t} + \nabla \bullet (\rho \vec{V} \vec{V}) - \nabla \bullet \widetilde{P} - \rho \vec{f} = 0$$

Where
$$\widetilde{P} = -P\widetilde{I} + \widetilde{\tau}$$
, and tensor $\widetilde{I} = \begin{vmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{vmatrix}$; and $\widetilde{\tau} = \begin{vmatrix} \sigma_{xx} & \tau_{yx} & \tau_{zx} \\ \tau_{xy} & \sigma_{yy} & \tau_{zy} \\ \tau_{xz} & \tau_{yz} & \sigma_{zz} \end{vmatrix}$

Re-writing the equation after substitution leads to:

$$\begin{aligned} \frac{\partial(\rho\vec{V})}{\partial t} + \nabla \bullet (\rho\vec{V}\vec{V}) - \nabla \bullet (-P\widetilde{I} + \widetilde{\tau}) - \rho \vec{f} &= 0 \\ \frac{\partial(\rho\vec{V})}{\partial t} + \nabla \bullet (\rho\vec{V}\vec{V} + P\widetilde{I} - \widetilde{\tau}) - \rho \vec{f} &= 0 \end{aligned}$$

Since $\nabla \bullet (P\widetilde{I}) = P\nabla \bullet \widetilde{I} + (\widetilde{I} \bullet \nabla)P = \nabla P$

Therefore:

$$\frac{\partial(\rho \vec{V})}{\partial t} + \nabla \bullet (\rho \vec{V} \vec{V}) + \nabla P - \nabla \bullet \tilde{\tau} - \rho \vec{f} = 0$$



The Navier-Stokes Equations



The Navier-Stokes Equations

$$\rho \frac{D\vec{V}}{Dt} + \nabla P - \nabla \bullet \tilde{\tau} - \rho \, \vec{f} = 0$$

Stress Tensor

The stress tensor has nine components:

$$\widetilde{\tau} = \begin{vmatrix} \sigma_{xx} & \tau_{yx} & \tau_{zx} \\ \tau_{xy} & \sigma_{yy} & \tau_{zy} \\ \tau_{xz} & \tau_{yz} & \sigma_{zz} \end{vmatrix}$$

Newtonian fluid,

$$\widetilde{\tau} = \mu [\nabla \vec{V} + (\nabla \vec{V})^T - \frac{2}{3} (\nabla \bullet \vec{V}) \widetilde{I}]$$

For incompressible $\underline{flow, in}$ Cartesian coordinate system

$$\tau_{ij} = \mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

$$\tau_{xy} = \tau_{yx}; \qquad \tau_{xz} = \tau_{zx} \qquad \tau_{zy} = \tau_{yz}$$



• Newtonian and non-Newtonian fluids