

# **LECTURE 10: PRESSURE SENSITIVE PAINT (PSP) & TEMPERATURE SENSITIVE PAINT (TSP) - PART 01**

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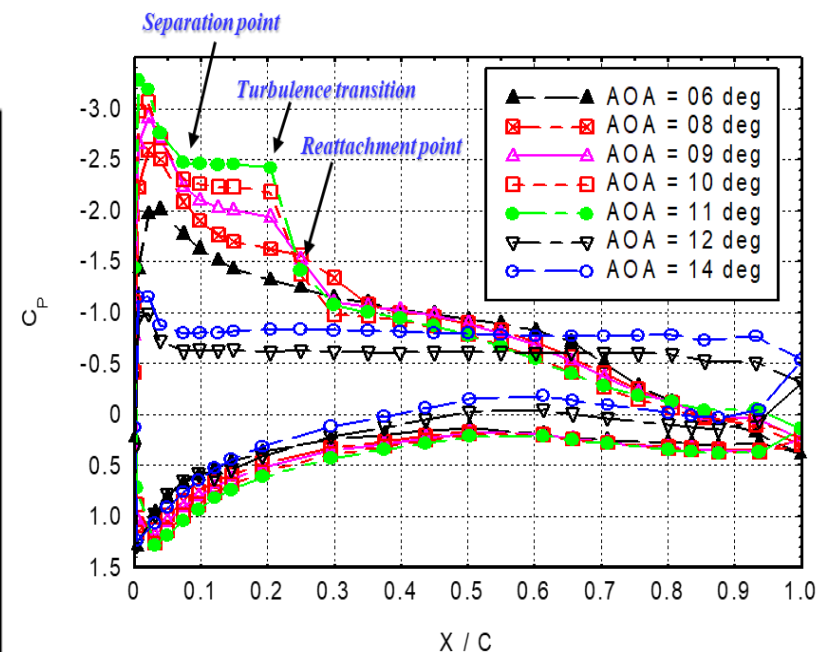
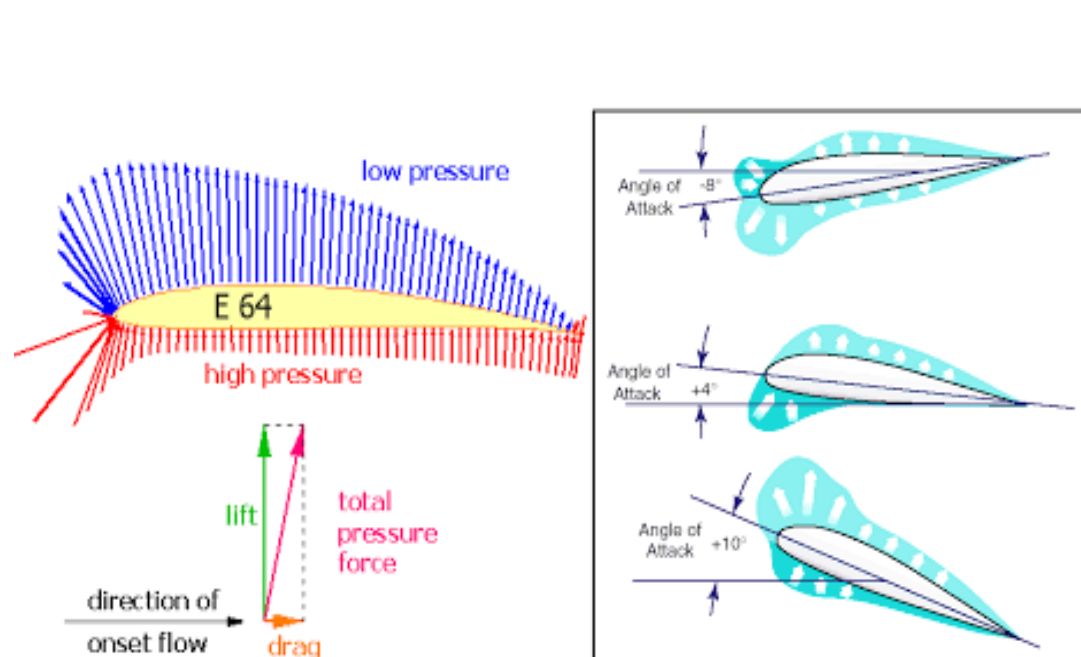
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# INTRODUCTION

- ❖ *Pressure measurements are the primary measurements made in most practical aerodynamic testing or basic fluid mechanics experiments.*
- ❖ *Surface pressure measurements are used for:*
  - ❖ *Identifying specific flow phenomena (boundary layer separation, shock wave impingement, etc.) that are not easily measured by “standard pressure tap” measurements.*
  - ❖ *Validation of computational codes*
  - ❖ *Loads calculations by integration of the surface pressures*

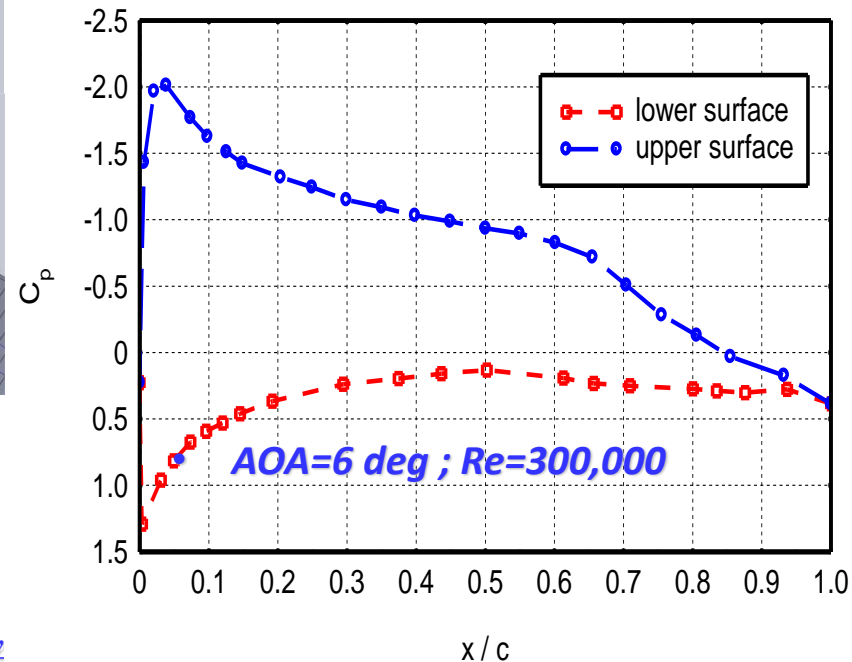


# INTRODUCTION

- ❑ **Conventional pressure measurements: Transducers or taps**
  - Discrete pre-determined locations
  - Very high accuracy ( $< 0.05\%$  FS)
  - Well understood with long testing background
  - High data rate with scanned systems (1000+)
  - Limitations to where they can be installed
  - Potential effect on the flow field – intrusive measurements
  - Expensive installation costs



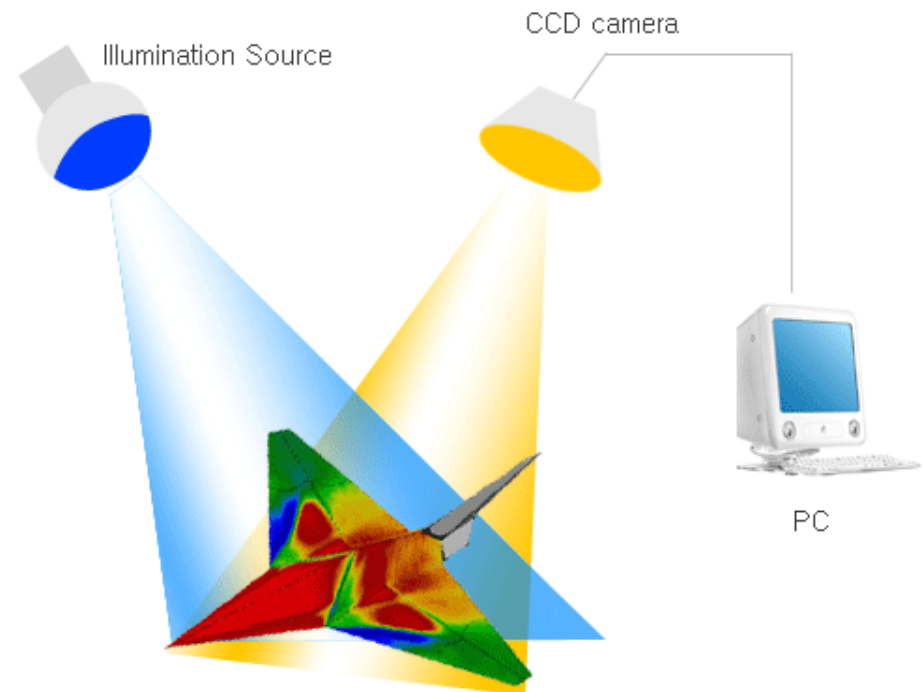
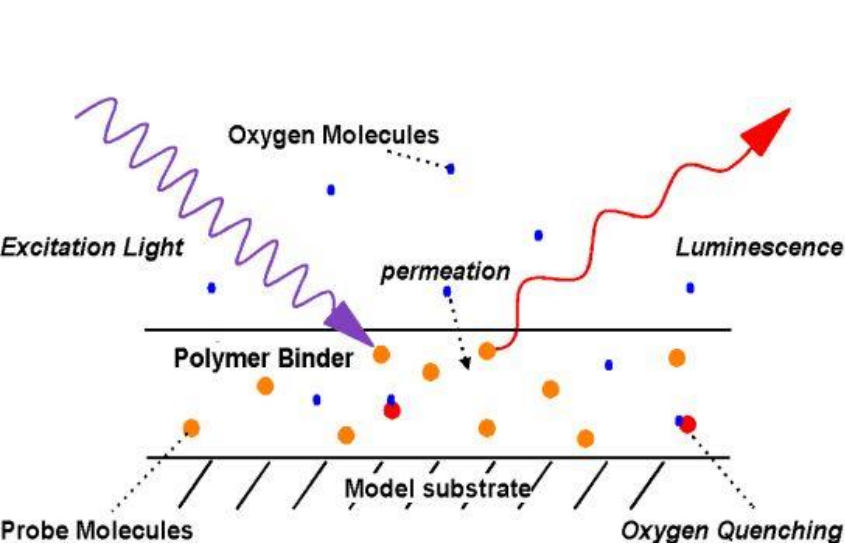
DSA3217 (Shown)



by Dr. Hui Hu @ Iowa

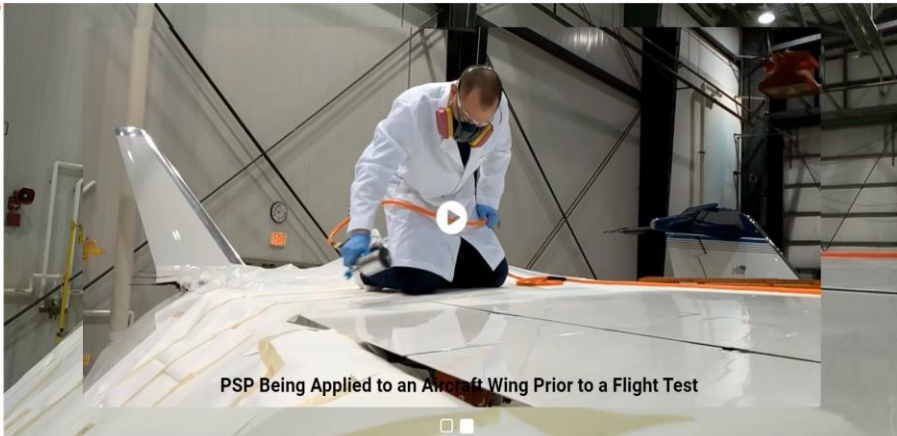
# □ PRESSURE SENSITIVE PAINT (PSP)

- *Sprayed over entire exterior surface*
- *non-intrusive pressure measurements*
- *high spatial resolution with resolution limited only by detection system*
- *Limited to optical access applications*
- *Inexpensive application costs*
- *Relatively expensive initial costs to setup the system*
- *High-speed applications*
- *Newer method that is still being fully explored for low-speed applications.*

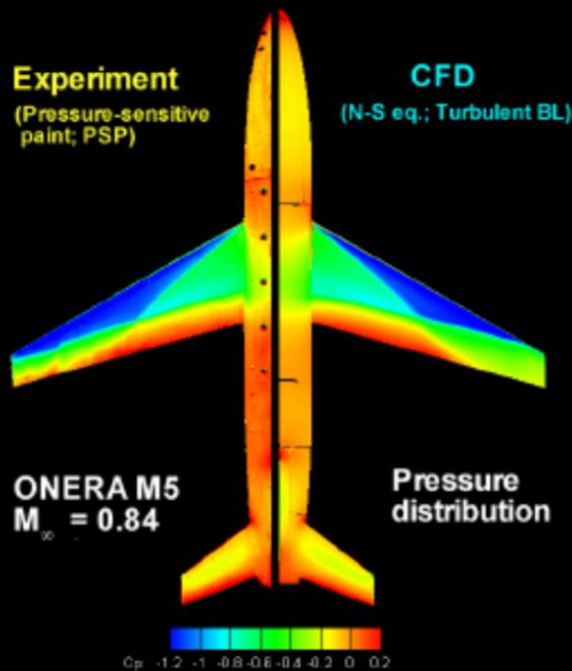




# □ PRESSURE SENSITIVE PAINT (PSP) TECHNIQUE



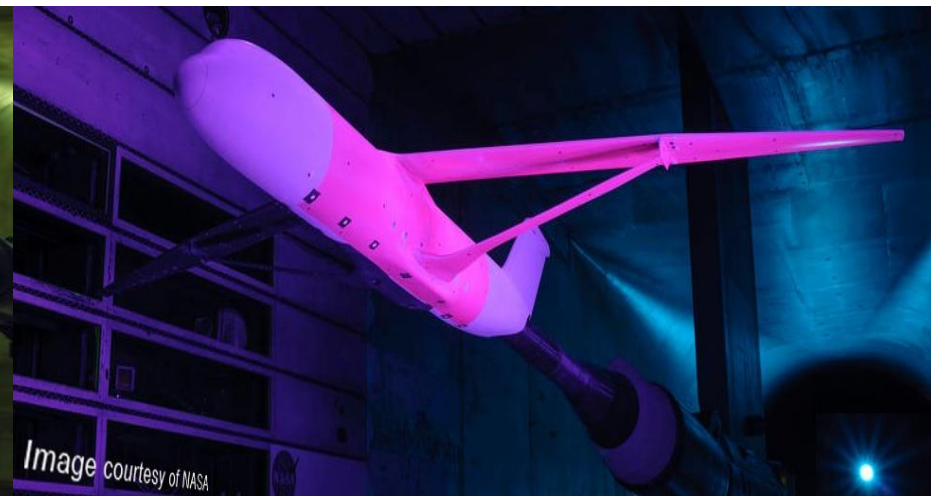
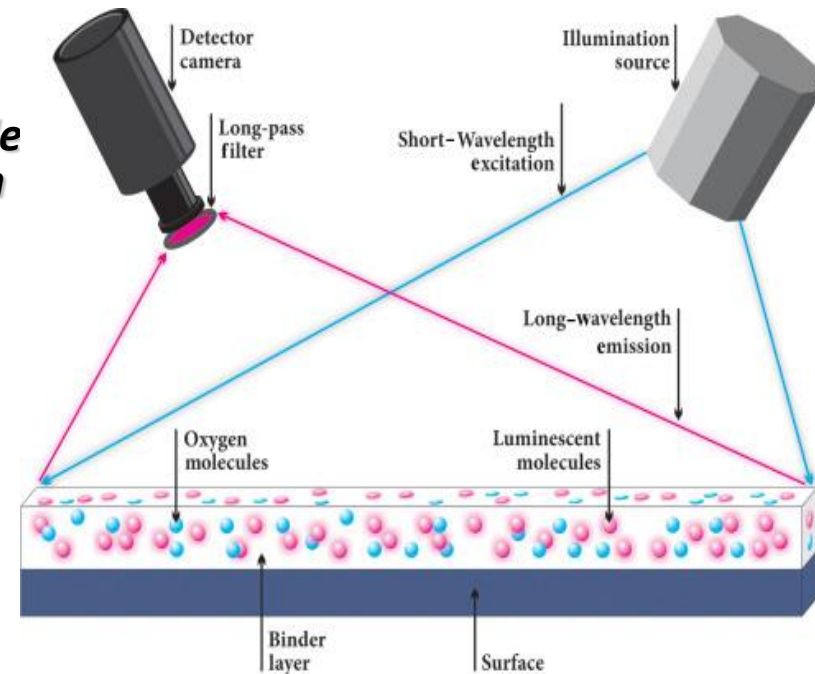
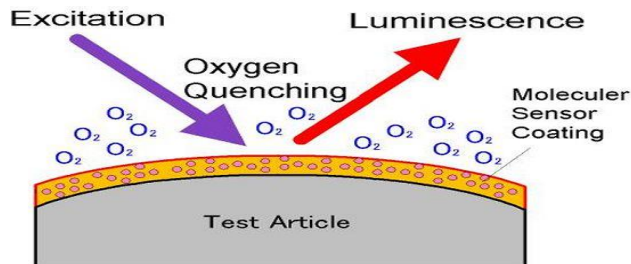
- <https://innssi.com/psp/>



- <https://www.youtube.com/watch?v=cWKUZJroVN8>

# ❑ BASIC PRINCIPLES OF PRESSURE SENSITIVE PAINT (PSP)

- **Composition of Air:** 78.08%  $N_2$ , 20.95%  $O_2$ , 0.93% Ar, 0.03%  $CO_2$ , 0.002% Ne, plus lesser amounts of Methane, Helium, Krypton, Hydrogen, Xenon.
- **The pressure of air can be determined if the particle pressure of oxygen (i.e., oxygen concentration) can be measured.**
- **A typical pressure sensitive paint is comprised of two main parts: an oxygen sensitive fluorescent molecule and an oxygen permeable binder**



# ❑ BASIC PRINCIPLES OF PRESSURE SENSITIVE PAINT (PSP)

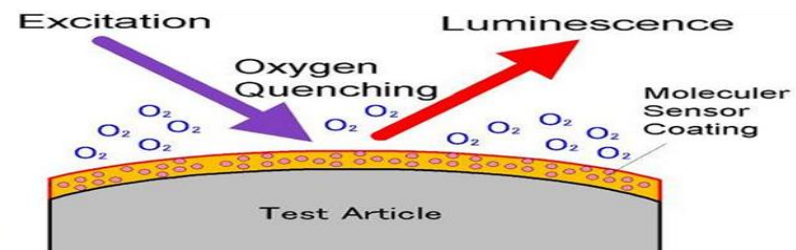
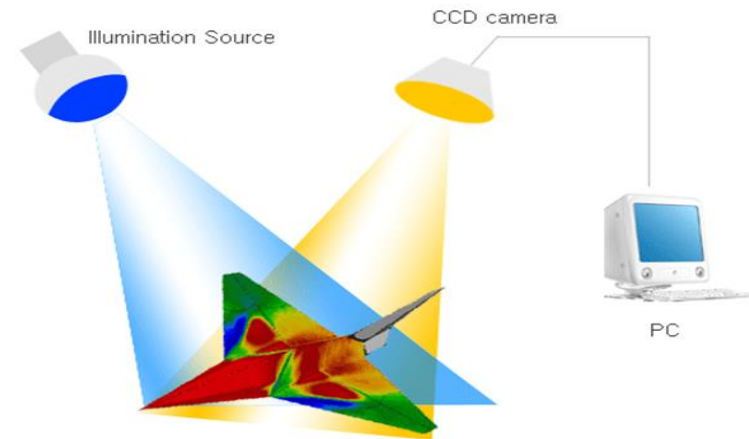
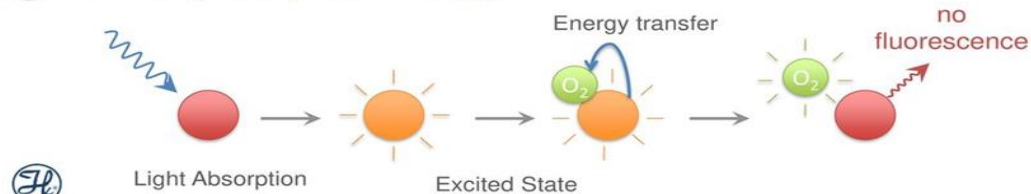
- **The pressure sensitive paint method is based on the sensitivity of certain luminescent molecules to the presence of oxygen.**
  - **When a luminescent molecule absorbs a photon, it is excited to an upper singlet energy state. The molecule then typically recovers to the ground state by the emission of a photon of a longer wavelength (i.e., fluorescence or phosphorescence ).**
  - **In some materials, oxygen can interact with the molecule so that the transition to the ground state is radiationless, this process is known as **oxygen quenching**.**
  - **The rate at which these two processes compete is dependent on the partial pressure of oxygen present, with a higher oxygen pressure quenching the molecule more, thus giving off a lower intensity of light.**

## Fluorescence Quenching by Oxygen

### 1 Fluorescence process (without oxygen)



### 2 Quenching in the presence of oxygen



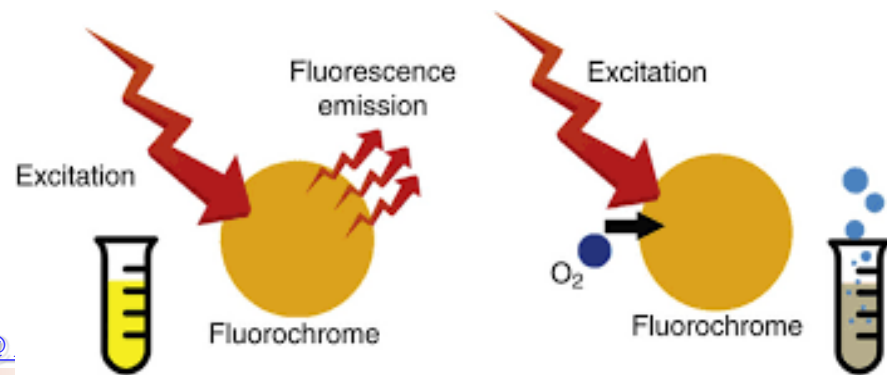
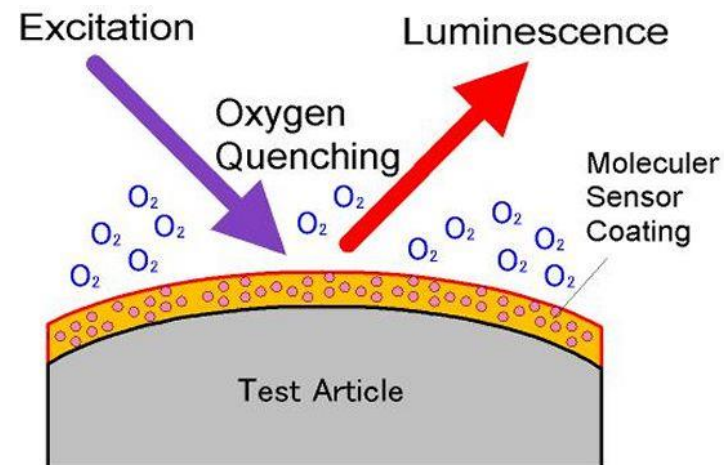
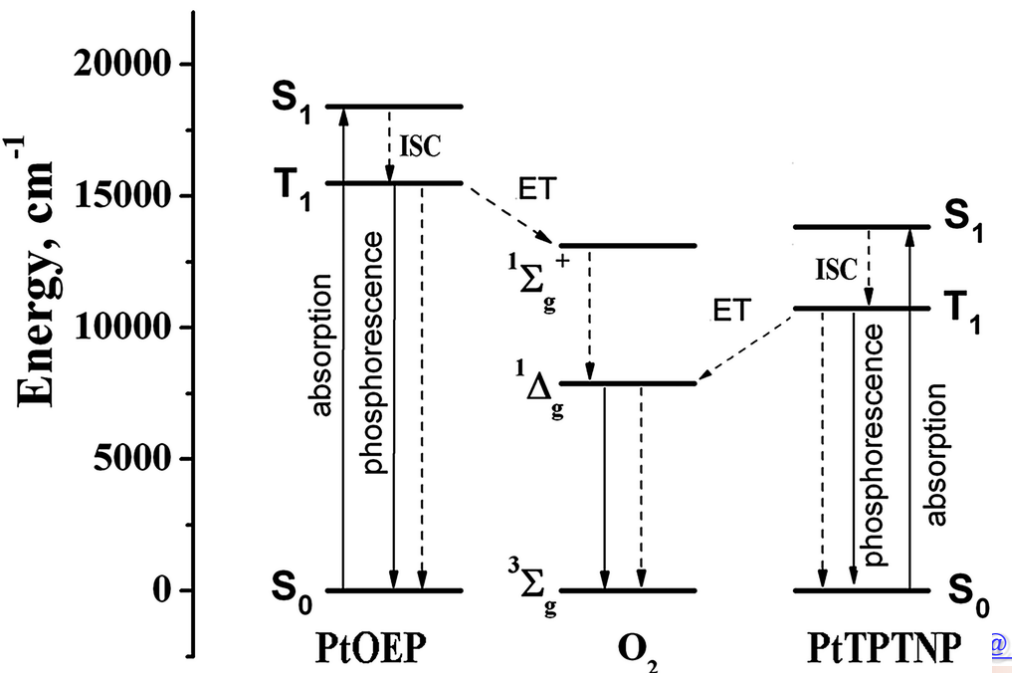


# ❑ BASIC PRINCIPLES OF PRESSURE SENSITIVE PAINT (PSP)

- For oxygen quenching, the intensity decrease can be described by the well-known **Stern-Volmer equation**:

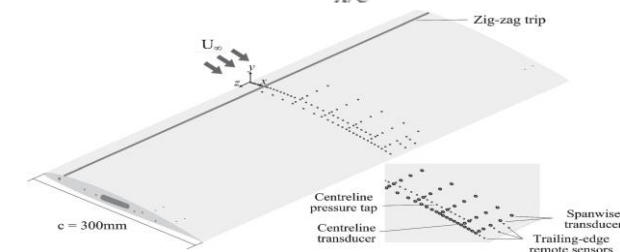
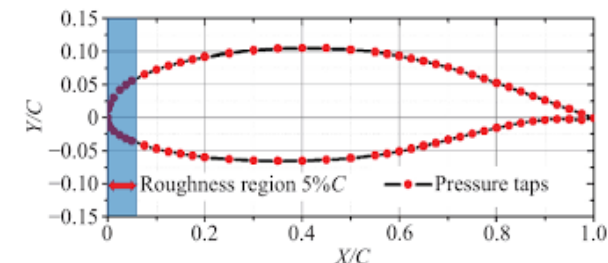
$$\frac{\tau_0}{\tau} = 1 + K_{SV}Q \quad \text{or} \quad \frac{\tau_0}{\tau_{O_2}} = \frac{I_0}{I_{O_2}} = 1 + K_{SV}P_{O_2}$$

- $\tau$  is the lifetime,  $I$  is the intensity
- $K_{SV}$  is the Stern-Volmer constant
- $Q$  is the quencher or partial pressure of oxygen



# ❏ ADVANTAGES OF PRESSURE SENSITIVE PAINT (PSP)

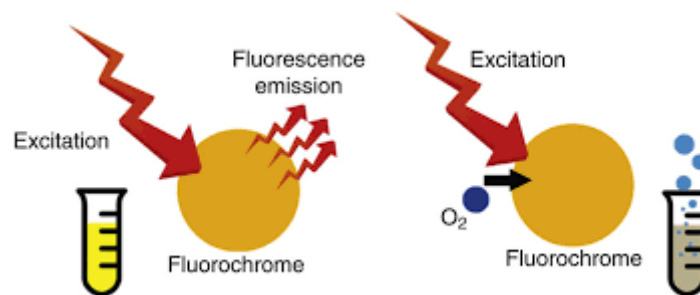
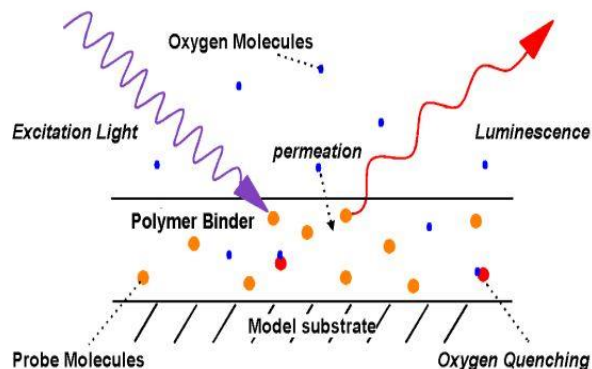
- **Pressure sensitive paint has numerous advantages over conventional pressure taps and transducers.**
  - **PSP is a field measurement technique, allowing for a surface pressure determination over the entire model.**
  - **PSP provides a much greater spatial resolution than pressure taps, and disturbances in the flow are immediately observable.**
  - **Use of PSP does not affect the flow around the model, allowing its use over the entire model surface.**
  - **The use of PSP eliminates the need for a large number of pressure taps, leading to less time and money for surface pressure measurements**
  - **Since holes do not need to be drilled in the model for the installation of pressure taps, the model strength is increased, and higher Reynolds numbers can be obtained.**
  - **Not only does the PSP method reduce the cost of the model construction, but it also reduces the cost of the instrumentation needed for data collection.**
  - **The equipment needed for PSP costs less than pressure taps, but it can also be easily reused for numerous models.**
- **PSP has the potential to save both time and money in aircraft design.**
  - **The continuous data distribution on the model provided by PSP can easily be integrated over specific components, which can provide detailed surface loads.**
  - **Since a model for use with the PSP technique is faster to construct, this allows for load data to be known much earlier in the design process.**



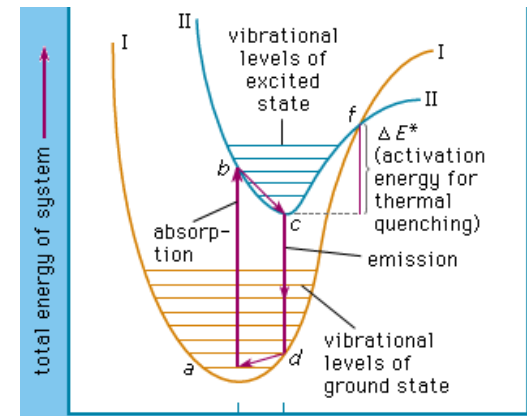


# ❑ DISADVANTAGES OF PRESSURE SENSITIVE PAINT (PSP)

- One of these characteristics is that the *response of the luminescent molecules in the PSP coating degrades with time of exposure to the excitation illumination*.
  - This degradation occurs because of a photochemical reaction when the molecules are excited.
  - Eventually, this degradation of the molecules determines the useful life of the PSP coating.
  - This characteristic becomes more important for larger models, as the cost and time of PSP reapplication becomes a significant factor.
- A second undesirable characteristic of PSP is that the emission intensity is *affected by the local temperature*.
  - This behavior is due to the effect of temperature has on the energy state of the luminescent molecules, and the oxygen permeability of the binder.
  - This temperature dependence becomes even more significant in compressible flow tests, where the recovery temperature over the model surface is not uniform.



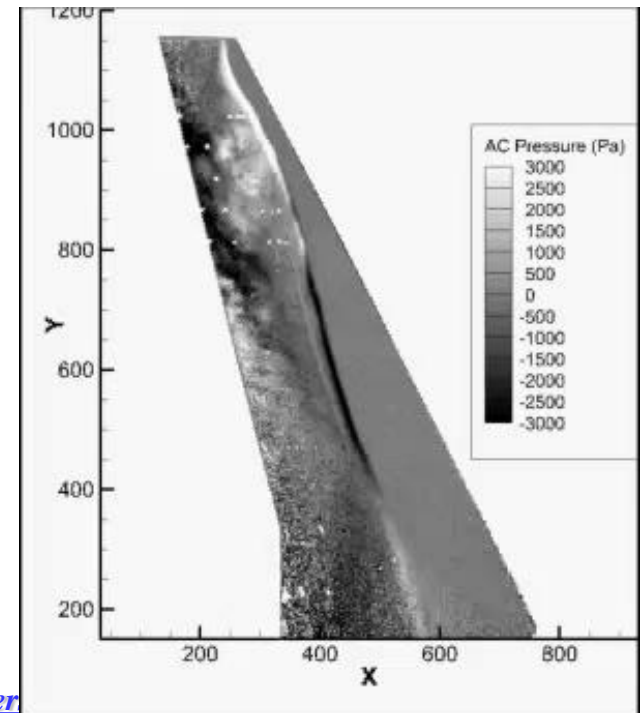
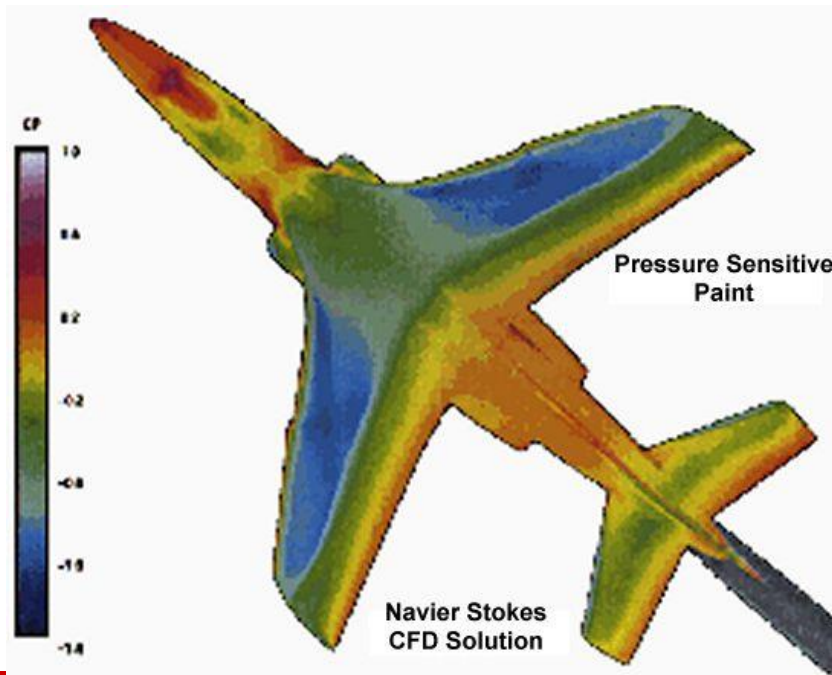
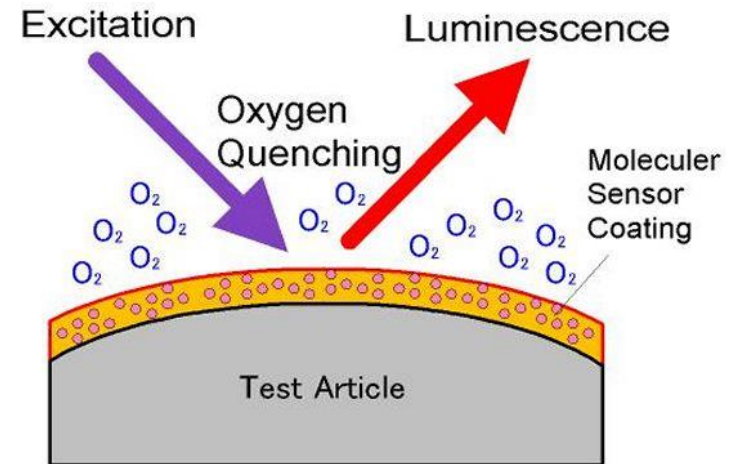
• *Oxygen quenching effect*



• *Thermal quenching effect*

# ❑ BASIC PRINCIPLES OF PRESSURE SENSITIVE PAINT (PSP)

- *Intensity based Methods (most common)*
  - *Full-field using camera*
  - *Point systems using scanning laser*
- *lifetime based Methods (lifetime decay)*
  - *Full-field using camera*
  - *Point systems using scanning laser*

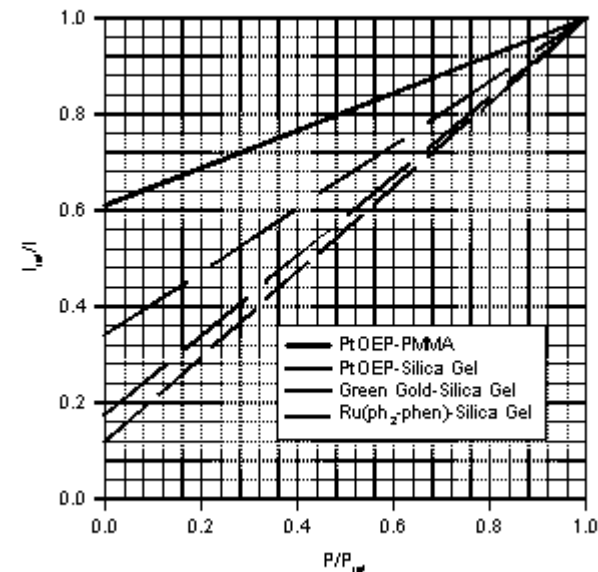
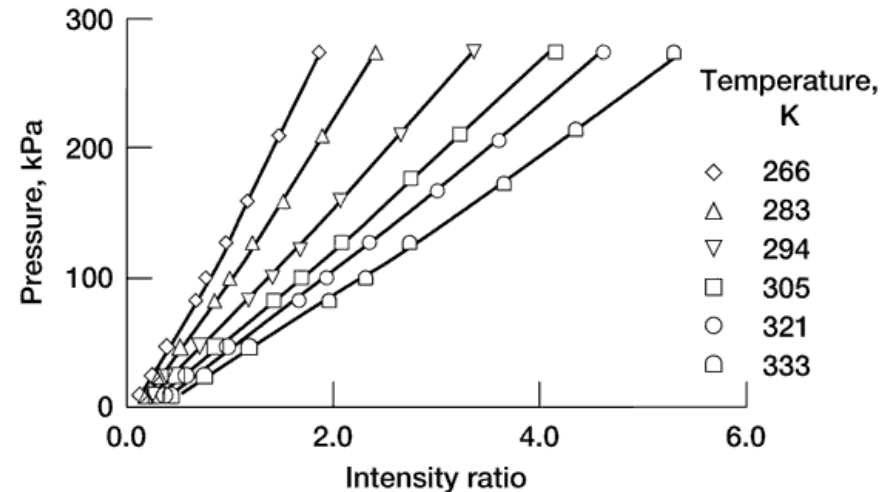
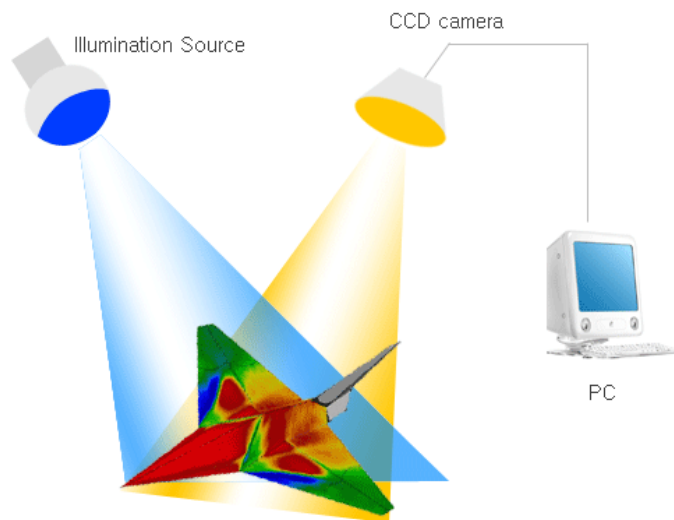


# □ INTENSITY-BASED PSP

- The Stern-Volmer equation is rewritten in the popular intensity ratio form:

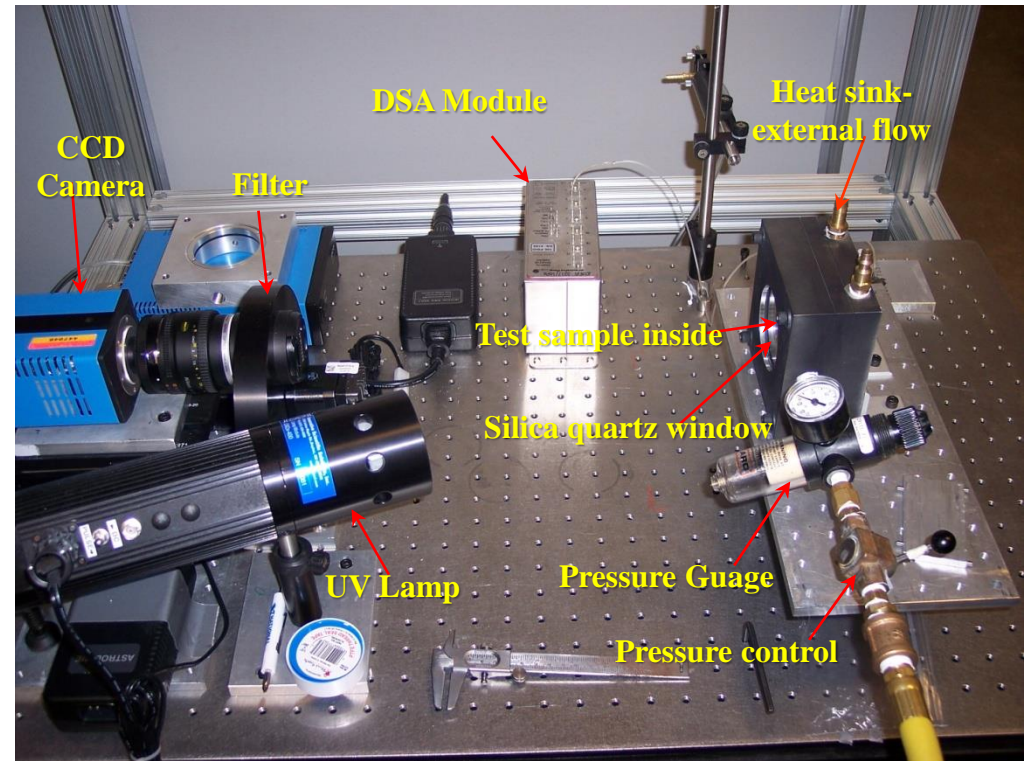
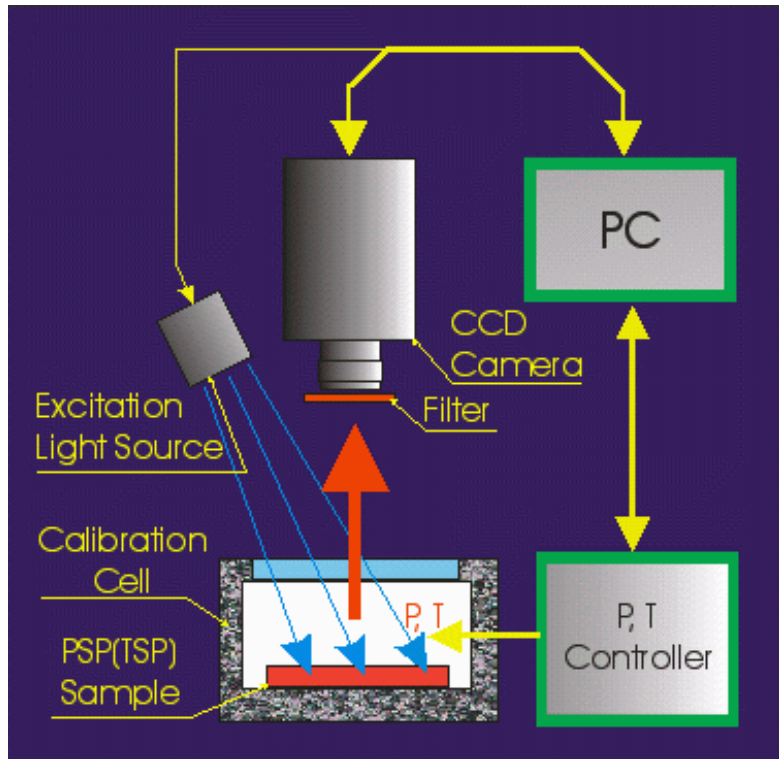
$$\frac{P}{P_{REF}} = A + B \frac{I_{REF}}{I}$$

- $A$  and  $B$  are highly dependent on the luminophore and binder material as well as the temperature sensitivity of the materials used to make the paint. A 2nd order curve generated from calibration data is most often used.





# CALIBRATION SETUP FOR PSP MEASUREMENTS

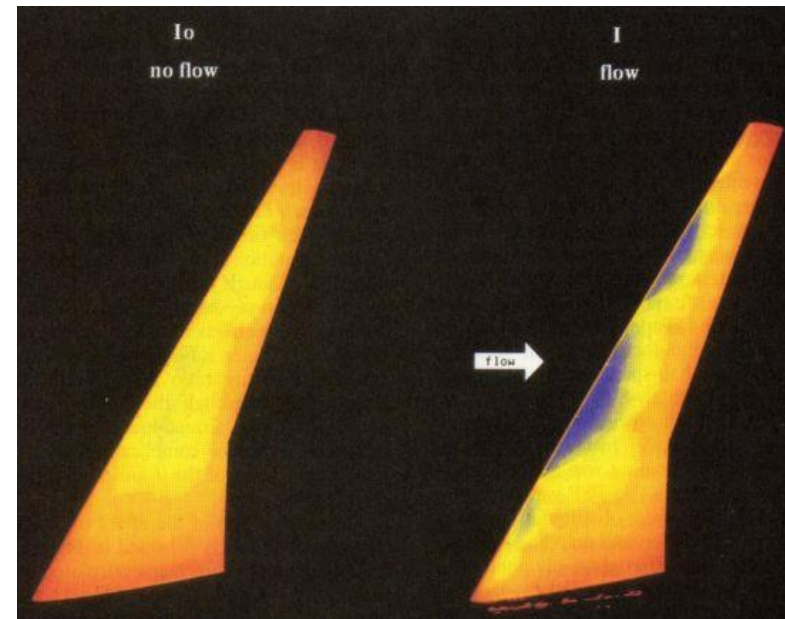
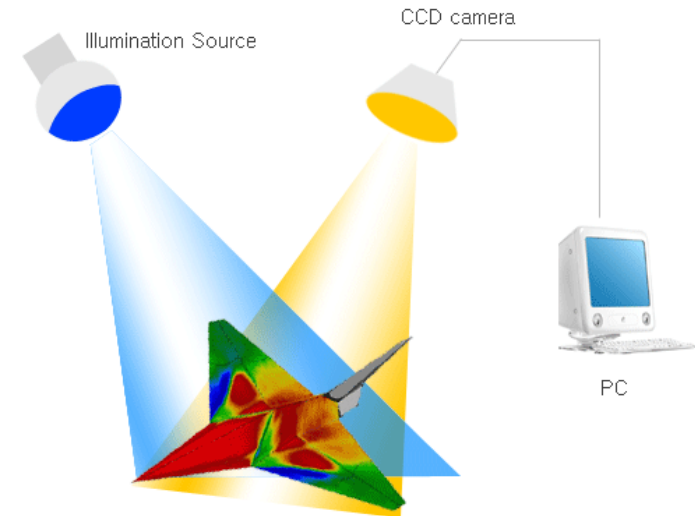


- *Pressure air pipe to control the pressure in the chamber*
- *Water recirculation to control the temperature on the sample plate*

# □ INTENSITY-BASED PSP

$$\frac{P}{P_{REF}} = A + B \frac{I_{REF}}{I}$$

- *Requires two readings, a reference at constant pressure (wind off) and an unknown data point (wind-on)*
- *Ratio of intensities  $I_{REF}/I$  is inversely proportional to the air pressure*
- *The excitation and detection systems must be spectrally separated, (>10<sup>-6</sup> attenuation in stop band).*
- *Simplest technique, most sensitive*
- *Very sensitive to motion between wind-off and wind-on*
- *A long period of time can elapse between reference and data.*
- *images resulting in significant changes in contamination of paint, light stability, etc. that cannot be normalized by the reference condition.*



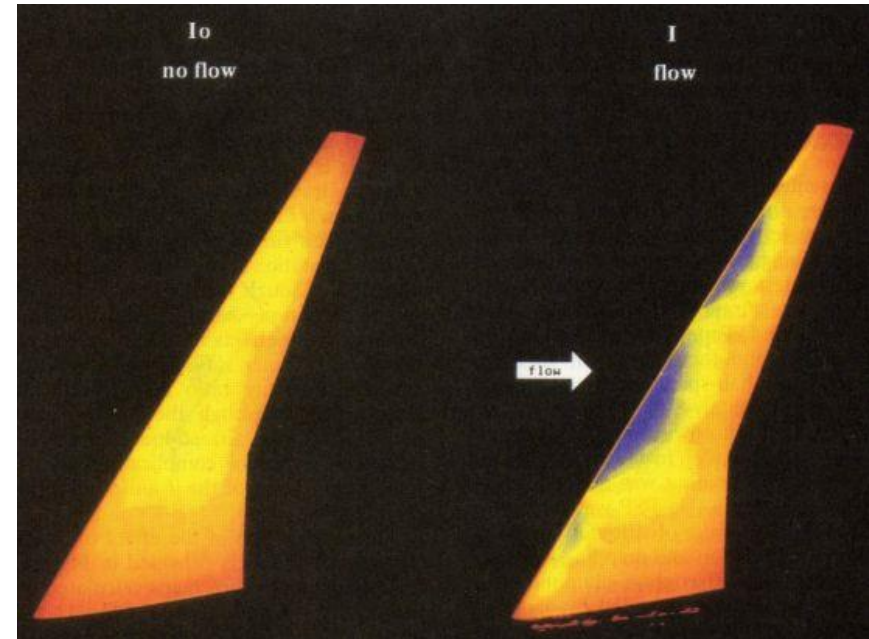


# □ INTENSITY-BASED PSP

$$\frac{P}{P_{REF}} = A + B \frac{I_{REF}}{I}$$

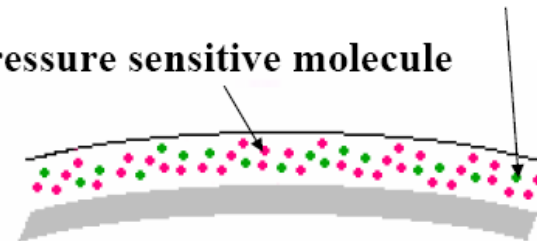
## Advantages:

- *Eliminate wind off images and image registration problems. It works in theory*
- *In practice, due to homogeneity problems of dispersing of two kinds of molecules, it actually requires a double set of ratios, often called ratio of ratios method.*



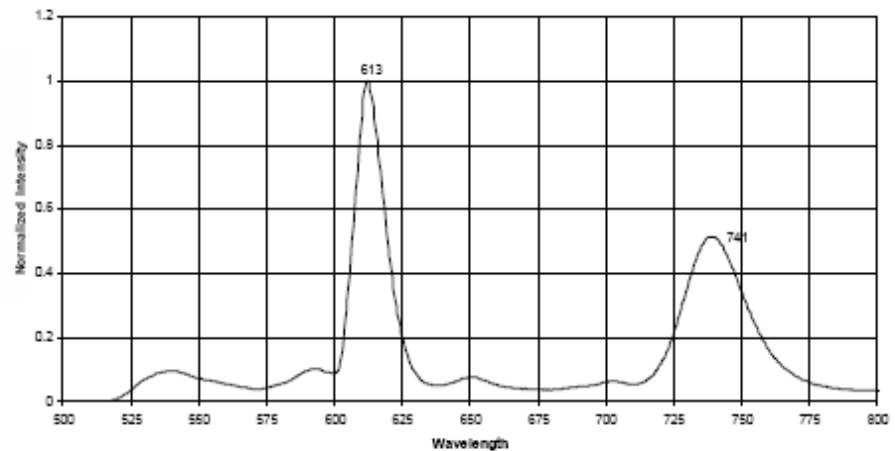
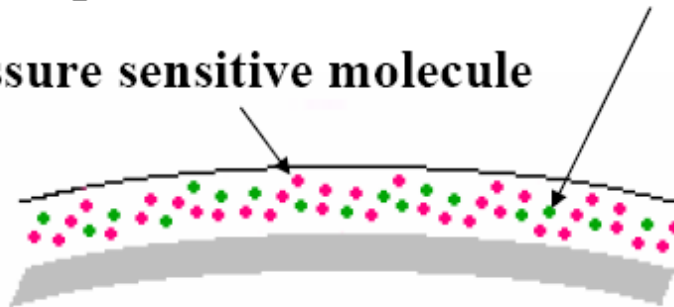
## Self-Referencing paints

Pressure insensitive molecule  
Pressure sensitive molecule



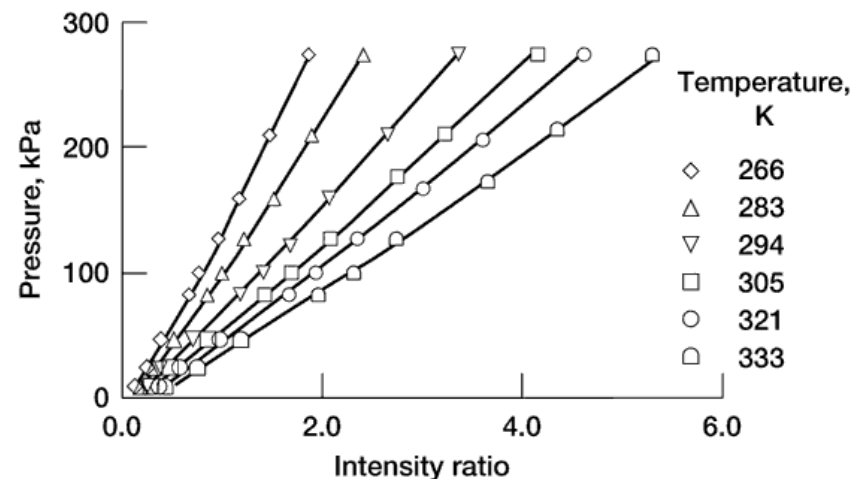
# □ INTENSITY BASED PSP-TEMPERATURE COMPENSATION

Temperature sensitive molecule  
Pressure sensitive molecule



## Advantages:

- Measure temperature to compensate for temperature sensitivity of PSP.
- This technique requires all four images to be aligned.



# LIFETIME-BASED PSP MEASUREMENTS

- *Easiest to do with a point measurement but can use time resolved cameras to measure lifetime decays of the probe molecules.*
- *Point measurements require a pulsed light source and detector (PMT, PD)*
- *Time resolved imaging requires a double pulse type experiment to measure the decay times (gated camera, interline transfer camera capable of multiple flash integration).*

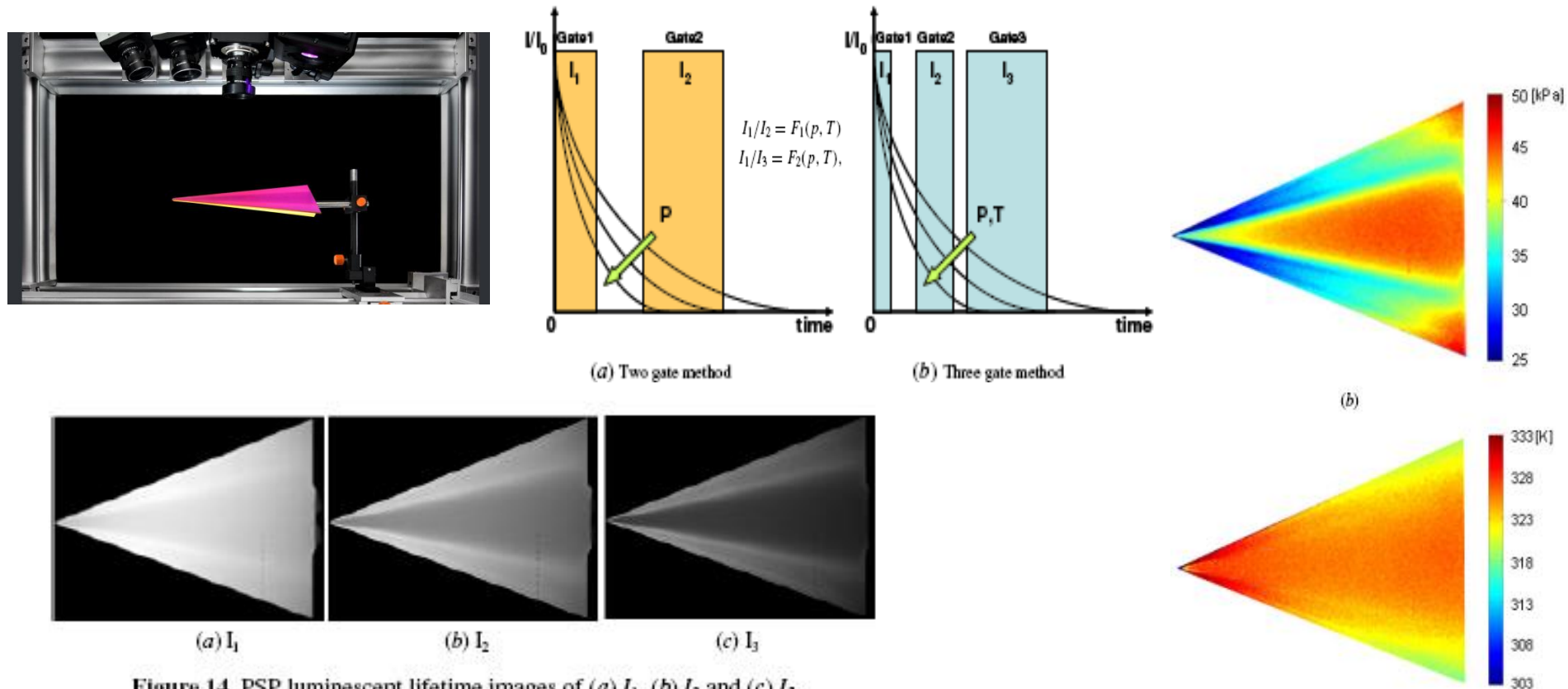


Figure 14. PSP luminescent lifetime images of (a)  $I_1$ , (b)  $I_2$  and (c)  $I_3$ .

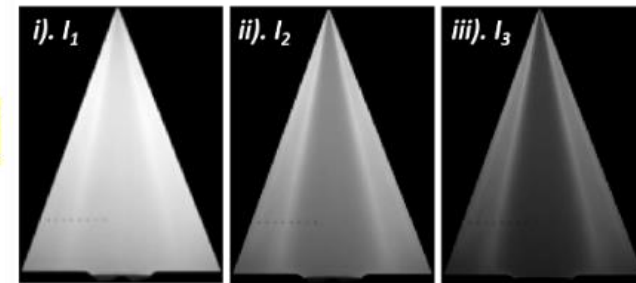
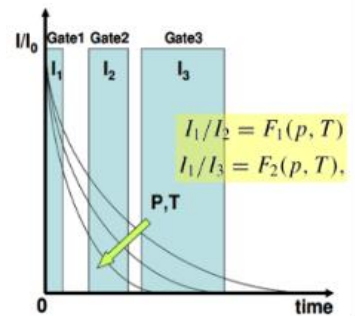
# LIFETIME-BASED PSP

- **Benefits:**

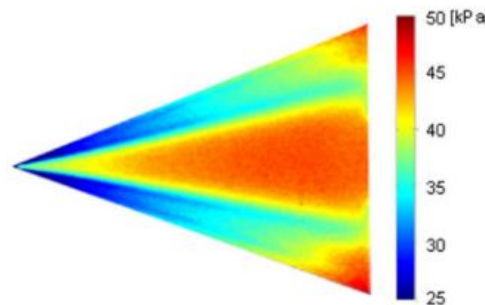
- *Eliminates the need for aligning two or three images since the pair of (or three ) images are taken at the same condition relatively close in time (micro-seconds).*
- *Pressure and temperature distributions can be determined simultaneously.*

- **Disadvantages:**

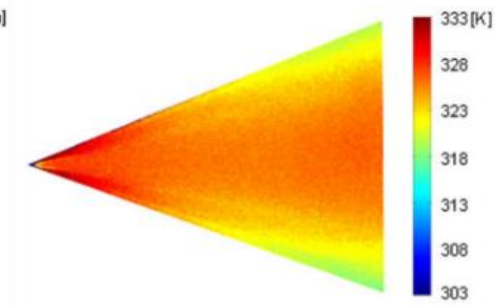
- *Requires three gates to generate two equations of gate ratios to solve for pressure and temperature at each point (pixel).*
- *Camera noise is much higher, especially gated intensified cameras.*
- *Paints have tended to be more spatially noisy from lifetime differences between molecules (homogeneity problem).*



(b). Acquired raw images for the 3-gated, lifetime-based PSP measurements



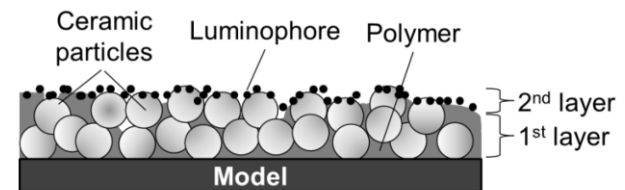
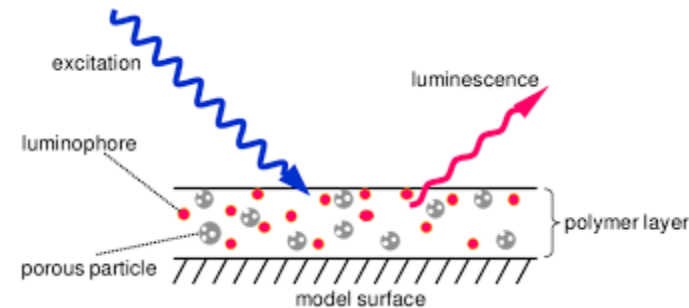
(c). The measured pressure distribution over the test model



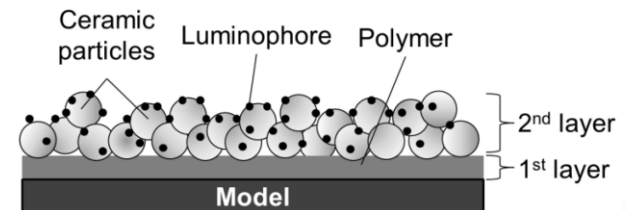
(d). The measured temperature distribution over the test model

# □ PSP/TSP COATINGS

- **PSP coatings used at NASA GRC**
  - Boeing PF2B – ruthenium bathophenanthroline in silicone rubber binder (soft paint, chlorinated solvent)
  - UW (ISSI) FIB – PtTFPP in FIB copolymer binder (hard, good steady state paint)
  - NASA Langley – PtTFPP in FEM (very hard, very smooth finish)
  - ISSI sol-gel – Ru(ph2-phen) and PtTFPP on sol-gels (higher frequency response)
  - Anodized aluminum – dip coated Ru(ph2-phen) on anodized surface (very high freq. response)
  - UW PtOEP in MAX acrylic copolymer (ice paint)
- **TSP coatings**
  - Boeing TSP (range: 0 to 100°C, sensitivity ~ -3%/°C)
  - EuTTA in commercial clear or shellac (-20 to 80°C, ~ -4%/°C)
  - Thermographic phosphors in high temp binders (-20 to >1000°C)



(a)

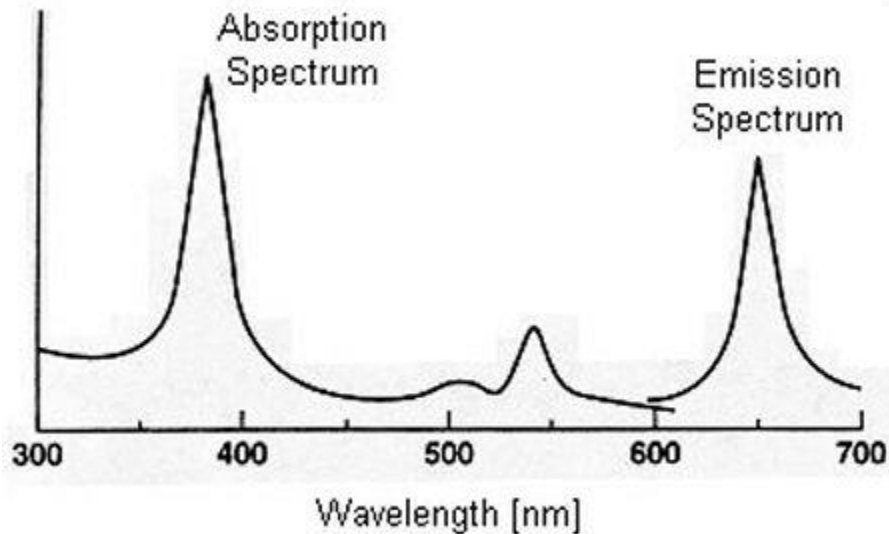
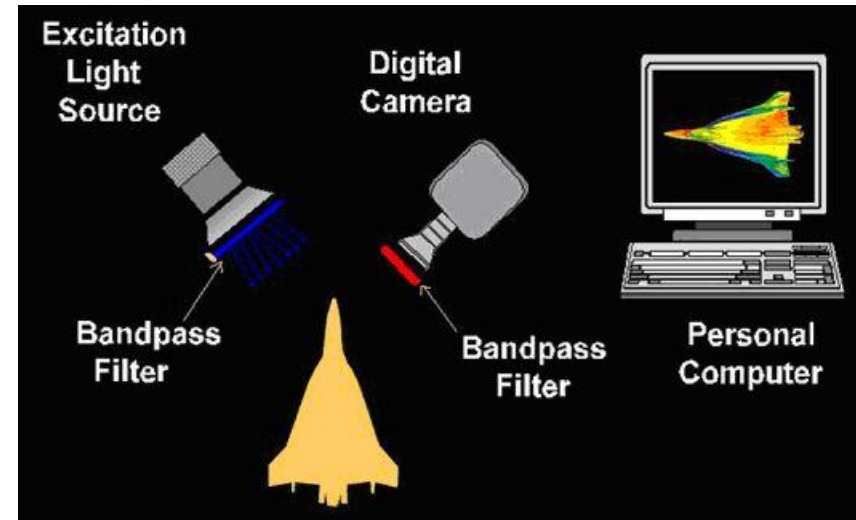


(b)

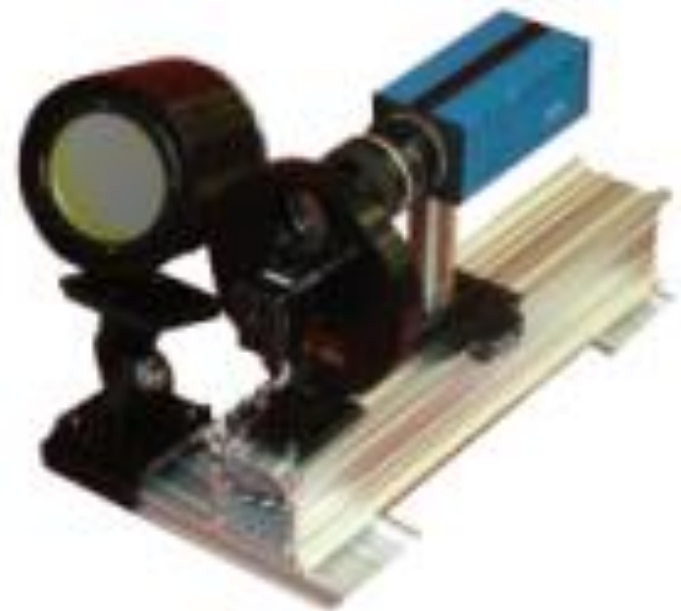


# □ Intensity based PSP

- **Excitation:**
  - *Continuous Sources: LEDs, Filtered lamps (Halogen, Xenon), Lasers*
  - *Pulsed Sources for instantaneous or periodic measurements: LEDs, Xenon, strobes/flash*
- **Detectors**
  - *Cooled Scientific grade CCD cameras (slow scan, low noise), PMT, PD*



*Typical PSP absorption and emission spectra  
[from McLachlan and Bell, 1995]*



# □ Calibration for PSP

## A-priori Calibrations

- *Paints are typically calibrated in a cell that varies pressure and temperature and has a reference measurement – this calibration is used when no on-model instrumentation exists.*

## In-situ Calibration

- *Uses standard on-model instrumentation to calibrate the paint/images in place*
- *Compensates for temperature differences from reference data, spatial temperature differences are averaged among all the points used to generate a calibration*
- *In practice both calibrations are typically used*

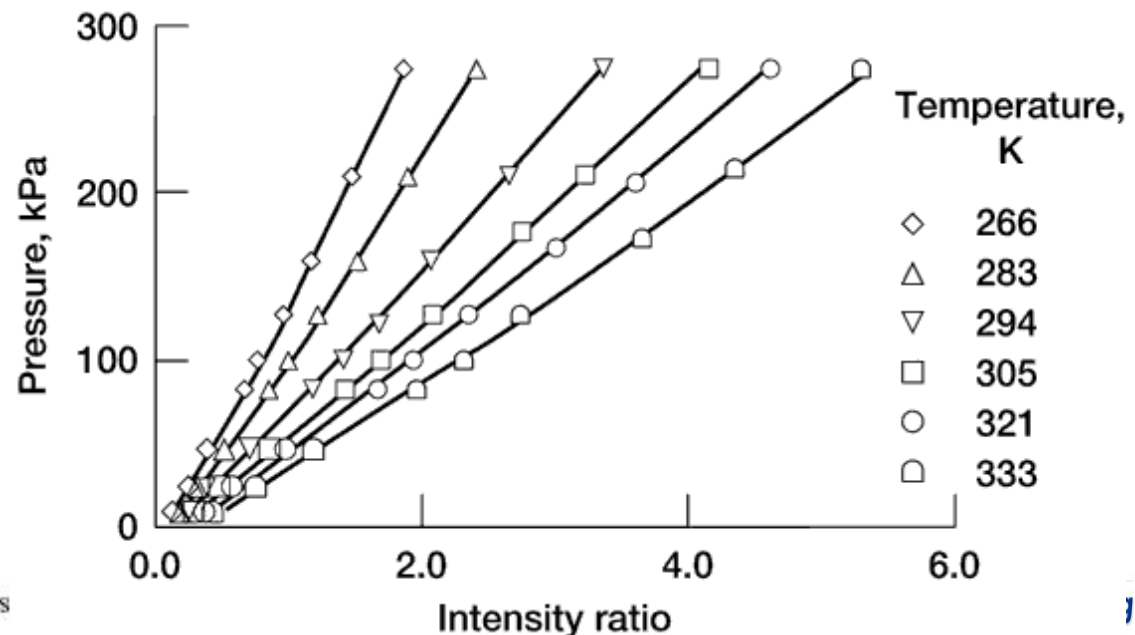
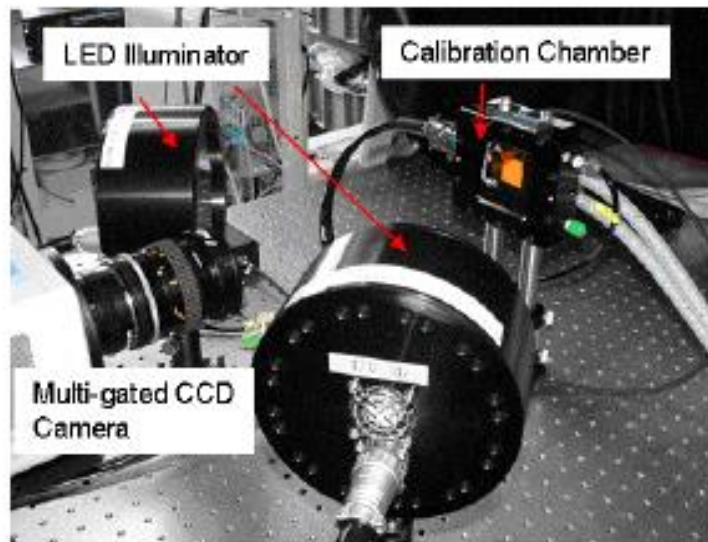
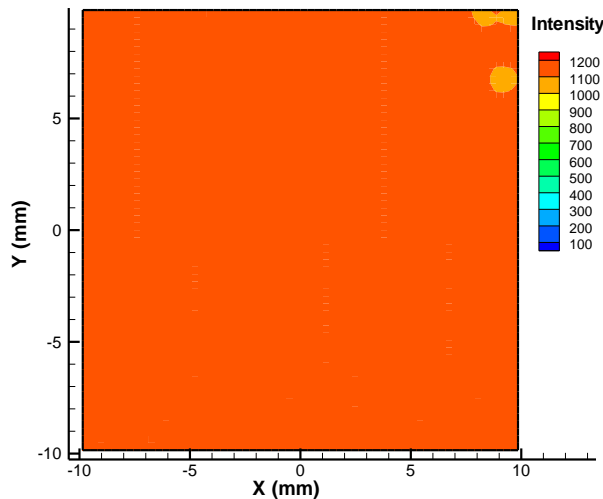
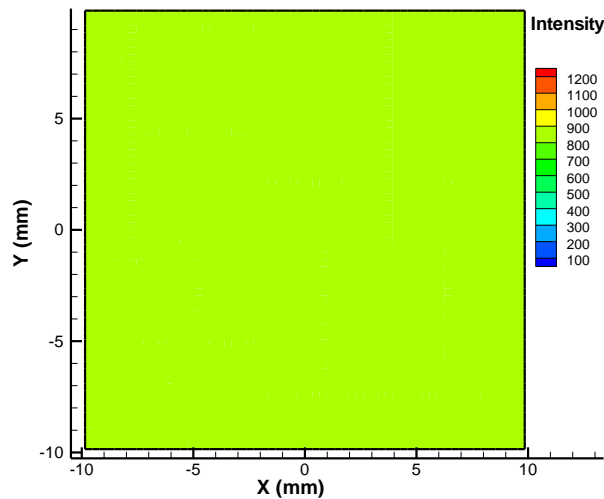


Figure 3. Photo of a multi-gated camera and LED illuminators

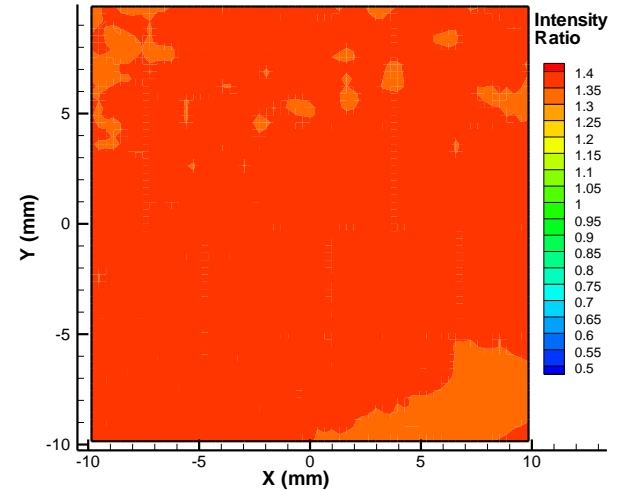
# PSP calibration image process



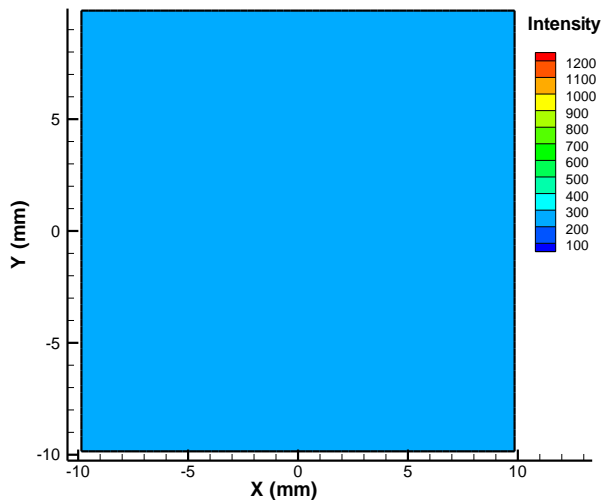
Reference Intensity:  $I_{ref}$



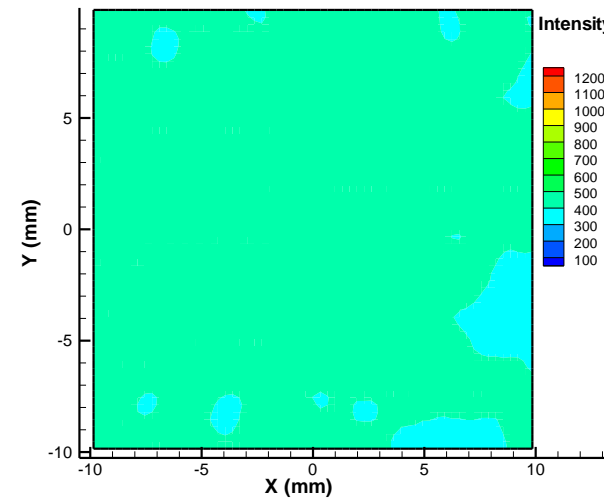
$P/P_{ref}=7.2$ : Intensity



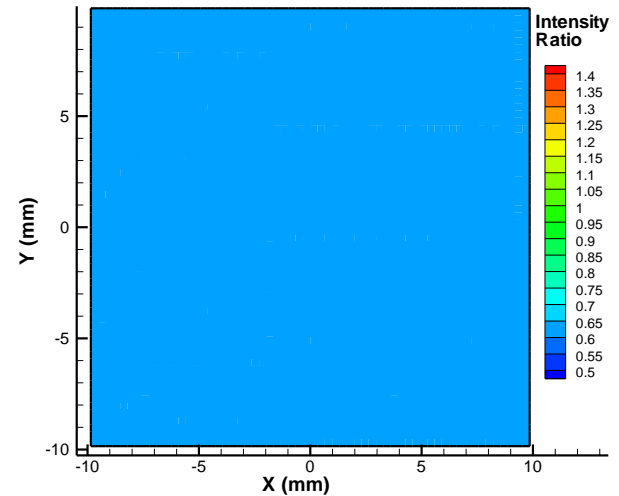
Intensity ratio:  $I_{ref}/I$



Reference Intensity:  $I_{ref}$

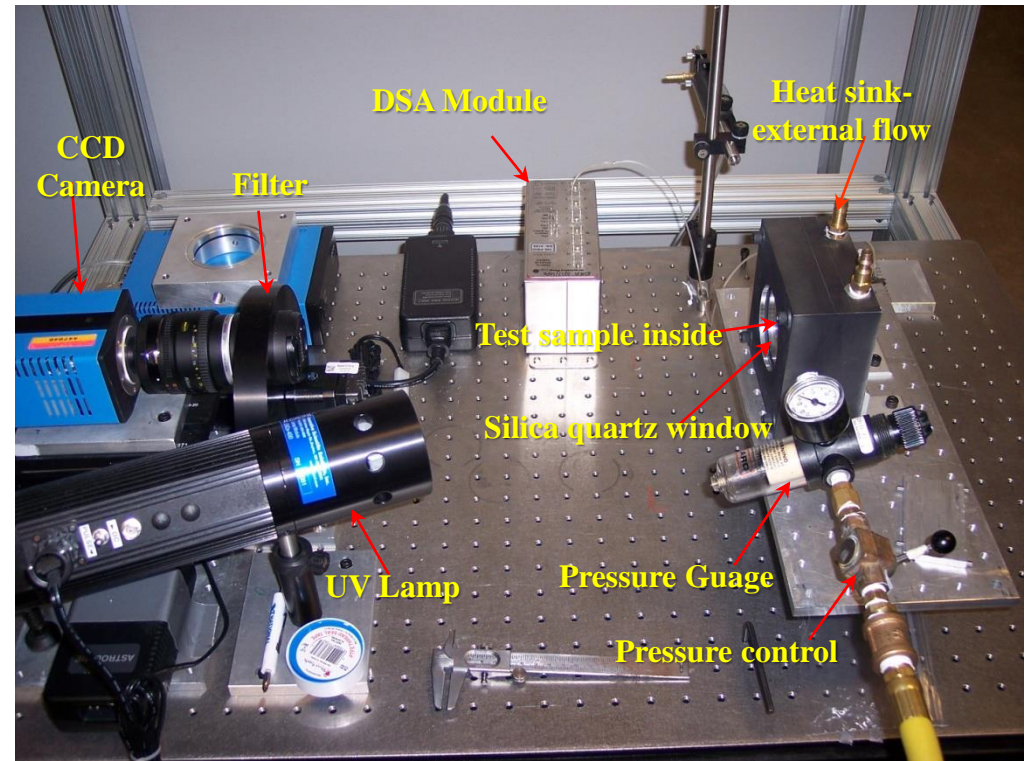
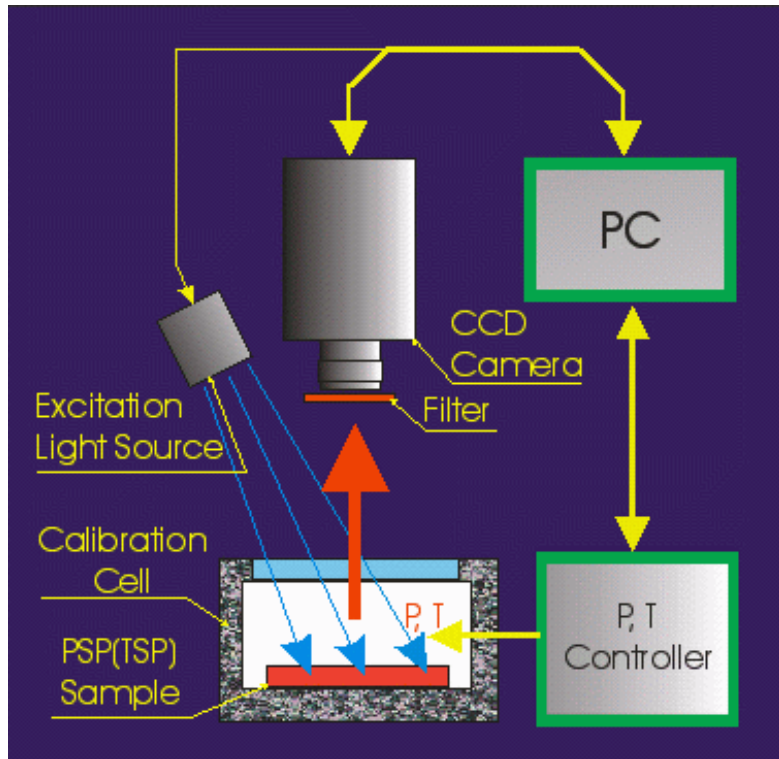


$P/P_{ref}=0.23$  Intensity



Intensity ratio:  $I_{ref}/I$

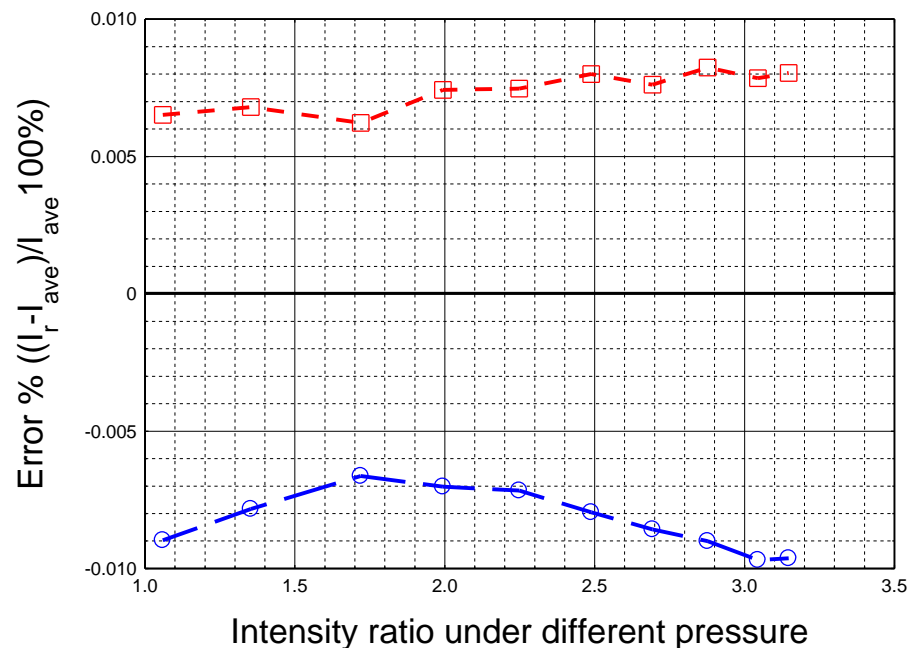
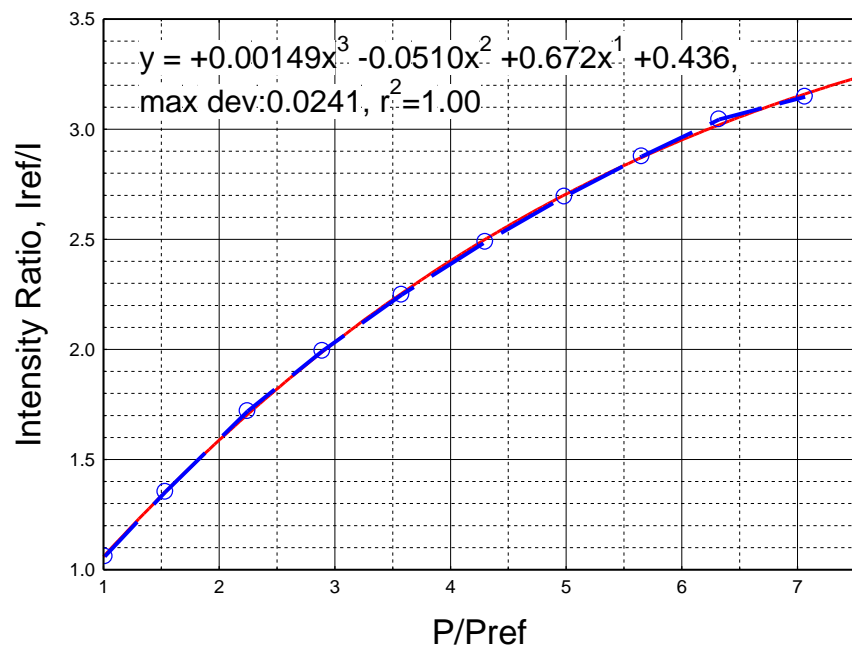
# CALIBRATION SETUP FOR PSP MEASUREMENTS



- *Pressure air pipe to control the pressure in the chamber*
- *Water recirculation to control the temperature on the sample plate*



# CALIBRATION CURVE – POSITIVE PRESSURE

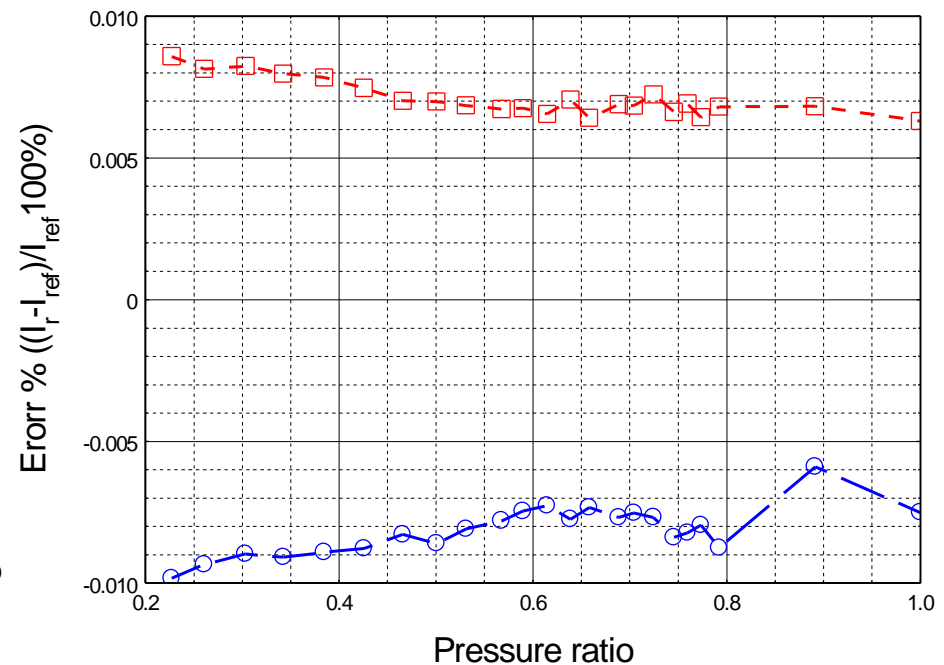
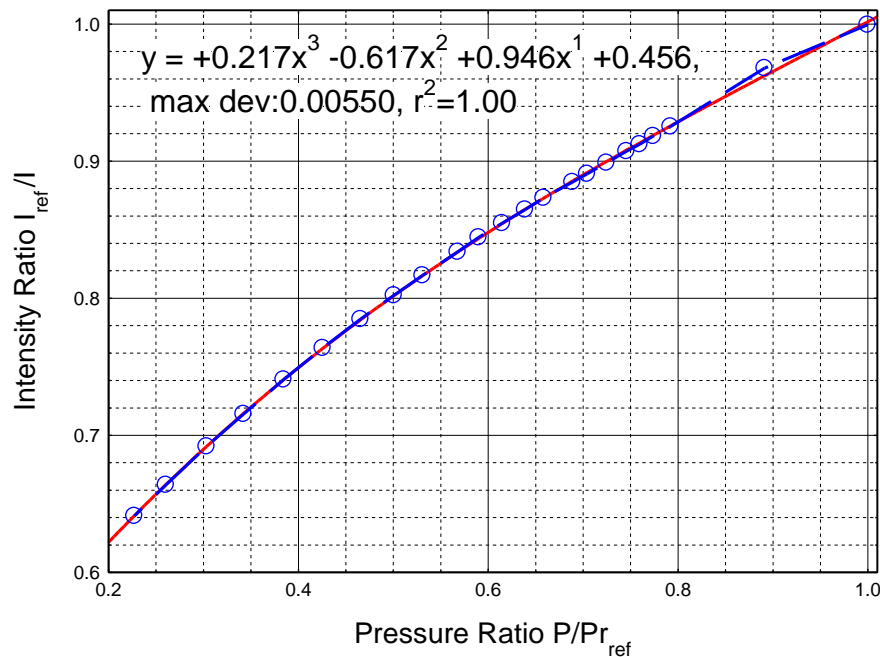


***Calibration curve for the paint positive pressure and error analysis***

- Fit function:  $y = 0.00149x^3 - 0.0510x^2 + 0.672x + 0.436$
- Error level is below 1%



# CALIBRATION CURVE – VACUUM PRESSURE



Calibration curve for the PSP paint under vacuum pressure and error analysis

- Fit function:  $y = 0.217x^3 - 0.617x^2 + 0.946x + 0.456$
- Error level is below 1%

# □ UNCERTAINTY FOR PSP MEASUREMENTS

- *Characterization of the paint and calibration errors (a-priori, in-situ calibration, photo degradation, paint contamination, paint intrusiveness, time response)*
- *Measurement system errors (detector noise, illumination spectral and temporal stability, spectral leakage)*
- *Signal analysis errors (registration from model motion and deformation, incomplete temperature compensation, self illumination, resectioning on a non-deformed grid)*
- *The major contributor is temperature uncertainty which can account for up to 90% of the total uncertainty*

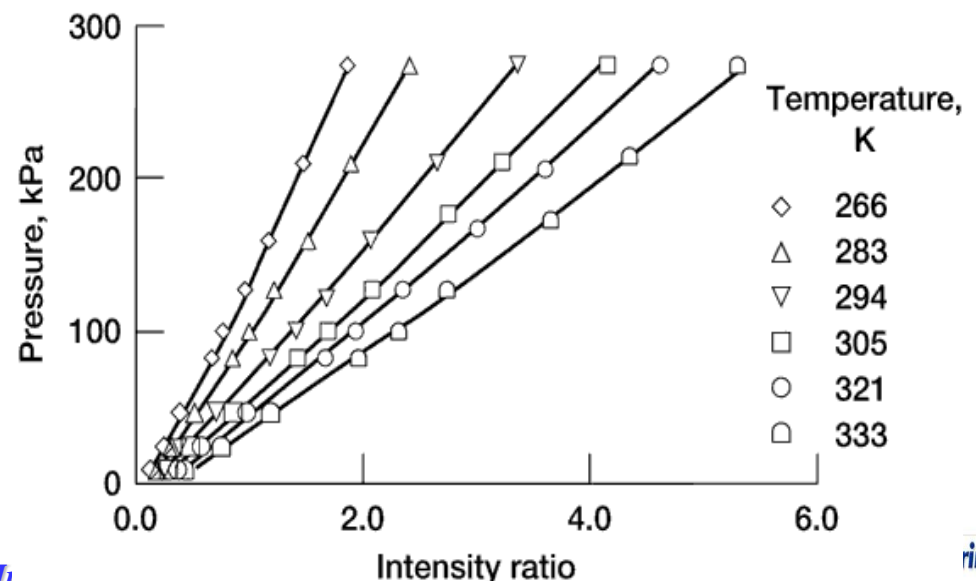
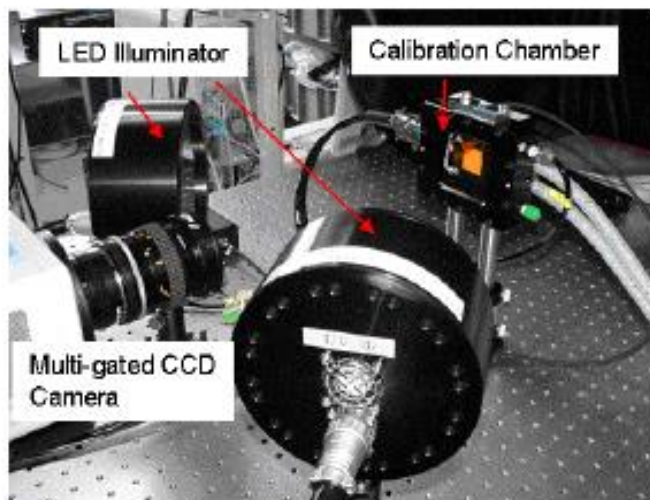


Figure 3. Photo of a multi-gated camera and LED illuminators.

# PSP APPLICATION EXAMPLES

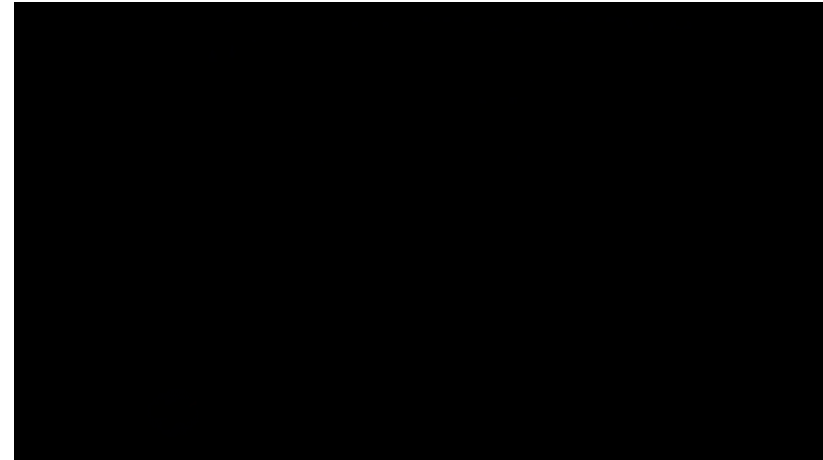
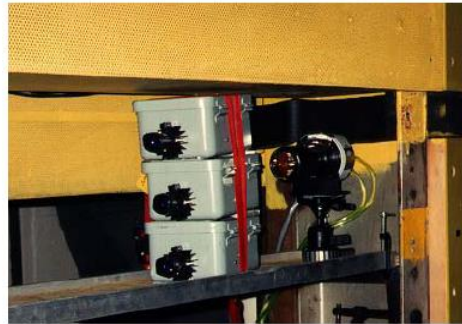


## Rotating PSP/TSP on 22" Fan Model

GRC 9'x15'LSWT



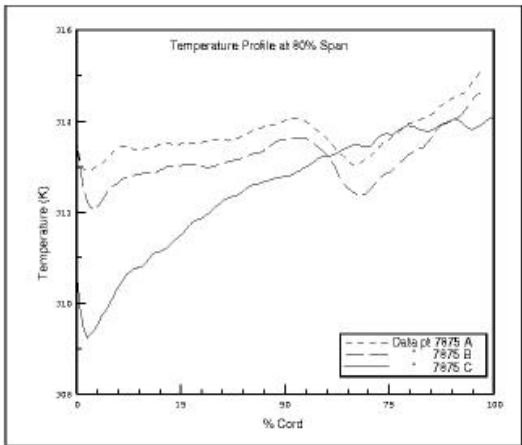
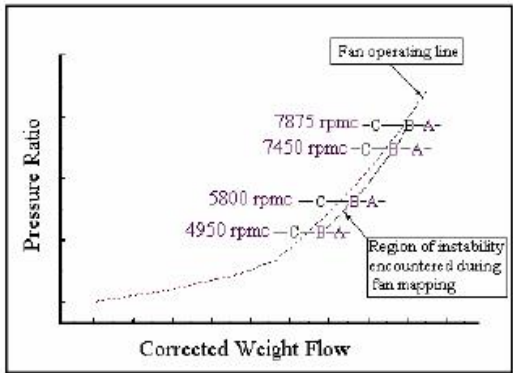
<https://www.youtube.com/watch?v=T-T-njklwKE>



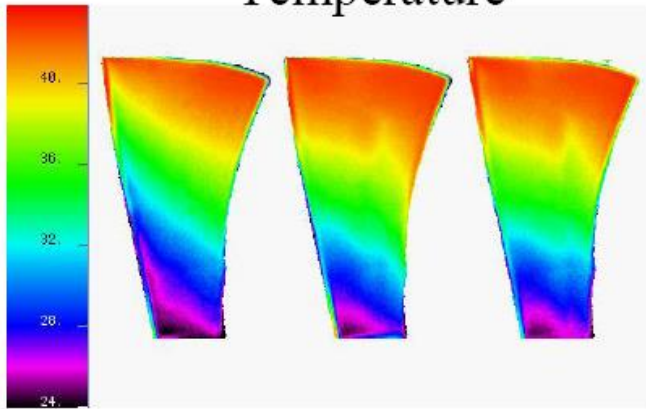
# PSP APPLICATION EXAMPLES



## Rotating PSP/TSP



Temperature

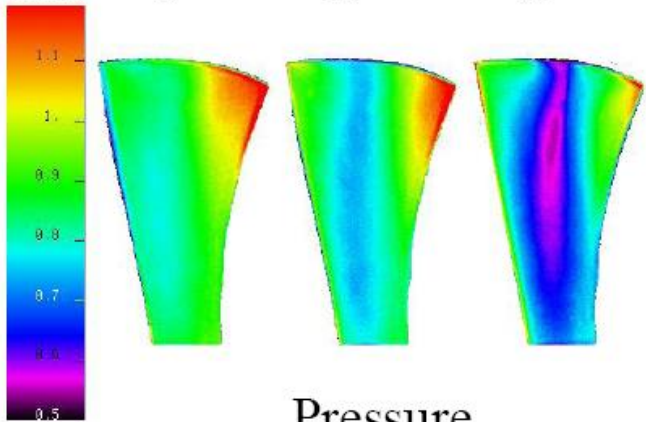


7875 RPM

C

B

A



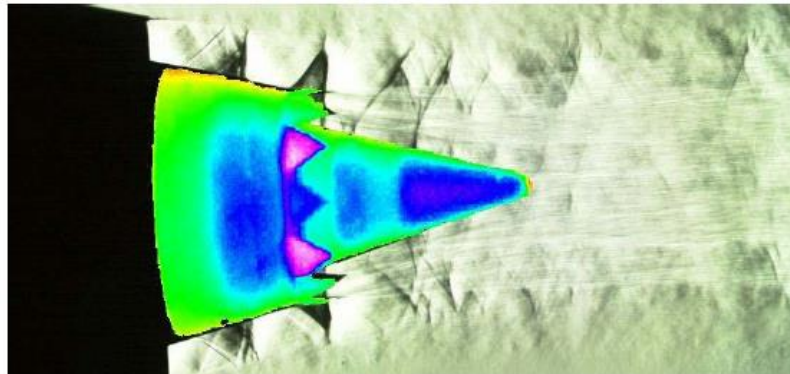
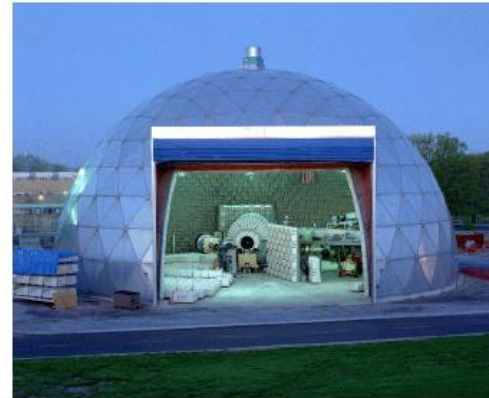
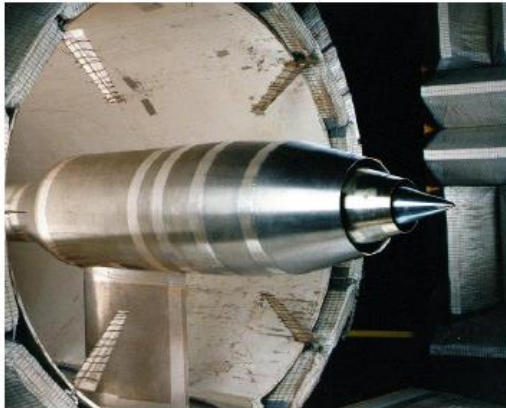
Pressure



# PSP APPLICATION EXAMPLES



## Nozzle Test in APL using Lifetime PSP

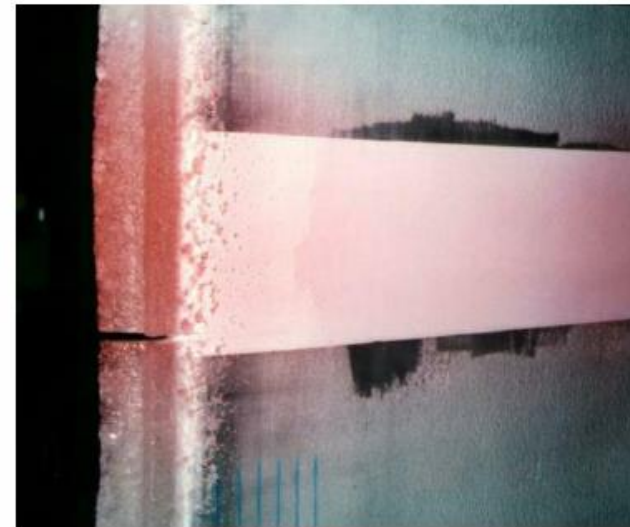
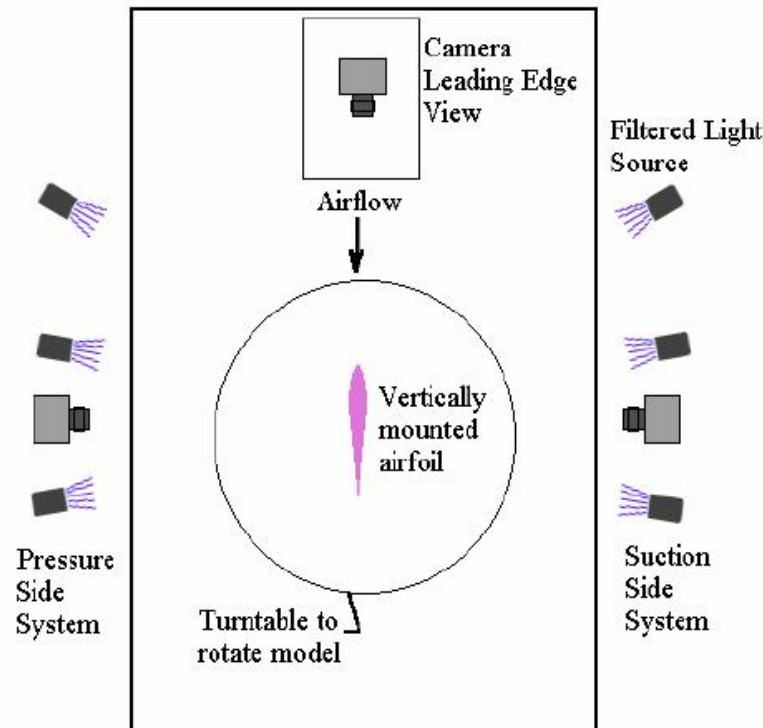


# PSP APPLICATION EXAMPLES

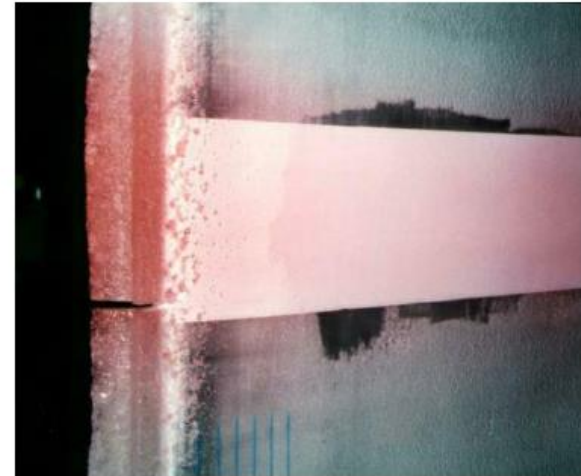
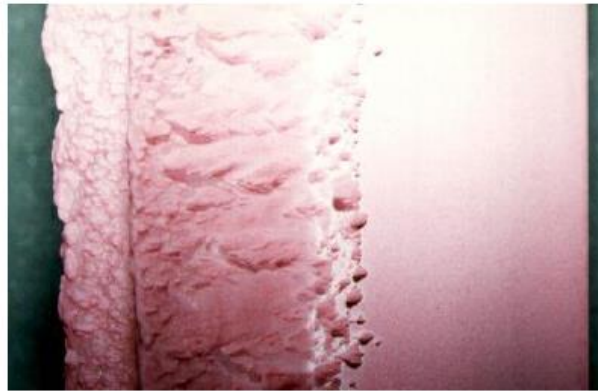


## PSP on Ice

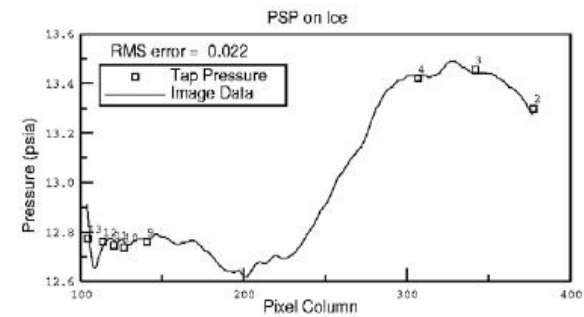
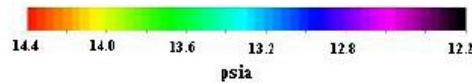
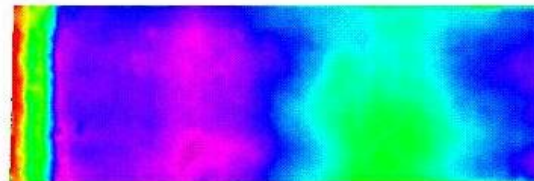
### Test setup using PSP on Ice in the IRT



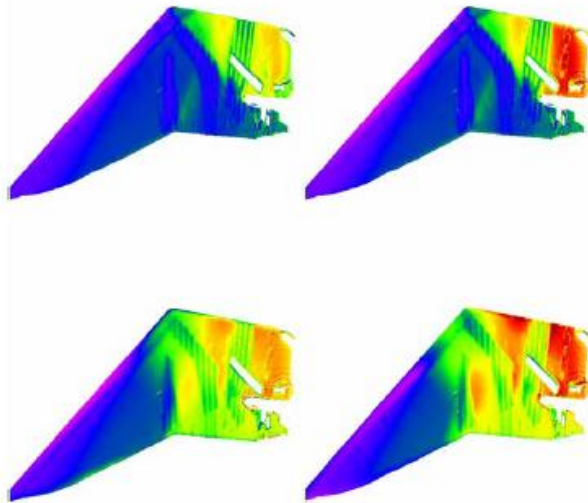
# PSP APPLICATION EXAMPLES



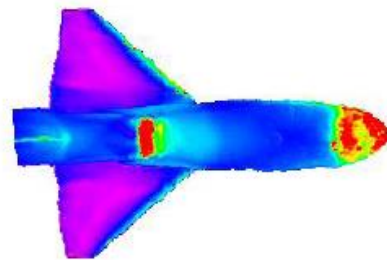
## PSP on Ice in the IRT



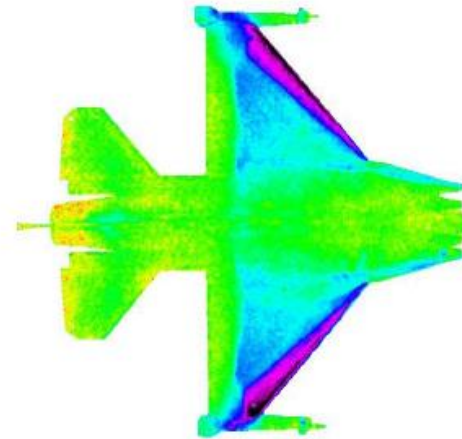
# PSP APPLICATION EXAMPLES



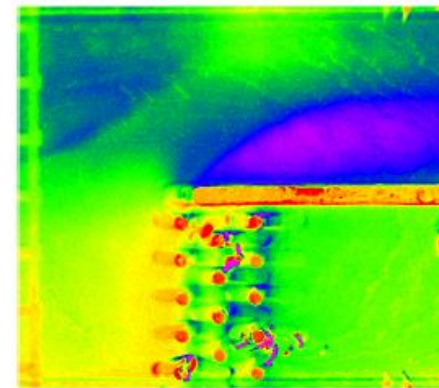
Inlet Test in the 10'X10'



Pioneer Rocketplane in the 1'X1'



Lifetime PSP F16 Test at ARC

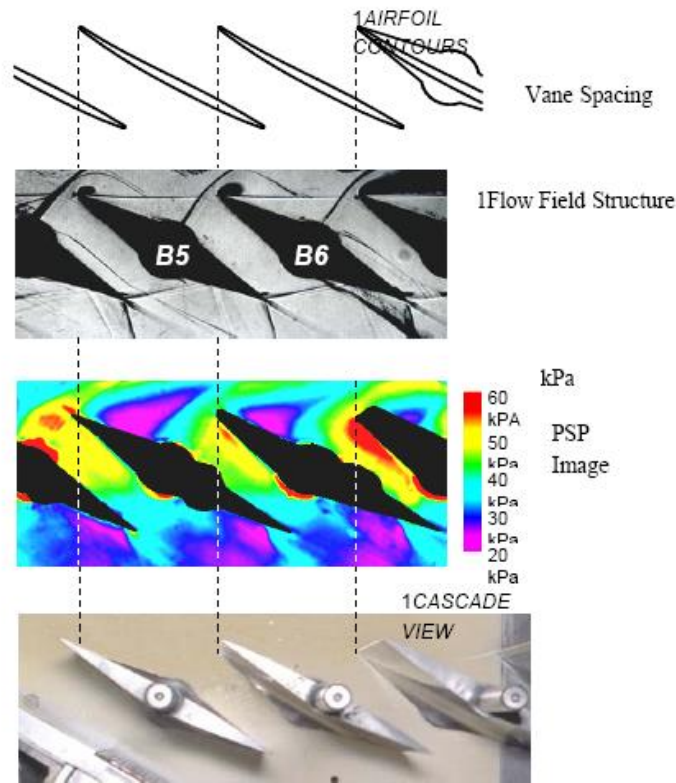


Turbine Cooling Passage  
simulation





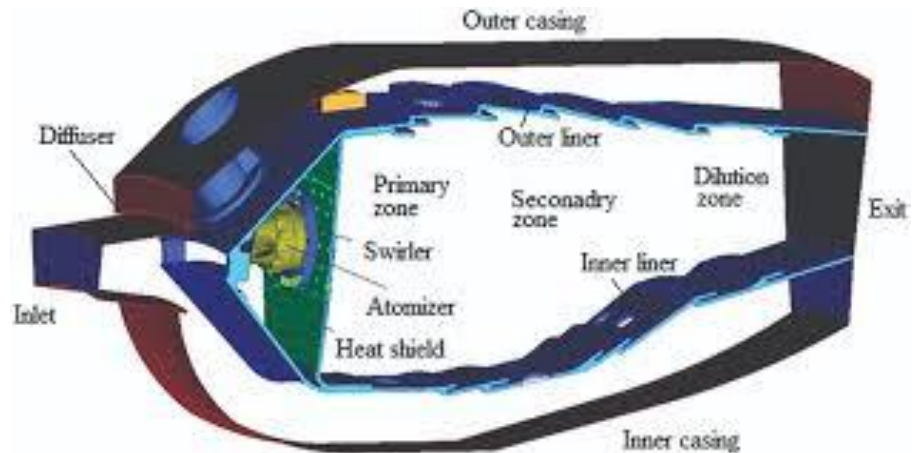
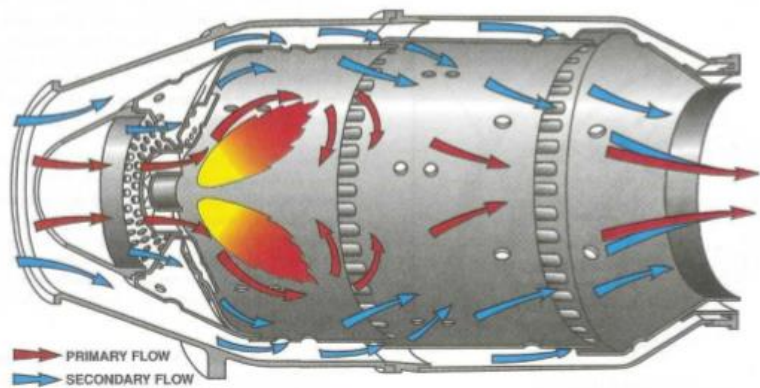
## Transonic Linear Cascade Facility



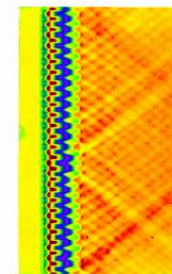
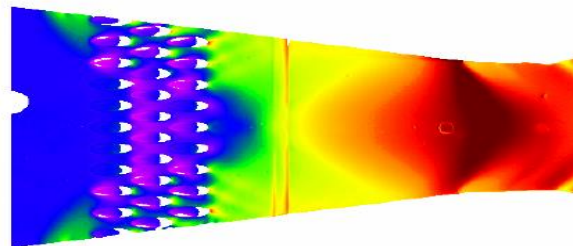
Borescope imaging of confined space surfaces

Cascade flow field for Mach number 1.3 showing relation of blades, shocks and pressure profile on the back wall.

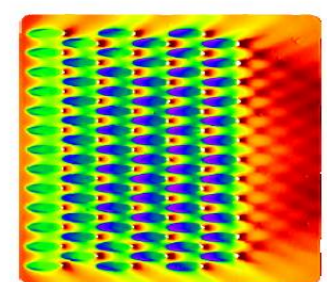
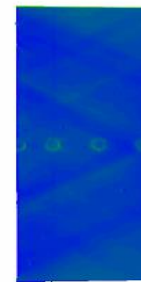
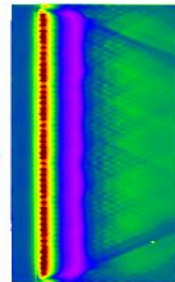
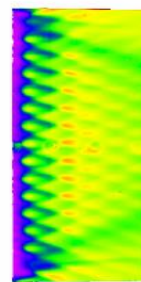
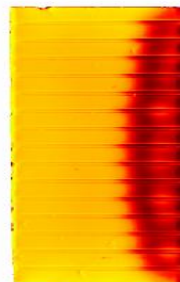
# 



## Boundary Layer Control Tests in the 1'X1'SWT



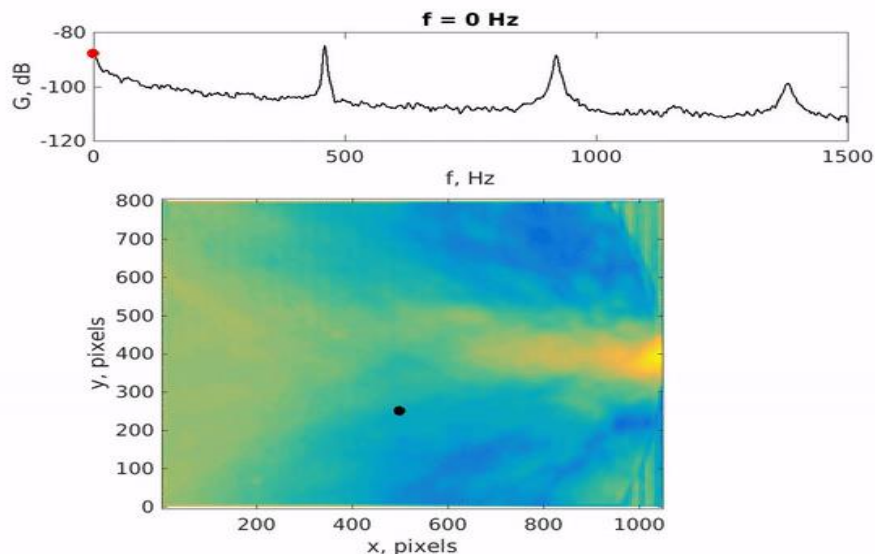
Methods using suction and blowing for boundary layer enhancement



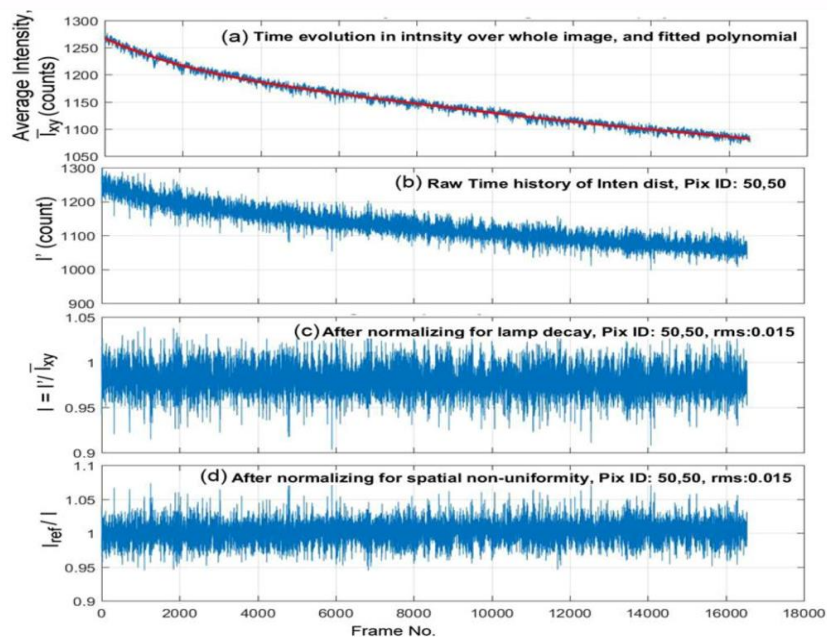




# PSP Application – UNSTEADY PRESSURE MEASUREMENTS

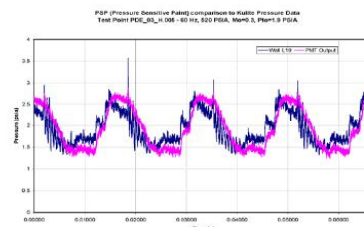


<https://www.youtube.com/watch?v=aRde4PdWc50>

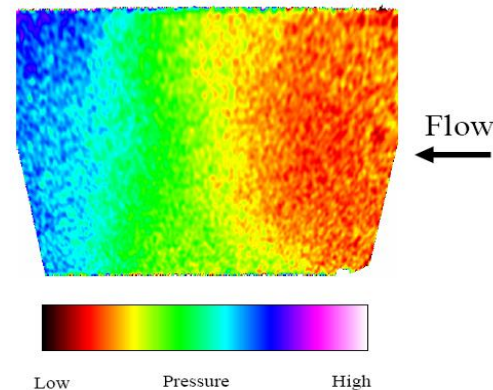


## Dynamic PSP Test: Pulse Detonation Engine Sidewall

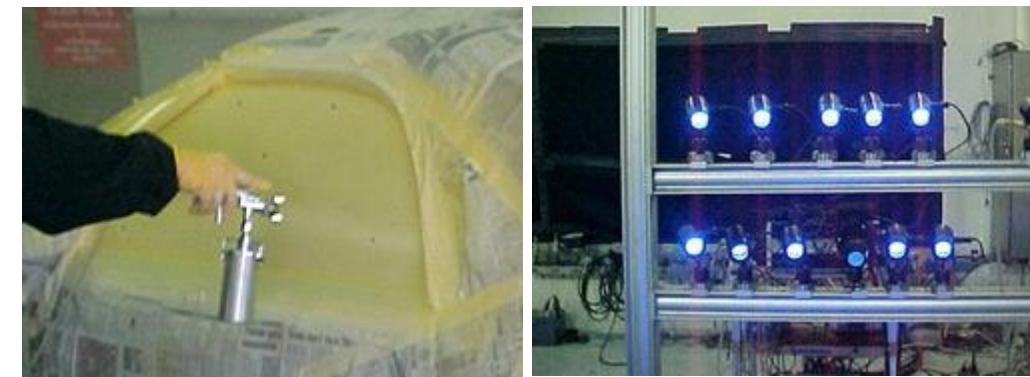
Instantaneous image of pressure field of moving pressure wave



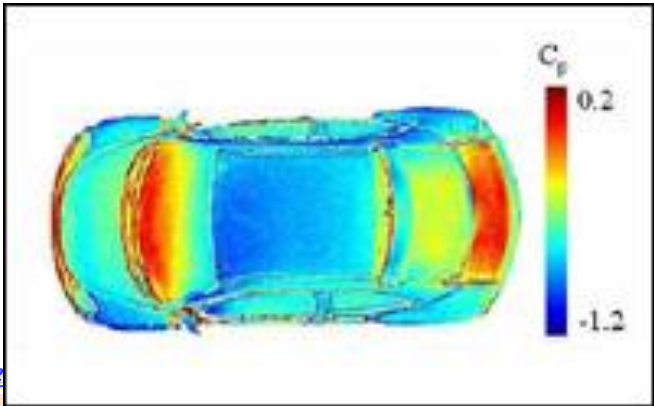
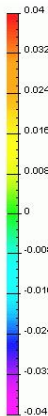
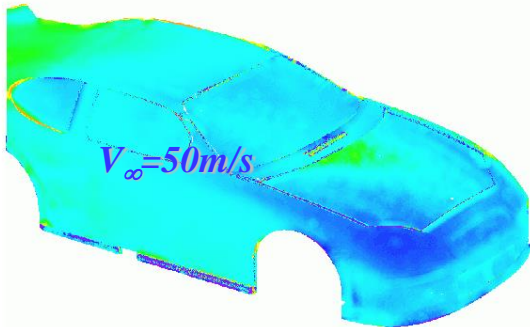
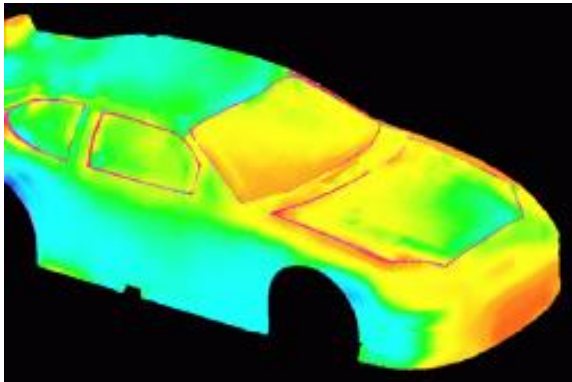
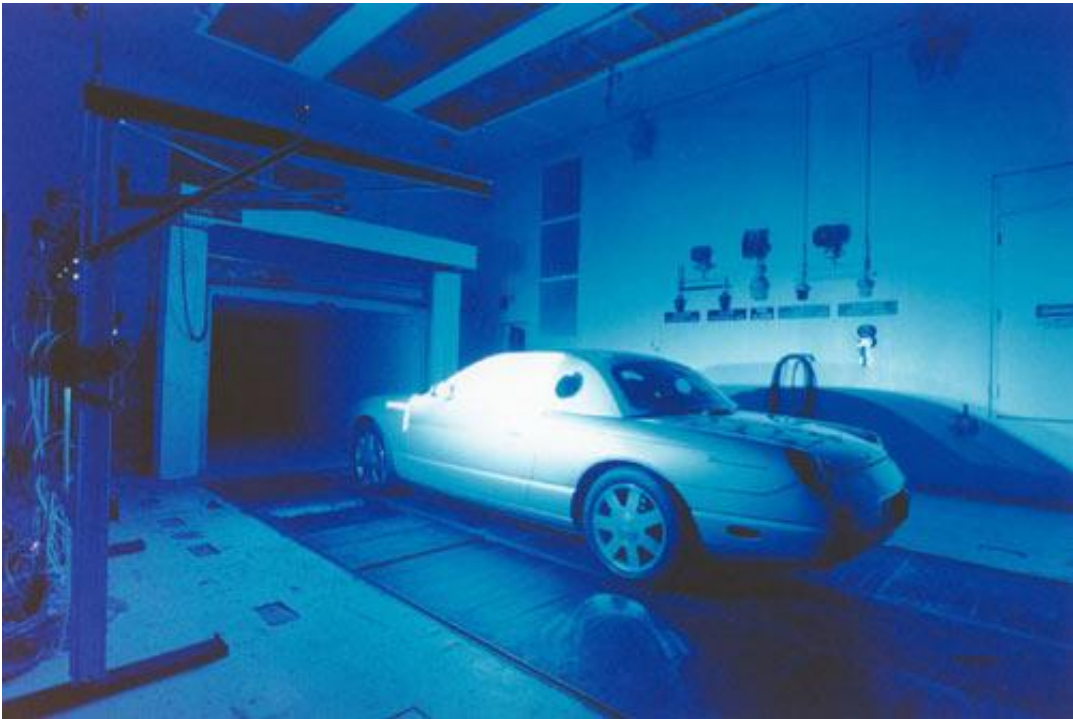
Point PSP measurements vs pressure transducer, paint has lag and  $\sim 1 \text{ kHz}$  response



# ❑ PSP Technique for Low-Speed Applications



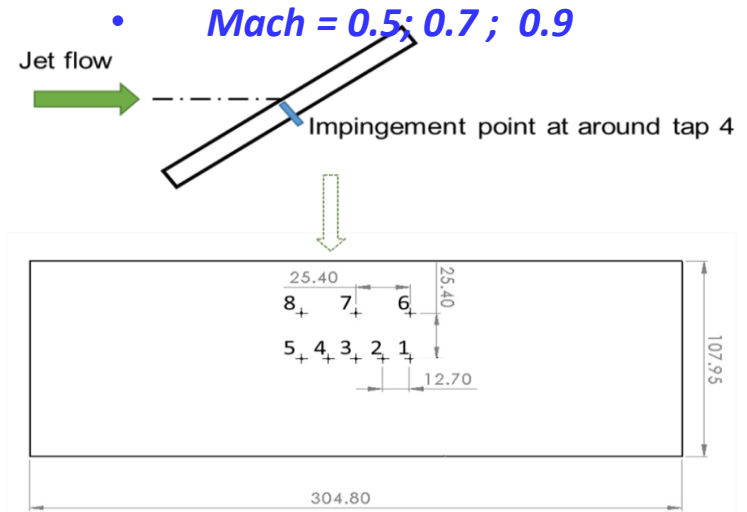
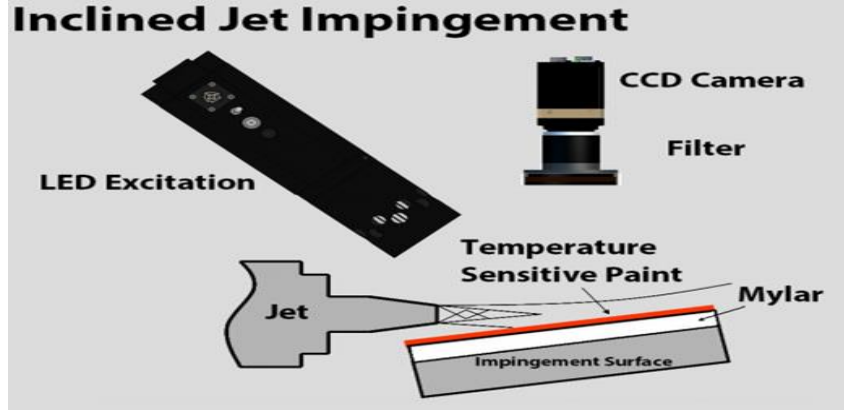
- PSP measurements of a 2002 Ford Thunderbird





# □ AERE545/AERE445X LAB #02 – PSP LABORATORY

- Pressure Distribution Measurements of a Transonic Impinging Jet onto a Flat Plate by using PSP technique



$$\frac{P_{jet}}{P_{ref}} = -0.053032170 + 0.922217040 * \frac{I_{ref} - I_b}{I_{jet} - I_b} + 0.135149696 * \left( \frac{I_{ref} - I_b}{I_{jet} - I_b} \right)^2 - 0.008236630 * \left( \frac{I_{ref} - I_b}{I_{jet} - I_b} \right)^3$$

